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SORTIE DURATION AND
HELICOPTER COMPONENT
FAILURES (AN EMPIRICAL STUDY)

INVENTORY RESEARCH OFFICE

May 1983



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US Army Inventory Research Office
US Army Materiel Systems Analysis Activity
800 Custom House, 2d & Chestnut Sts.
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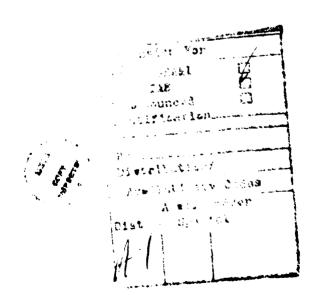


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SUMMARY

1. Background

The purpose of this study was to investigate the effects that sorties have on component failures for Army aircraft. The underlying hypothesis was that the stresses occurring during take off and/or landing cause more failures of aircraft components than does continuous flying. Under this hypothesis an increase in the number of sorties for a given flying hour program results in an increase in demand for spare parts. The Army's current method of forecasting demand is based solely on flying hours and past experience and thus is inconsistent with this hypothesis.

The scope of the study was limited to appraising the implications of the hypothesis as applied to forecasting demands for repair parts at the wholesale level. The referenced studies done on fixed wing jet aircraft gave various models to be evaluated while analyzing the data. These studies suggested other variables; some of these were also considered in this work.

2. Findings

The analysis was done primarily on aggregate information. The direction of the study was dictated by the availability of data and many of the suggested models developed by others could not be directly tested. By looking at worldwide (peacetime) and Vietnam (wartime) data, it was apparent that sorties for Army aircraft were quite a bit shorter in duration ** and less variable in absolute terms for both peacetime and wartime than those observed in earlier work done on fixed wing-jet aircraft. This short duration of sorties made it impossible to detect any failure time pattern (i.e. time to fail within a sortie).

Analysis of about 20 years of flying hour records (DA Form 1352) for five types of aircraft showed a consistently high correlation between sorties and flying hours. This finding was corroborated using three months of Sample Data Collection (SDC) data on Blackhawk helicopters collected during the Bright Star Exercise.

Comparison of peacetime and wartime data (1352 records) showed both a higher utilization rate (flying hours : hours available) and longer duration sorties for the wartime environment. Based on the referenced studies for fixed wing aircraft, both of these factors may cause lower failure rates per flying hour for wartime planning when one ignores combat produced damage.

Since little actual failure data could be obtained, several indirect methods were used to try to find the relationship between failures and sorties and between failures and flying hours. The 1352 records contain aircraft availability and other related data. The Bright Star data contain maintenance events and man-hours. Regressions of these variables against sorties and flying hours resulted in low correlations. The IRO wholesale demand data did show increased demands for increases in flying hours, but the causes of the increase in demands were confounded with wartime effects. Several other IRO studies indicated that the use of flying hour data improved the forecast for individual item demand at the wholesale level.

3. Conclusions

This study was not able to show any relationship between failures and flying hours, sorties, or utilization. The data were limited in that other causal effects could not be removed from the analysis, nor could the explanatory variables be controlled adequately to statistically quantify the study's findings. Other IRO studies did demonstrate the favorable use of flying hours in making demand forecasts and the lack of results of this study should not be construed as negating these findings. This report was written to document the work done even though there were no positive conclusions. It includes many graphs in the appendix depicting 20 years of 1352 data which may be useful in other research work.

CHAPTER I

INTRODUCTION

1.1 Background

The Army currently invests several hundreds of millions of dollars in safety level stock at the wholesale level to guard against the errors made when forecasting demand for repair parts in the supply system. In an effort to impact this investment, the Army Inventory Research Office (IRO) has conducted several research studies [4], [8], [11] with the intent of improving the forecast method used in the wholesale management inventory system known as the Commodity Command Standard System (CCSS).

These studies considered various demand models and forecast schemes and used the IRO demand data base for empirical investigation and evaluation. The IRO data base [6] includes 15 years of requisitions and demand for aviation parts by quarter accumulated from the Troop Support and Aviation Readiness Command (TSARCOM) Demand Return and Disposal (DRD) files from 1967 thru 1982. Actual flying hour history is also included in this data base.

In each study, program dependent forecast methods performed best (see [10] for evaluation methods) when evaluated using a simulated supply system. The program variable, flying hours, was applied to the data in a "straight line" fashion, i.e. a forecast for demand per flying hour was made, then this forecast was multiplied by the number of flying hours scheduled for the forecast horizon to determine the forecasted demand. This straight line method is what is currently being used in CCSS. Similarly the straight line assumption is being considered by the wartime planners, i.e. if we double flying hours during war, we will also double requirements for repair parts.

Maurice Shurman of the Boeing Corporation has shown [14], [15], using data from fixed wing, jet aircraft that this straight line assumption may be incorrect and that the effect of takeoffs should be considered. He claims that a large portion of the failures occur early in a flight and that these failures are caused by the stress of takeoff. Hence a unit flying short sorties (flying time between takeoffs) over a fixed flying hour program will generate more failures than a similar unit flying longer sorties for the same program.

Others, [1], [3], [13] have followed Shurman's lead and have supported his finding that the straight line assumption is incorrect. These studies

were all performed on fixed wing aircraft and may not apply to helicopters the way they are employed by the Army. Details of these findings will be cited in the next chapter.

1.2 Problem

It is felt that if Shurman's findings are applicable to Army aircraft (primarily helicopters) a sortie dependent model can be formulated and an improved demand forecast algorithm developed.

1.3 Objectives

The following is a list of objectives which describe the anticipated course of the study.

- a. Determine if data are available to measure the variability of sortie lengths. Flying hours per sortie is a major consideration of the study and should vary enough between time frames, aircraft, etc., to make the variable meaningful at the wholesale (aggregated) level of demand. Wartime vs peacetime scenarios should be considered.
- b. Determine if failure or demand data are available to relate failures to sortie length. Time to failure within a sortie would be the principal data element but other data types may be considered. Aggregate data may be used and experimental design layouts may be useful.
- c. Analyze the data and determine the relationship between failures and sorties. Consider the findings from cited reports in developing the relationship.
- d. Develop a forecast algorithm capturing the sortie/flying hour relationship.
- e. Add sortie length and/or the number of sorties flown to the IRO demand history file.
- f. Test the sortie dependent forecast model using the IRO demand history file comparing it with the current forecast method.
- g. Determine the feasibility of extending the results to other weapon systems.

1.4 Methodology

The methodology follows the list of objectives. First an intense literature review was done including site visits and telephone calls to a few of the referenced authors. Some of their findings are summarized in the next

chapter. Next several discussions were held with TSARCOM and with the Reliability, Availability and Maintainability Division of AMSAA to identify availability of data.

At the outset it was apparent that helicopter sorties were quite a bit shorter in duration than those flown by fixed wing Army aircraft. To get a better feel for the magnitude and variability of helicopter sorties a request was made to TSARCOM to derive quarterly statistics on five aircraft (OV-1, AH-1, CH-47, UH-1, OH-58) from the 1352 flying hour report. A description and analysis of the data are found in Chapter III. Since this was the first time a historical survey was made of the 1352 data, all the time series graphs made from the data are included in Appendix C. This data represents actual field usage information as opposed to developmental data which is commonly used for reliability estimation.

The 1352 analysis showed that, in the aggregate, sorties were highly correlated with flying hours. The direction of the study was then shifted to determine a failure rate/flying hour relationship. Chapter III describes the attempts at relating 1352 reliability/availability statistics to flying hours.

A continuation of the failure rate analysis is given in Chapter IV where Sample Data Collection data and demand data were considered. Alternative causal variables are also discussed in this chapter. The conclusions and recommendations are found in Chapter V.

CHAPTER II

RELATED STUDIES

2.1 Inventory Research Office

The Army as well as the other Services have in the past assumed a direct proportional relationship between component failures for aircraft and flying hours. This relationship has been indirectly tested by IRO in several previously mentioned studies by comparing forecast algorithms with and without program (flying hour) adjustments using actual demand and flying hour history. In each case the program dependent algorithm worked best. Additional attendate IRO to statistically compare flying hours with demands using regression techniques have not been very successful due to a problem of relating dema to specific usage and aircraft.

2.2 Boeing

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Maurice Shurman of the Boeing Corporation [13-15] studied failure data on various fixed wing jet aircraft. His initial work included time to failure analysis on six aircraft (F-106, E-3A, B-52, F-4, C-141, 707) where he noted that a high percentage of the failures occurred early in the flight or sortie. By plotting these data with a common format relating percent of total failures per sortie against time to fail as a percent of the nominal sortie length for each system he generated a family of curves whose bounds were similar to those shown in Figure 1.

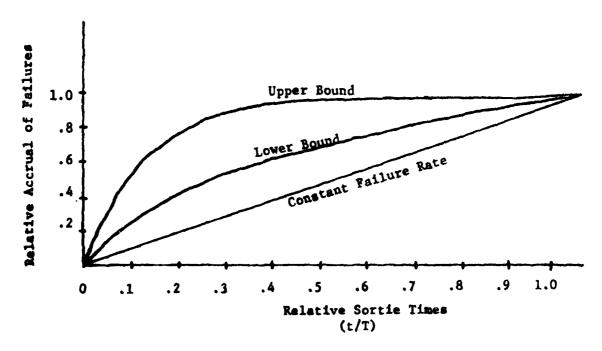


Figure 1. Common Plot

Using these curves he generated reliability equations. He confirmed his findings with other jet aircraft including the C5-A, C141-A.

His results indicate that sortie length should be considered when developing failure estimates, i.e., an increase in sorties per flying hour would result in an increase in failures per flying hour. This conjecture if true for Army helicopters would impact many areas such as contingency planning, reliability estimates and demand forecasting.

2.3 Air Force

COL Chris Shaw of the Air Force [13], in his analysis of C-5, C-130, C-141, and P-3C aircraft, claims that both sorties and flying hours affect failures. His model combines the failures due to start up, landing, and steady state flight in the following linear manner:

$$Y = A + B (X-1) + C$$

where

Y = failures per sortie

A = failures due to start up of the sortie

B = steady state failure rate during cruise (failures per flying hour)

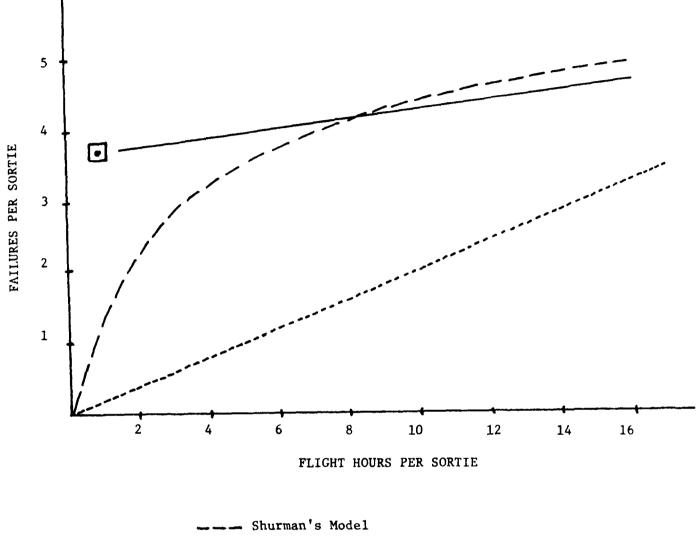
C = failures due to landing of the sortie

X = sortie length in hours

He confirmed his model by plotting and regressing average failures per sortic against sortic length. He deleted short sortics (less than one hour) from the regression and made a point estimate of these observations. His data did show a non zero intercept indicating the sortic effect. Figure 2 compares Shaw's finding with Shurman's.

2.4 Institute for Defense Analyses

P. B. Buck of the Institute for Defense Analyses [1] has compared maintenance man-hours to utilization rates for the 6th and 7th fleets for the A-6E, A-7E, F-4J, and F-14 aircraft. He used multiple regression analysis which included the explanatory variables: utilization, flying hours, sorties, and dummy variables indicating different groups. His preliminary findings indicate that utilization rate (hours flown/hours available) has a much greater impact on failures than do sorties. This relationship is hyperbolic in function with failures per flying hour decreasing with an increase in flying hours per month for the aircraft. Hence the assumption of a constant failure rate (failures per flying hour) is incorrect



---- Shurman's Model
----- Constant Failure Rate Model
----- Shaw's Model
Short Sorties

Figure 2. Comparison of Failure Models

and should be adjusted to reflect the utilization of the aircraft. This implies that a small peacetime utilization would generate a larger failure rate estimate (failures per flying hour) than would be expected during wartime where the utilization would be greater.

2.5 Others

Linda Cavalluzzo of the Center for Naval Analyses [3] has demonstrated in her preliminary work using F14-A data that the number of sorties and the number of flying hours flown in given periods were highly correlated. She developed a linear relationship between failures and flying hours using an econometric method (two stage least square). (This method eliminates the secondary dependency that failures have on flying hours.) Her equations show a more severe slope on flying hour (approximately 1) than did the normal least squares estimate.

- H. Campbell of the Rand Corporation [2] considered seven other operational variables and concluded that only flying hours should be used to explain demand variation.
- T. S. Donaldson and A. F. Sweetland of the Rand Corporation [5] studied the relationship between maintenance manhours and aircraft flying hours. Their findings were mixed across the various systems studied and the analysis was confounded by mission type variables.
- G. Walker, D. Wilson, and D. Hindes [19] of the Boeing Corporation did an extensive study for the Air Force Human Resources Laboratory where they estimated the maintenance impacts of various characteristic parameters of the design, operational, and natural environments of the subsystems of various aircraft. Separate models were developed using multiple regression techniques for 29 equipment types. Thirteen of these models included a sortic variable.

Martin Cohen in [4] cites other references, all who have had various degrees of success in relating failures to operational variables.

2.6 Model Consideration

The work done by Shurman, Shaw, Buck, and IRO had the major impact on the conduct of the study. Each model was indirectly considered and each explanatory variable was investigated. The mode of evaluation was dictated by the availability of the data and the conclusions were somewhat subjective using data analysis inferences.

CHAPTER III

SORTIE VARIABILITY

3.1 Basic Questions

Since the referenced studies relating sorties to failures were all done using fixed wing aircraft data, a preliminary analysis was done to investigate the statistical characteristics of sorties as applied to helicopters and other Army aircraft. The basic questions of concern were as follows:

- a. How long are typical sorties for helicopters as compared to fixed wing aircraft?
 - b. Does sortie length vary between aircraft, time, units, and scenario?
- c. Is sorties variation sufficiently large to consider it as an explanatory variable?
 - d. Is there a difference between wartime and peacetime sortie lengths?
 - e. Are sorties correlated with flying hours?
- f. Is utilization (flying hours flown/hours available) correlated with flying hours and/or sorties?
 - g. Is reliability or maintainability affected by sorties?

3.2 Data Description

To help answer these basic data questions, TSARCOM went into their archives and retrieved "1352 - Army Aircraft Inventory Status and Flying Time" records from the early 1960s to 1982 for five systems: AH-1, CH-47, OH-58, OV-1, UH-1. (Three years of UH-60 Blackhawk data were also collected.) These data are collected routinely on a monthly basis for each tail number by unit at organization level. The data contain flying hour and sortic information as well as maintenance and availability statistics. Appendix A contains sketches of these aircraft.

Since the study deals mainly with failures (demands) as seen at the wholesale level, the data were collected in terms of quarterly aggregate statistics which are described in the next section. During the war years, the data were separated into Vietnam and Non-Vietnam (worldwide) categories.

3.3 Data Variables

To determine a profile of usage for each system, 16 variables or statistics were computed for each quarter of available data. These statistics give the number of system and units fielded, the gross number of flying hours and sorties,

the average and standard deviation of usage (utilization) per aircraft (tail number), the average and standard deviation of sortie length, and attempts at reliability and availability measurements. The definition of each variable is found in Appendix B.

3.4 Time Series Analysis

The initial analysis consisted of time series plots of each variable with Vietnam data superimposed on the world wide (excluding Vietnam) series. These graphs are found in Appendix C and will be referred to as "time series plots" throughout this report. Averages and standard deviations of the plotted (non-zero) quarters are included for both Vietnam and worldwide data.

The following is a list of observations gleaned from the time series statistics:

a. The average and coefficient of variation of the two sortie length variables (average duration of sorties, average duration of sorties by aircraft - refer to Appendix B for definition) over the time series are given in Table 1.

TABLE 1. SORTIE DURATION (HRS)

		AH	-1	CI	H-42	OI	H-58	70	<i>I</i> -1	τ	JH-1
		AVG	COEF-V	AVG	COEF-V	AVG	COEF-V	AVG	COEF-V	AVG	COEF-V
AVERAGE DURATION OF	WORLDWIDE	.33	.20	.30	.19	.33	.16	.86	.14	.32	.16
SORTIES	VIETNAM	.57	.23	.34	.12	.38	.28	1.66	.15	. 34	.12
AVFRAGE DURATION OF	WORLDWIDE	.30	.35	.31	.19	.37	.76	.79	.21	.34	.20
SORTIES BY AIRCRAFT	VIETNAM	.56	.32	.34	.30	.39	.32	1.56	5 .20	.38	.15

From these data we see that the sortie durations are quite a bit shorter for the Army aircraft than those experienced for the fixed wing/jet aircraft cited in the referenced studies. The coefficient of variation (ratio of standard deviation to average) is approximately .20 between the quarterly averages which give a range of \pm 60 percent (3 Std. Dev.) of the reported average of averages. Thus for a typical average sortie length average of .30 hours, the actual quarterly average may range from .12 to .48 hours. For a given 1000 flying hour program, the actual range of the number of sorties per 1000 hours may be between 2083 and 5555 and may

have significant impact on the number of failures per flying hour. This table also points out that the average sortie during Vietnam was longer for each aircraft than that experienced on the Non-Vietnam worldwide average and quite a bit larger for the AH-1 and OV-1.

b. Table 2 compares the average quarterly (averaged over all non-zero quarters) flying hour program for each type of aircraft. These data are non stationary and follow trend lines and hence maximum values will be given with each average as opposed to standard deviations. Minimums are not given since they are quite low and indicate the initial build up of density.

TABLE 2. FLYING HOURS (PER 3 MONTHS)

	AH-1		CH-47		OH-58		0V-1		UH-	L
	AVG	MAX	AVG	MAX	AVG	MAX	AVG	MAX	AVG	MAX
WORLDWIDE	13,908	35,000	10,517	16,000	60,913	120,000	5,706	7,000	166,923	250,000
VIETNAM	46,686	90,000	31,661	60,000	21,943	45,000	10,301	20,000	298,648	500,000

Except for the OH-58, the hours flown in Vietnam are two to three times greater than those flown during worldwide non Vietnam wartime missions. Thus if the straight line assumption about failures and flying hours is correct, then we will need two to three times more spare parts to support a conflict similar to Vietnam than to support peacetime activities. Note also the maximum Vietnam flying hours were on the magnitude of from four to six times the peacetime average.

c. The flying hour program reflects to some extent the density of items fielded. The density comparisons are shown in Table 3. Again maximum values are given instead of standard deviations.

TABLE 3. DENSITY

	AH-1		CH-47 OH-58		-58	ov	UH-1				
	AVG	MAX	AVG	MAX	AVG	MAX	AVG	MAX	AVG	MAX	
WORLDWIDE	569	1100	296	500	1569	2300	151	230	2668	4800	
VIETNAM	317	560	227	325	159	300	63	100	1539	2500	

The averages are somewhat misleading due to the density buildup over time. The comparison of maximums can be used to determine how much of the total density was fielded in Vietnam during its peak period. (Note World maximum occurred after the Vietnam war and reflects total density of equipment.) Hence for the AH-1, CH-47,

OV-1, and UH-1 approximately 50 percent of the total density was deployed in Vietnam at the peak but these aircraft flew three to six times the normal peacetime flying hour program. (Note:worldwide does not include Vietnam during the war years.)

d. Usage or utilization rate combines the flying hour program with density. It compares by tail numbers the total hours reported on the unit's property book with the number of hours flown (ratio) during the same period. Thus from the last two remarks it is not surprising to see that the Vietnam utilization is much greater than peacetime. Table 4 shows these differences.

TABLE 4. UTILIZATION - USAGE

	AH	-1	CH-47		ОН	OH-58		ov-1		OH-1	
	AVC	MAX	AVG	MAX	AVG	MAX	AVG	MAX	AVG	MAX	
WORLDWIDE	.01	.02	.02	.03	.02	.07	.02	.03	.03	.05	
VIETNAM	.07	.10	.06	.09	.07	.08	.07	.10	.09	.11	

The peacetime utilization is relatively constant at about 2 percent but the Vietnam data reflected a build up from 6 percent to about 10 percent. If Buck's findings are true for Army aircraft then the failures per flying hour rate for Vietnam (excluding combat damage) should be much smaller than those experienced during peacetime. Unfortunately we don't have data to test this hypothesis.

e. Even though our main interest in the study dealt with aggregate variability we also measured the variability between aircraft (tail numbers) by computing standard deviations within each quarter for a few of the variables. The time series of these standard deviations are found in Appendix C. Table 5a relates the average standard deviation to the estimated overall mean by giving the coefficient of variation for each system for usage and sortie duration.

TABLE 5a

AVERAGE COEFFICIENT OF VARIATION OF USAGE

	AH-1	CH-47	OH-58	0V-1	UH-1
WORLDWIDE	2.00	1.00	1.00	1.00	1.00
VIETNAM	.60	.70	1.60	.60	.55
AVERAGE	COEFF	ICIENT OF	VARIATION OF	SORTIE	DURATION
WORLDWIDE	1.20	.93	1.00	.90	1.0
VIETNAM	.67	1.1	.60	.50	.76

These high coefficients of variation indicate that the individual helicopter's experience may be quite different from that reported by the quarterly averages. Presently the variability within the aggregate is not considered when forecasting wholesale demands. If this variability as shown in the table is attributed to causal differences such as unit, mission, geography, then forecasting separately for homogeneous subgroups may be worth considering.

3.5 Regression Analysis

The analysis of the time series data indicated that utilization, flying hours, and sortic lengths exhibit enough variation to be considered as explanatory variables when predicting failures. Since these variables are associated with each other, regression analysis (ordinary least squares) was performed to determine if these variables are sufficiently pairwise independent to be considered simultaneously in a failure rate model. The author considers for this study variables with correlation coefficients of greater than .85 to be statistically the same (i.e. each variable contains the same information). The regressions considered are as follows:

Total Sorties vs Total Flying Hours Total Flying Hours vs Density Average Use vs Total Flying Hours

The scatter diagrams of these data are found in Appendix D and E. Table 5.b. contains the estimated correlation coefficients. It should be noted that all lines were fitted with an assumed zero intercept.

TABLE 5.b. ESTIMATED CORRELATION COEFFICIENTS

		AH-1	CH-47	OH-58	OV-1	UH-1
SORTIES VS	WORLDWIDE	.92	.86	.98	.86	.88
FLYING HRS	VIETNAM	.97	.97	.99	.98	.98
FLYING HRS VS	WORLDWIDE	.87	.65	.82	.38	.61
DENSITY	VIETNAM	.96	.95	.97	.95	.98
USAGE	WORLDWIDE	43	05	09	.61	.06
FLYING HRS	VIETNAM	.94	.91	.73	.56	.93

The following is a list of inferences drawn from the analysis.

a. The total number of sorties flown is highly correlated with total hours flown for the data sets considered. The estimated slopes of the regression lines are given in Table 6.

TABLE 6. SLOPE ESTIMATES SORTIES VS FLYING HOURS

	AH-1	CH-47	OH-58	ov-1	UH-1
WORLDWIDE	2.9	3.2	2.9	1.2	3.1
VIETNAM	1.6	3.0	2.8	.57	3.0

This table is consistent with Table 1 where the average sortie lengths were compared. (Note: Total Sorties = Slope x Total Flying Hours

Avg Sortie Length = 1 : Slope)

Longer duration sorties are flown during wartime conditions particularly for the AH-1 and OV-1. Within each class sorties and flying nours are the same predictor variables due to the high correlations but the effects of flying hours on failures may be different between the data classes because of the various slopes found between the classes. Shurman's model would project different failure rates (per flying hour) for those cells with different slopes.

b. Density is currently being used as a program factor in forecasting failures for other than aircraft systems. Appendix D shows a high correlation within each data class between total flying hours and density. The estimated slopes of the regression lines are given in Table 7.

TABLE 7. SLOPE ESTIMATES FLYING HOURS VS DENSITY

	AH-1	CH-47	он-58	o v-1	UH-1
WORLDWIDE	23	30	37	32	51
VIETNAM	156	140	143	150	182

There are extreme differences between the slopes for worldwide and Vietnam data. For the AH-1 we would expect to fly seven times the number of peacetime hours for the same density. Hence if the straight time flying hour model is correct, but density were used instead of flying hours to forecast demand, the wartime estimate would be too low.

c. Utilization did not correlate well with flying hours using worldwide data; the plot was nearly horizontal. The Vietnam data did show a higher correlation which probably reflects the different intensities of the war. Table 4 from the previous section shows the differences in utilization between the data cells. Utilization as a variable may capture the effect of a wartime scenario and should be considered in the failure rate model.

In an attempt to get a picture of failures from the 1352 data, quarterly availability and reliability variables as defined in Appendix B were computed from the maintenance and supply down times. The reliability (inherent availability) variable measures the percentage of time each aircraft would have been available if only maintenance time was considered when computing down time. Similarly the availability variable measures the percentage of time each aircraft was actually available with both maintenance and supply down time being considered.

These failure related measures were regressed against total flying hours, sortie length and usage. The scattergrams and statistics are included in Appendix E. The OV-1 was the only fixed wing aircraft in the analysis and was also the only aircraft with a consistent (but relatively low) positive correlation between the failure measures and both usage and sortie length. Hence it is questionable whether the findings from the fixed wing aircraft studies are applicable to helicopters.

Additional regression analysis was done between sortie length vs flying hours, sortie length vs usage, and usage vs flying hours. These statistics are found in Appendix F. In each case the correlations were low for all the data sets implying the pair wise independence between these variables.

3.6 Findings

Within each data subset of the 1352 analysis (system and scenario) the number of sorties flown was highly correlated with the total flying hours (statistically this implies the two variables contain the same information). The estimated slopes which reflect sortie length did differ between a few of the data cells and this difference should be considered in the model building as a cell effect.

The other variables, sortie length and usage, were shown to be individually uncorrelated with flying hours and with each other. These variables could be included in the model without causing any multi collinearity problems. (Note sortie length and cell effect are related and including both may cause problems with the model structure).

Failures in terms of reliability and availability as defined in this study did not correlate well with the explanatory variables, and this lack of correlation raises questions about the credibility of a failure rate model.

Additional failure rate analysis is given in the next chapter.

Sortie length should not be confused with the number of sorties which was evaluated in Section 3.4.

CHAPTER IV

FAILURE RATE ANALYSIS

4.1 Introduction

Since the model would be used at the wholesale level, aggregate failures were of main concern. The 1352 data files contain maintenance down time but this included scheduled maintenance which is done on a flying hour basis and would confound any analysis done with it. Other aggregate data sources were not found.

Another source of data was Sample Data Collection (SDC). This might be considered micro data in that it gives historical information on a few selected aircraft from particular units. The data contain only unscheduled maintenance along with monthly flying hours and sorties. It does not contain the aggregate dimension of variability that would be needed for an experimental design analysis. An analysis of SDC, Blackhawk, data is given in Section 2 of this chapter.

An indirect way of measuring aggregate failures is to look at the demands for repair parts at the wholesale level. The IRO has a history of this wholesale data for Army aircraft. The data were of limited use because of the problems encountered when trying to relate the demands to specific aircraft types. Further discussion and analysis of this data is given in Section 3.

In discussing this project with aircraft engineers, maintenance personnel, and data collectors, it was of consistent opinion that sorties do have some effect on failures. They also mentioned many other causal variables that they felt may have a greater impact. A synopsis of these discussions is given in Section 4.

Section 5 contains a subjective summary of the failure rate analysis.

4.2 Sample Data Collection

In an effort to determine if SDC data as "Micro data" could contribute to the "macro analysis," data collected during the Bright Star Exercise were analyzed. The data consisted of monthly non-scheduled maintenance events, maintenance hours, flying hours and sorties for each of fourteen Blackhawk helicopters for three months yielding approximately 42 observations. The analysis was similar to the regression experiments done with the 1352 data.

The scattergram of the number of sorties against flying hours is found in Appendix G. The correlation was high and the slope of 3.7 was of the same gross

magnitude as the slopes observed (worldwide - peacetime) with the 1352 data. Appendix G also contains scattergrams made in an attempt to relate failures to sorties or flying hours. The variables considered were: events per flying hour, number of maintenance events, and total maintenance hours. In each case the correlation of the variable with sorties and flying hours was extremely low. This analysis does not support the failure rate/flying hour - sortie model.

Additional work was done with quarterly SDC data taken from the summary reports. The analysis not presented here gave additional evidence that sortics are correlated with flying hours with a slope of approximately 1/3. Again there was no evidence to support the failure rate vs flying hour or sortic model.

4.3 IRO Demand Data

Since the results of the study are to be applied to demand forecasting, the most natural data base for the failure rate analysis is the IRO TSARCOM demand data file. As mentioned in Chapter 1, these data have been used in previous forecasting studies where the results indicated the favorable use of flying hours when predicting demand for replacement parts.

The file consists of quarterly demands for approximately 22,000 items from the years 1967 to 1982. Martin Cohen [4] describes the first use of the data base and gives a subjective analysis of the relationship between flying hours and demands. In [6], Sally Frazza gives a comprehensive description of the data along with the processing decisions that had to be made in creating the data file.

There are several problems with the data that precludes its use for model building. The 1352 data indicated that the biggest difference in sortic length occurred between the wartime and peacetime periods. The demand file also shows a substantial difference in demands but these demands are for both combat damage and failures hence confounding the analysis. Another consideration is that the data contain many "common items," i.e. items that are common to several systems. Identification of demand to a particular system is not given in the data base. In Cohen's study he picked selected items unique to particular systems and with these groups of items showed a definite relationship between demands and flying hours.

Figure 3 shows a time series of aggregate demand for 12,148 actually demanded items from the years 1967 to 1977. This compares the magnitude of demands between the peacetime and wartime periods. Since the demands during war were three to four times that during peacetime, these data do not conflict with the failure rate model but further detailed analysis and model building is not possible due to the previously mentioned constraints of the data. Thus the best we can say is that the demand data file is consistent with the flying hour/sortic models but does not support rigorous model testing experiments.

4.4 Other Variables

In discussing this project with Army helicopter experts (engineers, pilots, maintenance experts, data collectors) it became quite apparent that other variables would have to be considered when evaluating the effects of sorties and flying hours on failures.

First off, the notion of sortie would have to be qualified by the types of takeoffs and landings. Takeoffs may be classified in the following ways:

- a. Vertical and angular (to various degrees).
- b. Full power and constrained power.
- c. Maximum payload and minimum payload (crew and fuel).
- d. Cold startup and continuation of mission.

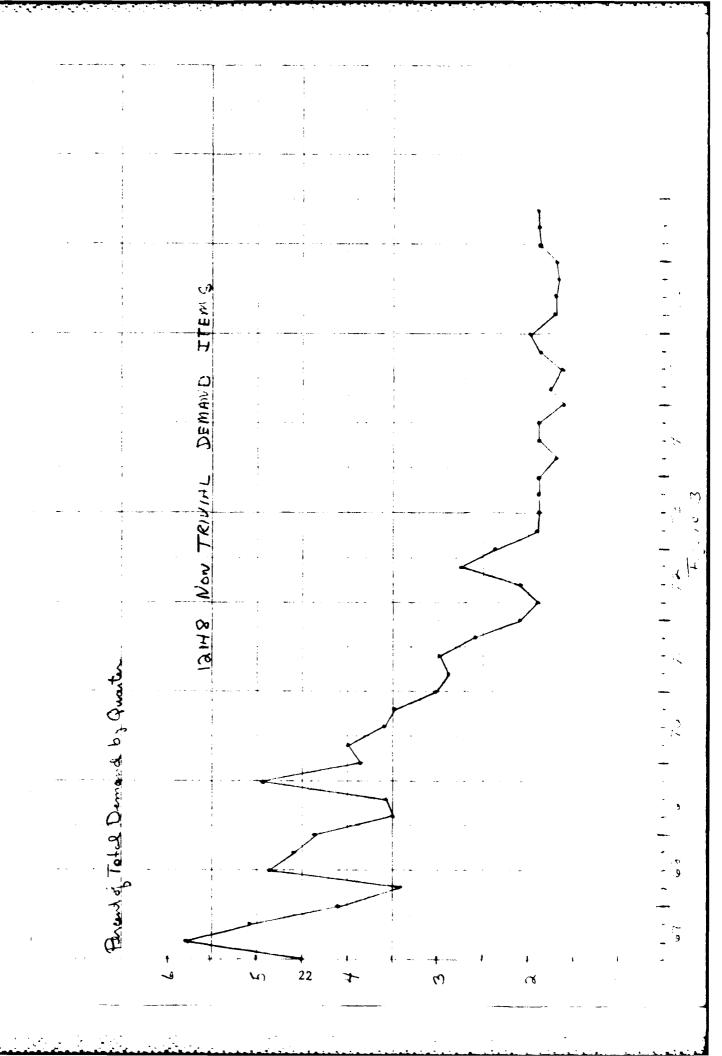
Similarly landings may be classified as:

- a. Autorotations (emergency landings with the engines stalled these are practiced as part of the training program).
- b. Hovering (where the aircraft doesn't quite touch the ground but stays in one location for loading or unloading).
 - c. Running (where the aircraft comes in from less than vertical descent).

Flying hours would also have to be classified according to the types of mission as follows:

- a. Nappe of the Earth (where the aircraft flies at tree level with extreme power surges to lift over obstacles).
 - b. Contour/low level flight (various altitudes).
 - c. Weight variation.
 - d. Special considerations.

Other conditions would affect failures. These include geographic conditions such as: sand, alkaline, ice, and snow, along with environmental conditions such as moisture, rain, heat, cold, and volcanic ash.



Subsystems react differently to the mentioned classes (e.g. nappe of the earth flying would impact the engine system differently than it would the electrical systems) and should be considered separately when developing a failure rate model. The failures themselves are inherently different. They may be caused by age, malfunction, accident, improper use, stress, combat damage, or neglect. Each of these factors may confound the effects of sorties and flying hours and should be appraised during the analysis.

Another problem in dealing with failures is the association of the failure with a specific sortie or mission. Very seldom are failures detected in flight; most of them are observed during inspection (daily, pre-flight, post-flight) or during routine or scheduled maintenance. In either case the specific time into a flight or sortie of a failure is not observed and is lost for the analysis. Hence stress of takeoff and landing cannot be directly measured.

The mentioned considerations lend themselves to a classical experimental design where each of the factors would be controlled or blocked during the experiment. Such an experiment would be extremely expensive and is beyond the scope of this study. The lack of such a control of the environment precludes the development of a failure rate model per se and puts emphasis on developing general relationships gleaned subjectively from the aggregate data. The next section summarizes these subjective observations concerning failures.

4.5 Findings

As mentioned in the previous section there are many variables that may affect failures in addition to sorties and flying hours. Many of these are related to the variable in question and their effects would be difficult to differentiate without an extensive experimental design effort. Since the results of this study are to be applied to demand forecasting at the wholesale level, the environmental variables in terms of the aggregation may not vary much over time except possibly between peacetime and wartime periods.

The 1352 failure data (reliability/availability) as analyzed in the previous chapter did not correlate well with sorties, flying hours, or utilization. Similar analysis using SDC data for a single unit resulted in the same finding.

The IRO demand history file showed more demands during the Vietnam era than during peacetime. This corresponded to an increased flying hour program but the relationship was confounded due to the fact that many of the demands were generated from combat damage and other wartime conditions not directly related to flying hours.

CHAPTER V

CONCLUSIONS/RECOMMENDATIONS

5.1 Summary of Findings

the model.

The analysis presented in this study was done primarily on aggregate information. The direction of the study was dictated by the availability of data and many of the suggested models developed by others could not be directly tested. By looking at worldwide (peacetime) and Vietnam (wartime) data it was apparent that sorties for Army aircraft were quite a bit shorter in duration and less variable in absolute terms than those observed in earlier work done on fixed wing jet aircraft. This short duration of helicopter sorties made it impossible to detect any failure time pattern as demonstrated by Shurman.

Analysis of about 20 years of 1352 flying hour records for five aircraft consistently showed a high correlation between the number of sorties and flying hours. This finding was corroborated using three months of SDC data on Blackhawk helicopters collected during the Bright Star exercise.

The correlation analysis of the 1352 data was done separately for the wartime and peacetime data. For each of the four helicopters tested, the slope was estimated as approximately 3 (assuming a zero intercept, Slope = number of sorties flying hours) except for the AH-1 wartime data where the slope was estimated as 1.6 (longer sorties during wartime). For the OV-1 (the only fixed wing aircraft) the slope was 1.2 for peacetime and .57 for war also indicating longer sorties during war. These differences in slopes between systems and scenario indicate that if sorties do have an impact on failures then a different flying hour model should be used for different aircraft and scenarios, or sorties should be included in

Another difference seen between the wartime and peacetime data was utilization. The utilization rate for wartime was three to six times that of peacetime. This difference in utilization should be considered for contingency planning.

Due to the paucity of failure data per se, several indirect methods of relating failures to flying hours and sorties were investigated. Using both 1352 reliability and availability statistics, there was no correlation found between these variables and sorties, flying hours or utilization. Similar findings were demonstrated using the SDC Bright Star data where maintenance events and maintenance manhours were regressed against sorties and flying hours. The IRO wholesale demand data did show increased demands for increases in flying hours but the

causes of increase in demand were confounded by wartime effects. Several other IRO studies indicated that the use of flying hour data improved the forecast for individual item demand at the wholesale level.

During the course of the study several Army aircraft experts were contacted and they all suggested that other factors such as mission, environment, and geographic location may have a greater effect on failures than do sorties and flying hours. The Boeing study [19] addressed these factors using Air Force data. These results suggested 29 different models to be used separately for different high failure equipment. Each model considered (multiple regression) as many as 54 explanatory variables. Results of this complexity would not be useful in the context of forecasting demand at the wholesale level. No additional analysis was performed on this topic.

5.2 Conclusions

This study was not able to show any relationship between failures and flying hours, sorties, or utilization. The data were limited in that other causal effects could not be removed from the analysis nor could the explanatory variables be controlled adequately to statistically quantify the study's findings. Other IRO studies did demonstrate the favorable use of flying hours in making demand forecasts and the lack of results of this study should not be construed as negating these findings.

5.3 Recommendations

The following is a list of recommendations based on the findings of this study:

- a. Until stronger evidence is found to support the study findings, continue to use flying hours as a program variable for demand forecasting.
- b. Continue to search for failure data where sortie, flying hour, and utilization effects can be statistically appraised and a failure rate model developed.
- c. Contingency planners should be aware of the large differences in utilization observed between the wartime and peacetime data. Additional work should be considered where a usage model similar to Buck's [1] could be studied.

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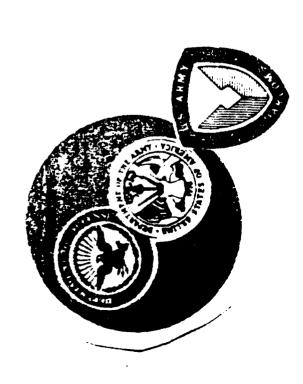
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APPENDIX A AIRCRAFT USED IN STUDY

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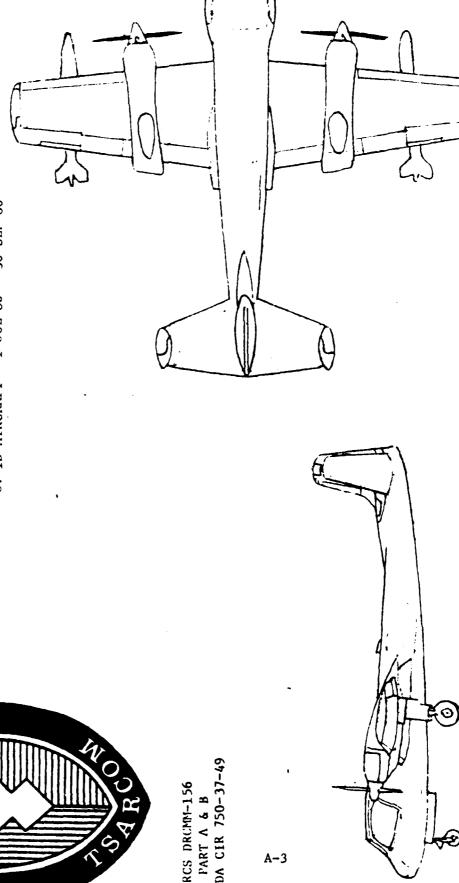
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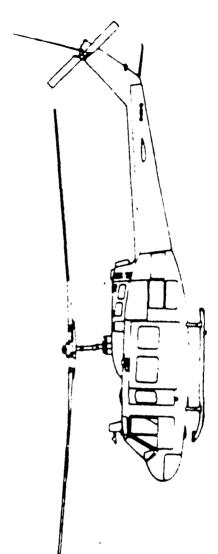
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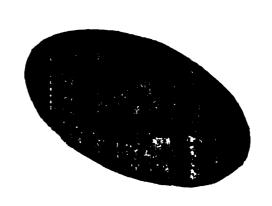
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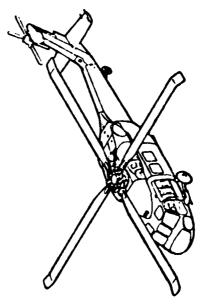
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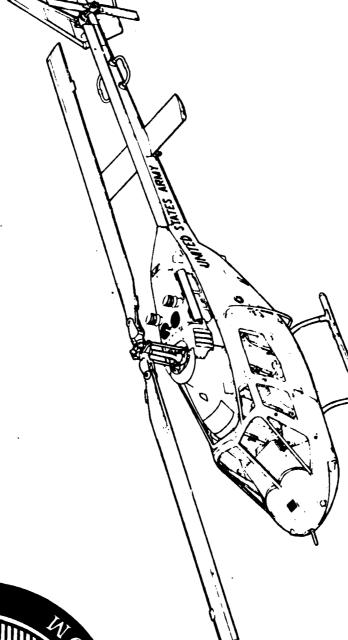
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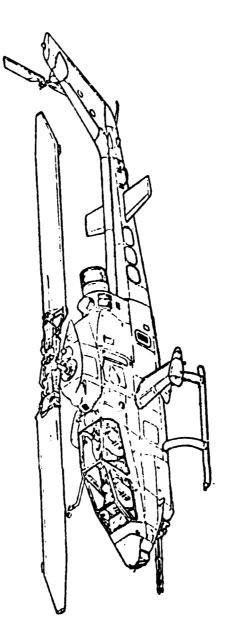
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APPENDIX B

DEFINITION - 1352 VARIABLES

	Data Elements	<u>Definitions</u>
a.	Density	The number of aircraft reporting data within the quarter.
b.	Number of Reporting Units	The number of units reporting data within quarter.
c.	Total Flying Hours	The sum of all hours flown within the quarter.
d.	Avg Hours On Hand	The average hours on hand per aircraft for for the quarter.
e.	Standard Deviation of Hours On Hand	Standard Deviation of (d).
f.	Avg Usage	The average (flying hours + hours on hand) per aircraft for the quarter.
g.	Standard Deviation of Usage	Standard Deviation of (f).
h.	Total Number of Sorties	The sum of all landings reported within the quarter.
1.	Avg Length of Sortie(Duration	c : h (Total Flying Hours : Total Sorties)*
j.	Avg Length of Sortie by Aircraft (Duration)	The average (flying hours : number of landings) per aircraft for the quarter. **
k.	Standard Deviat on of Length of Sortie by Aircraft	The standard deviation of (j).
L.	Avg Reliability	The average ((hours on hand - hours not mission capable due to maintenance) : hours on hand) per aircraft for the quarter.
m.	Std Dev of Reliability	Standard deviction of (L).
n.	Avg Availability	The average ((partial mission capable + full mission capable; hrs on hand) per aircraft for the quarter.
٥.	Standard Deviation of Availability	Standard deviation of (n).
p.	Number of Autorotations	The sum of all autorotations reported within the quarter.

^{*}Fleet Ratio

^{**} Average Ratio of Each Tail Number

APPENDIX C

1352 DATA TIME SERIES

Variables

Density

Number of Reporting Units

Total Flying Hours

Avg Hours on Hand

Standard Deviation of Hours On Hand

Avg Usage

Standard Deviation of Usage

Total Number of Sorties

Avg Length of Sortie

Avg Length of Sortie by Aircraft

Standard Deviation of Length of Sortie by Aircraft

Avg Reliability

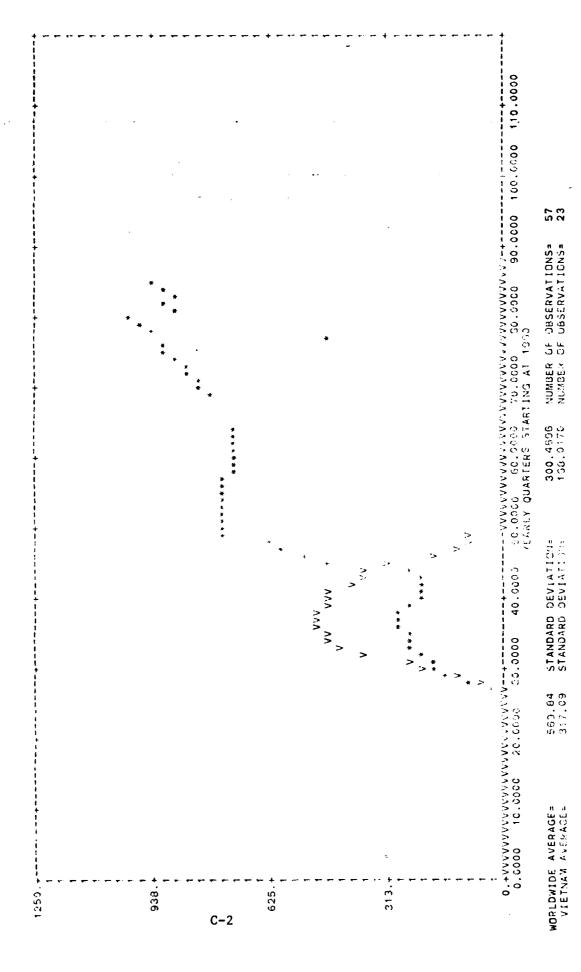
Std Dev of Reliability

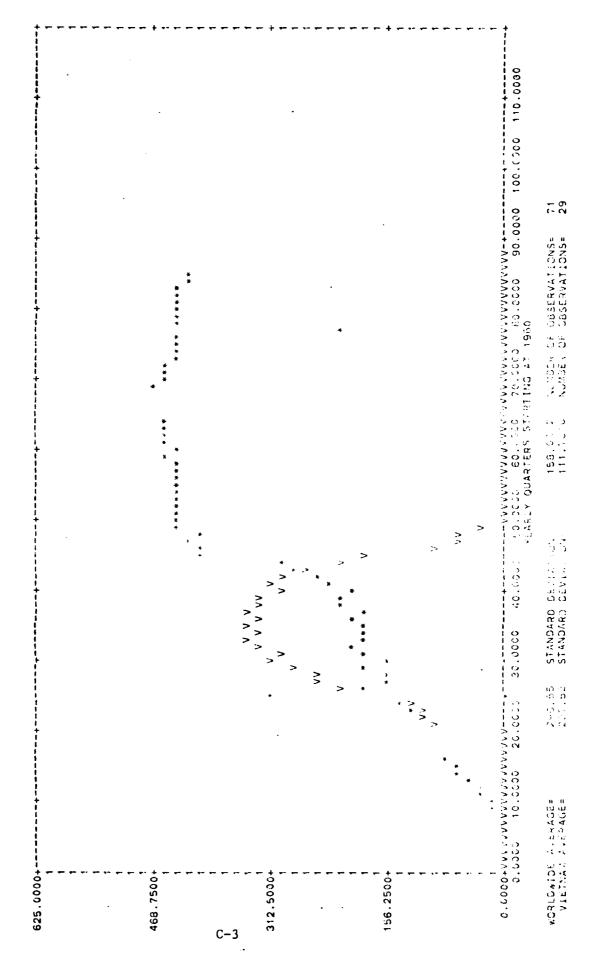
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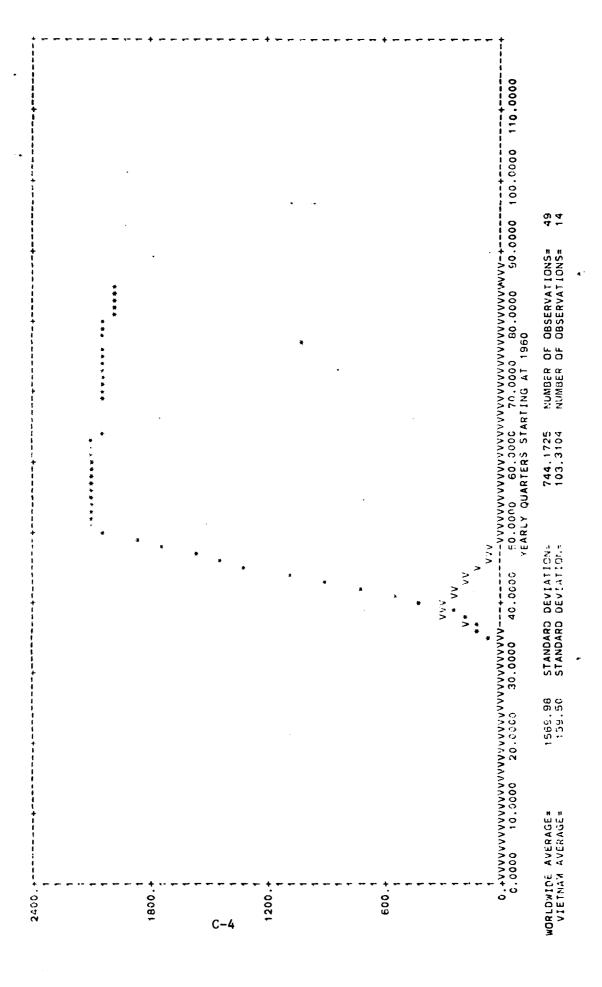
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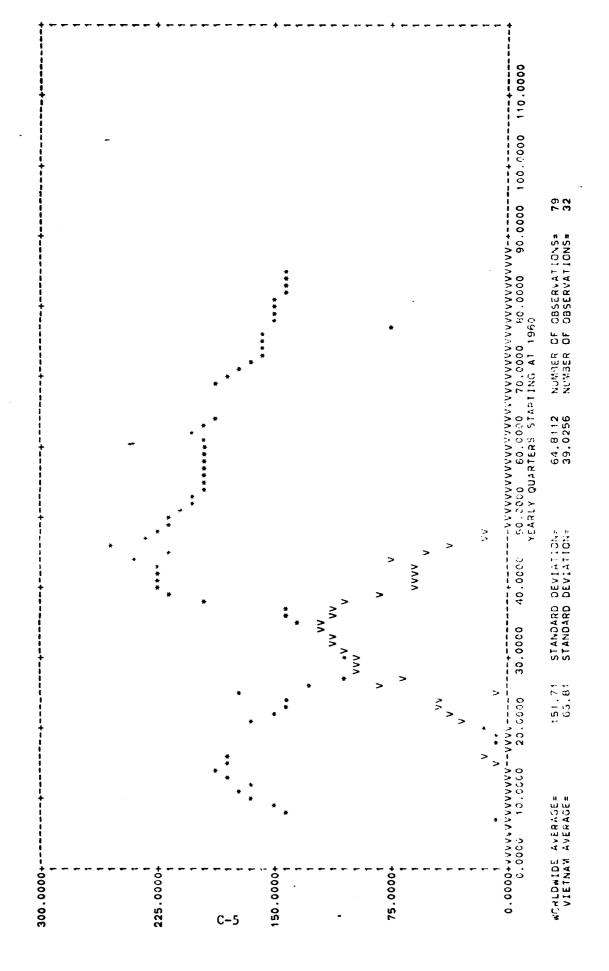
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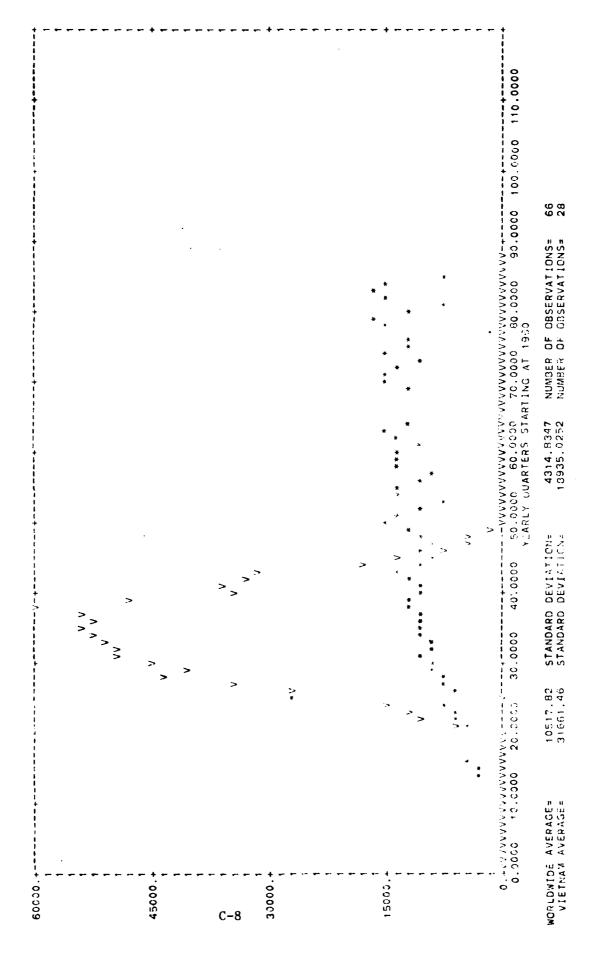




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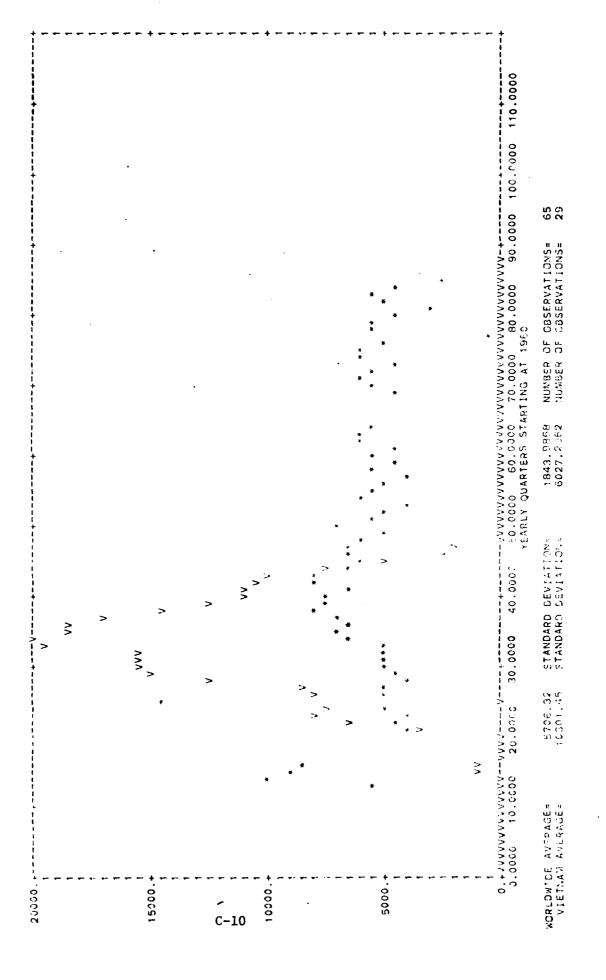
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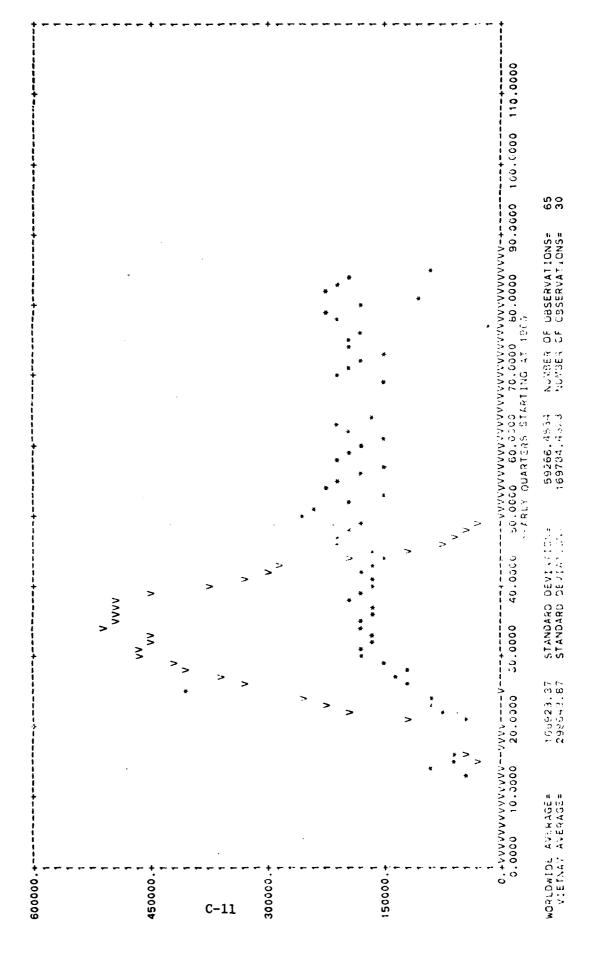
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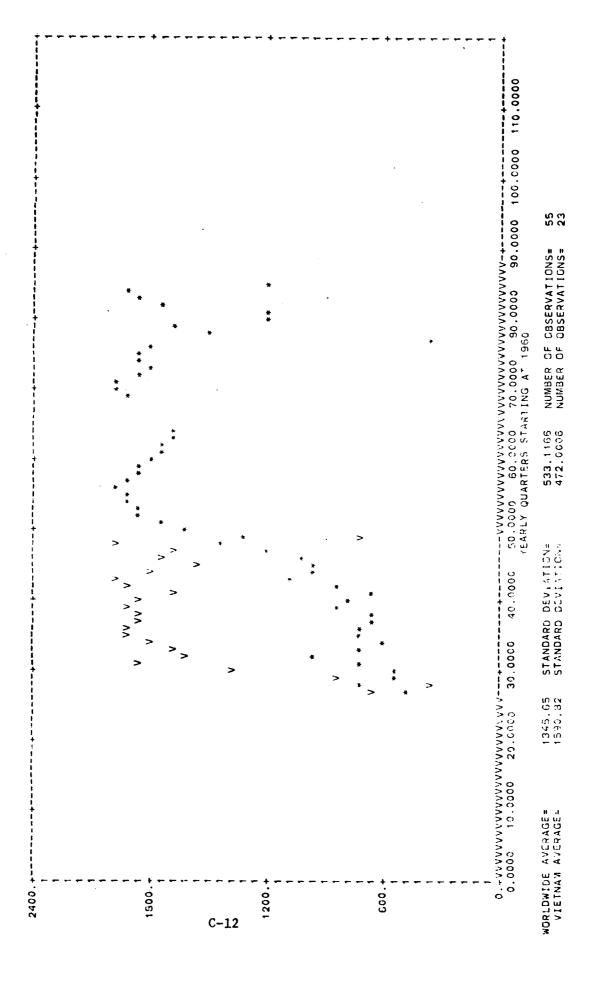


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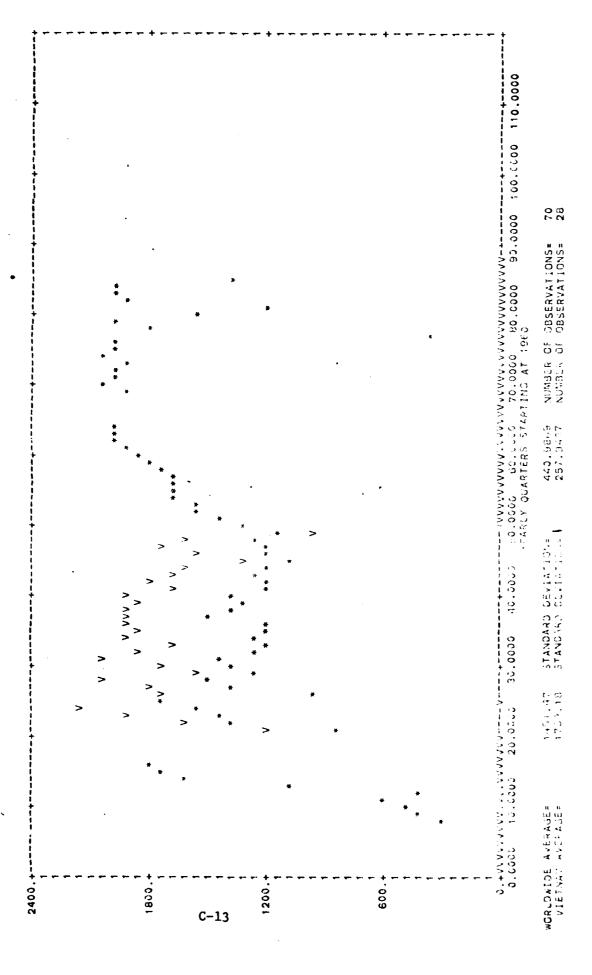
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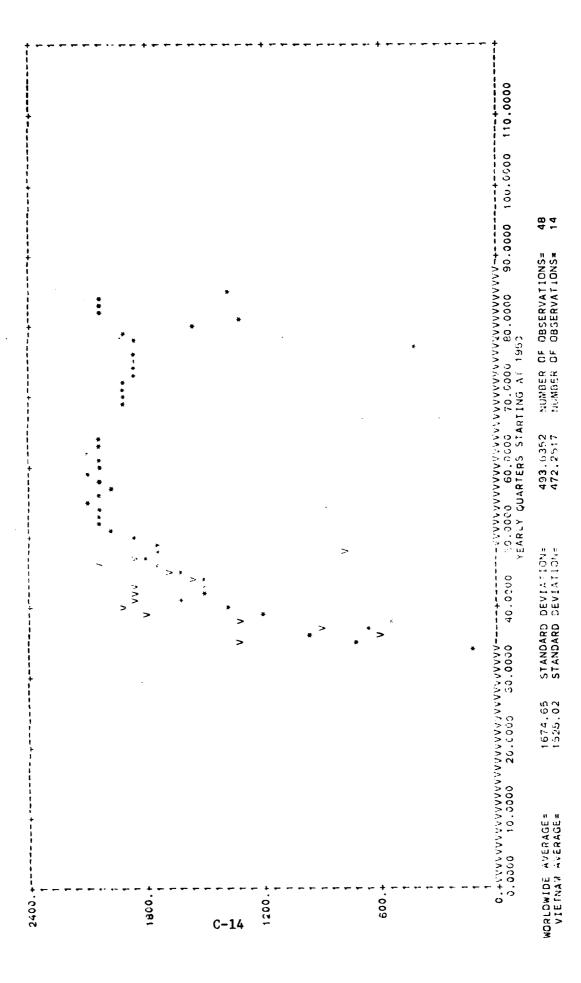




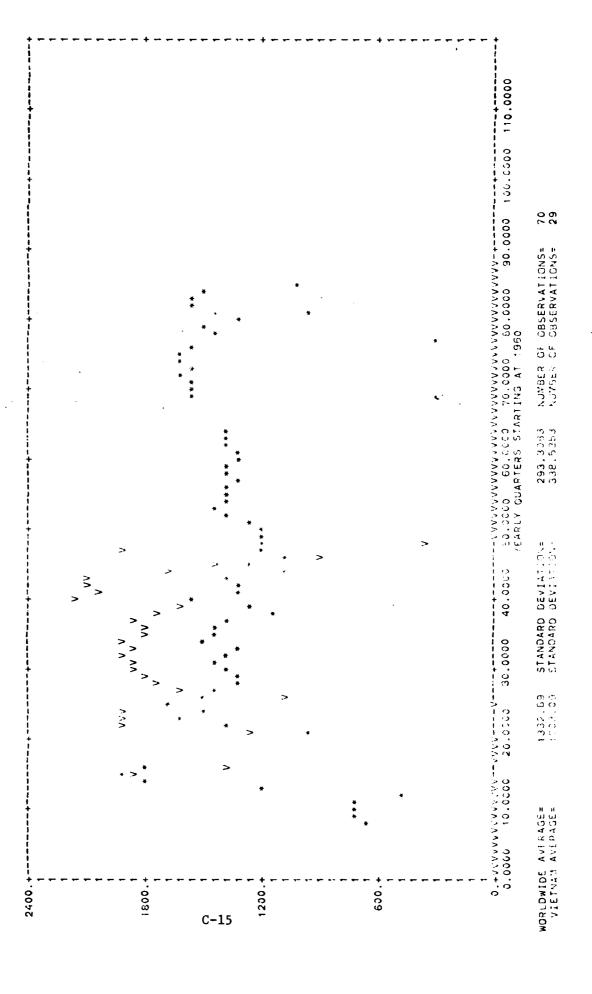


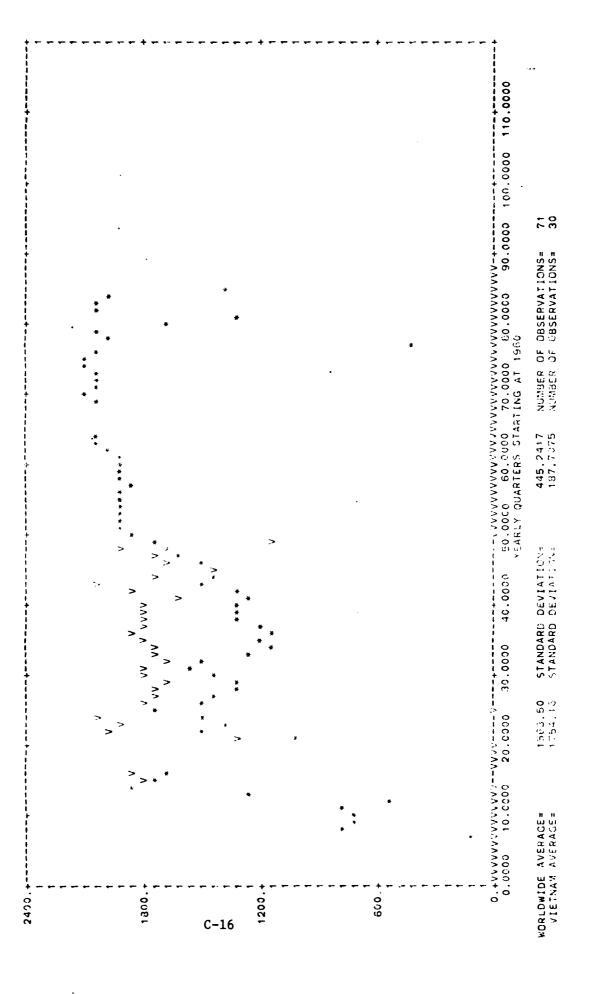
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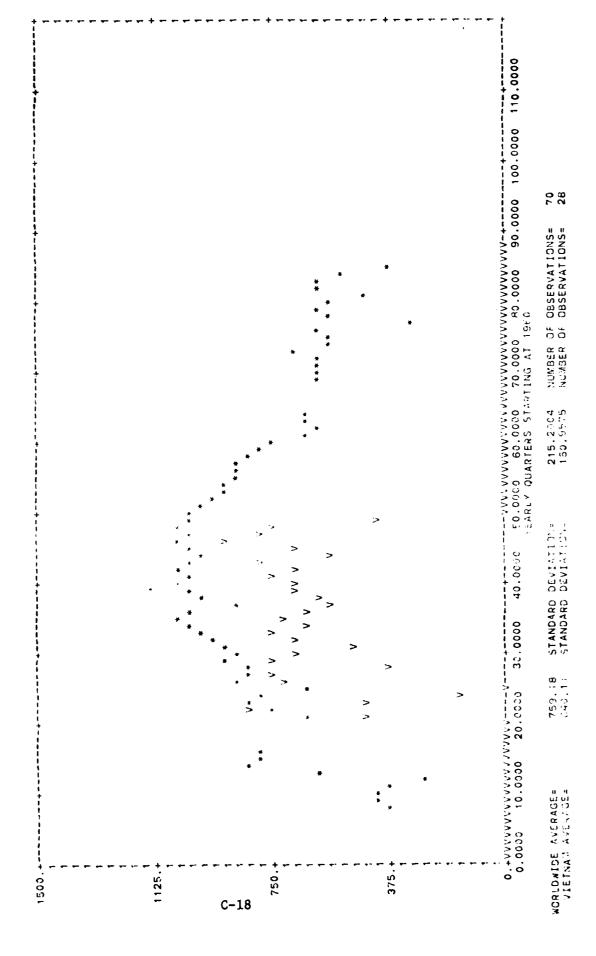
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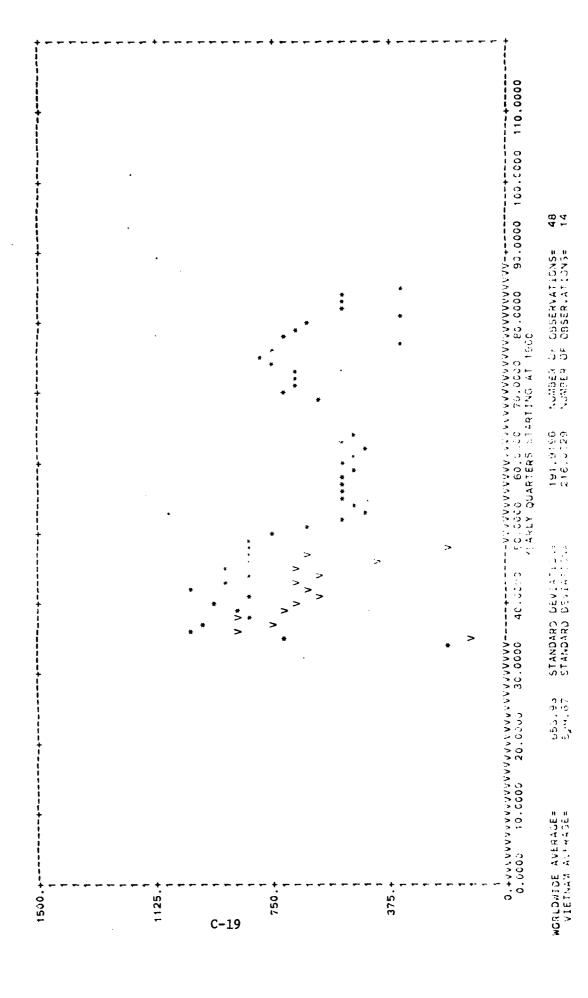




STANDARD DEVIATION OF HOURS ON HAND

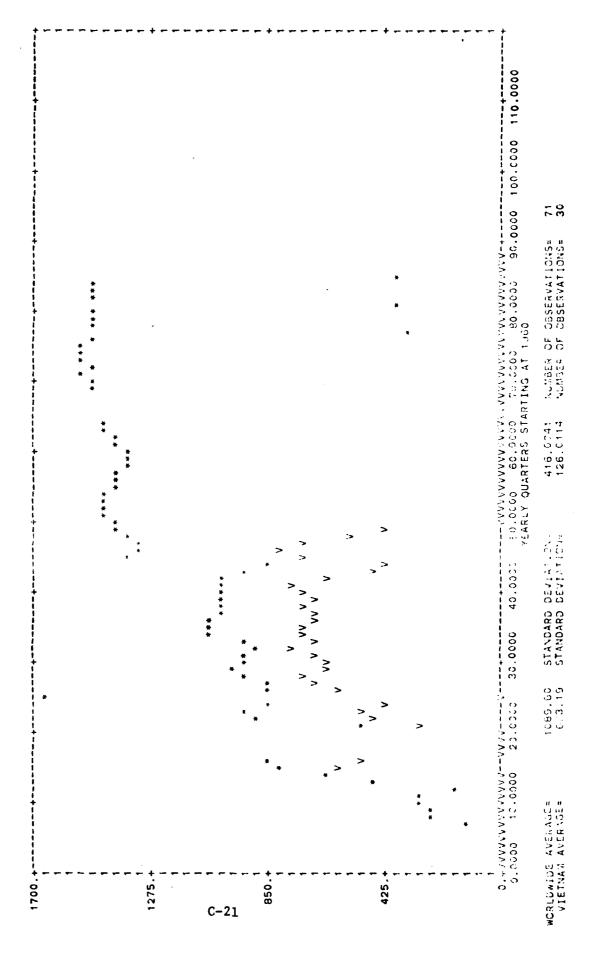
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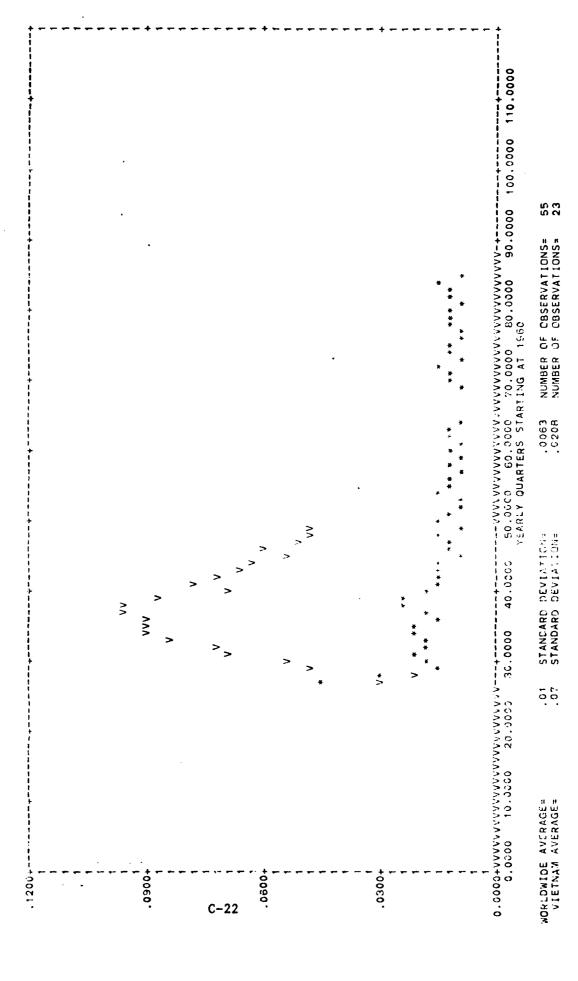


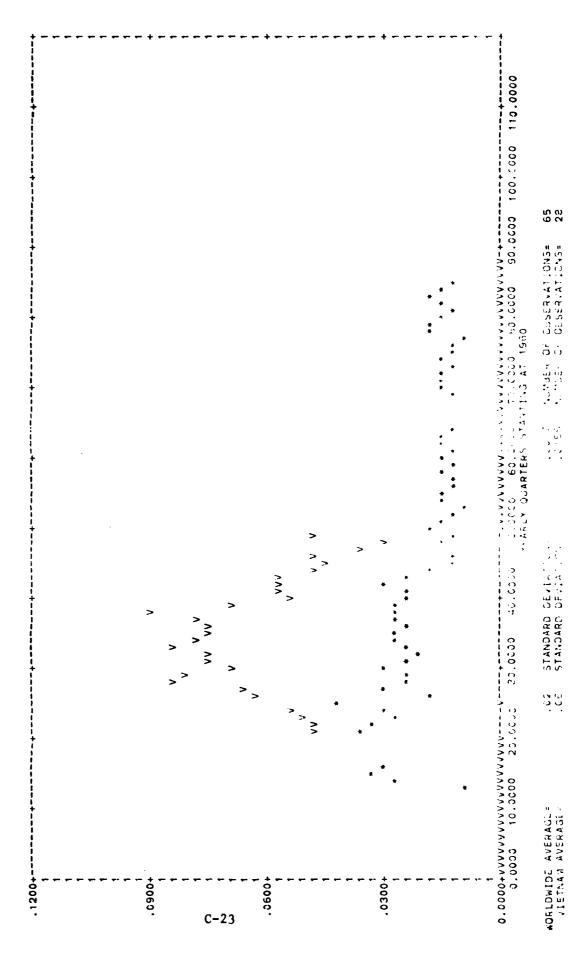
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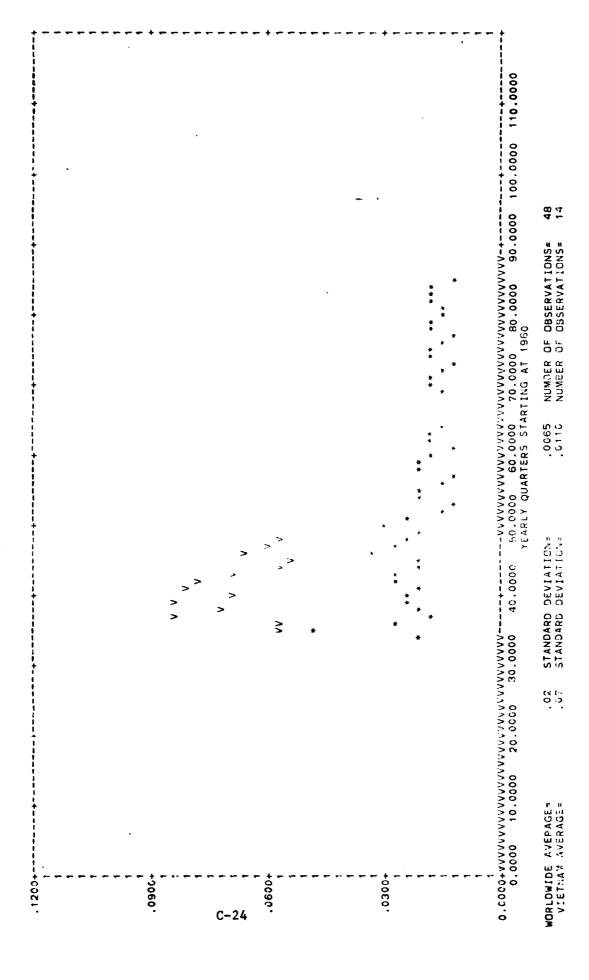
SYSTEM DV-1



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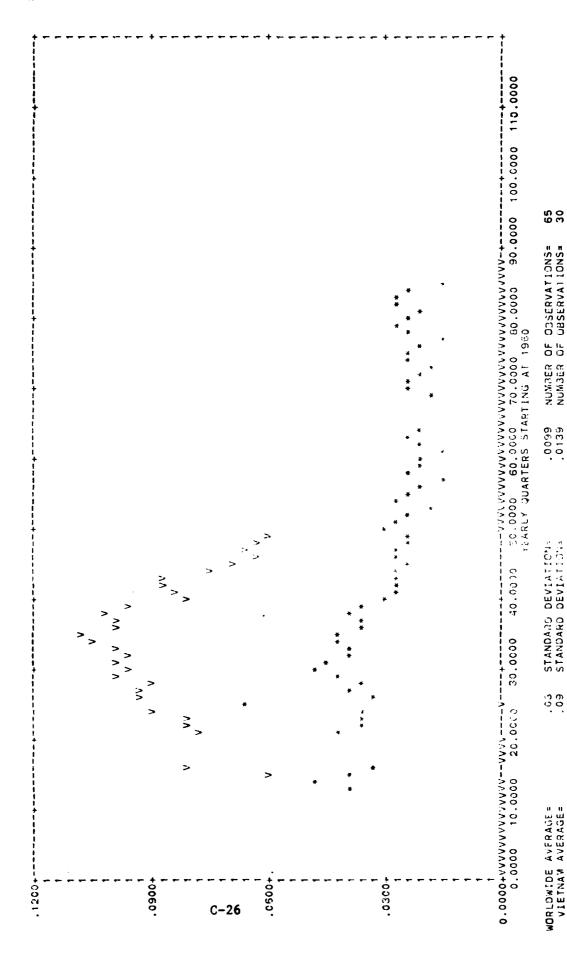


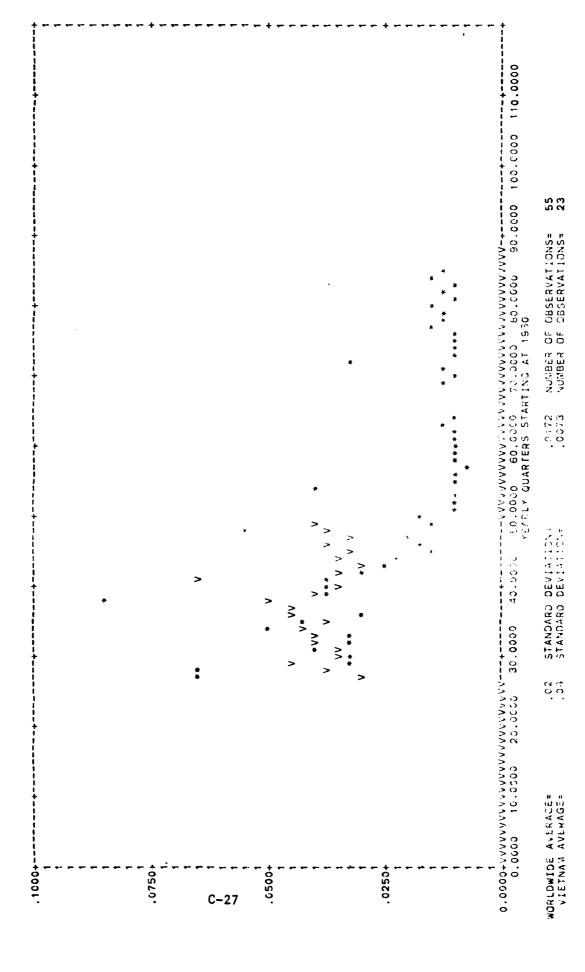


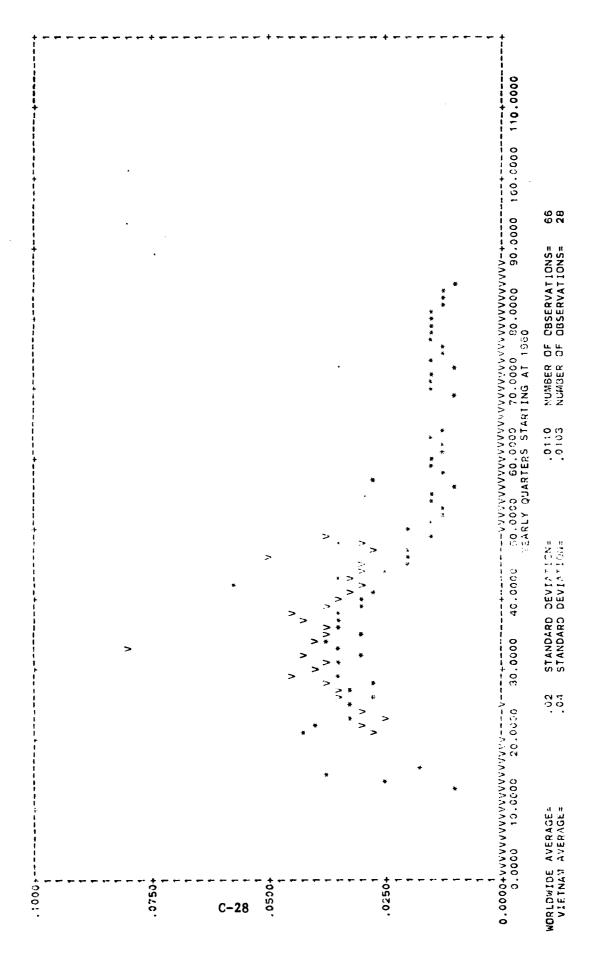


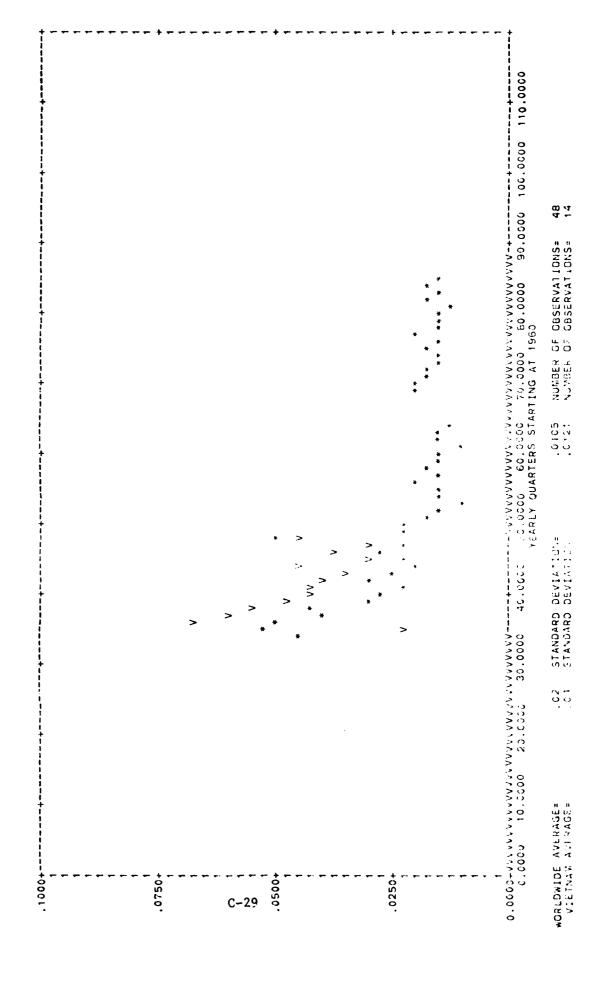
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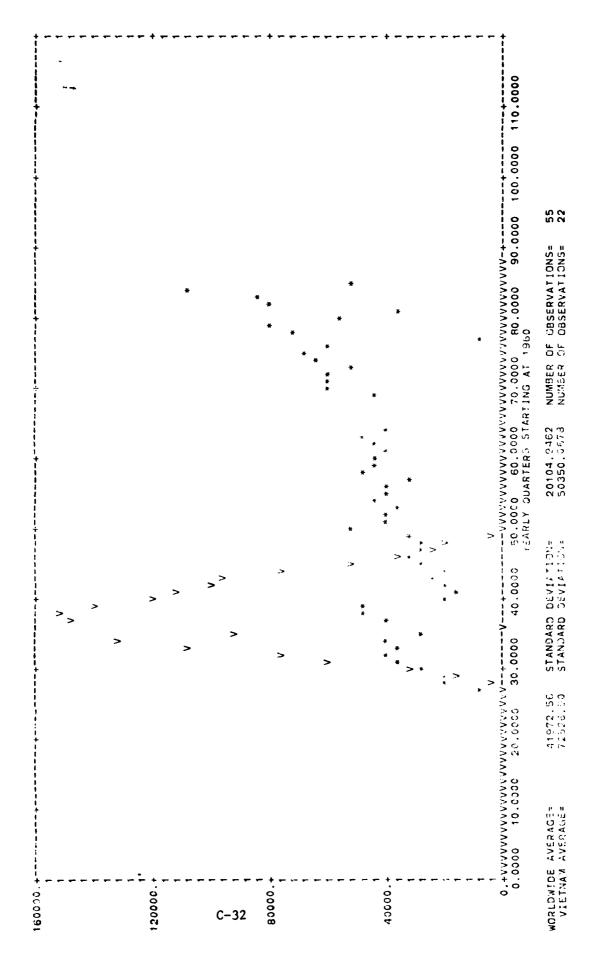




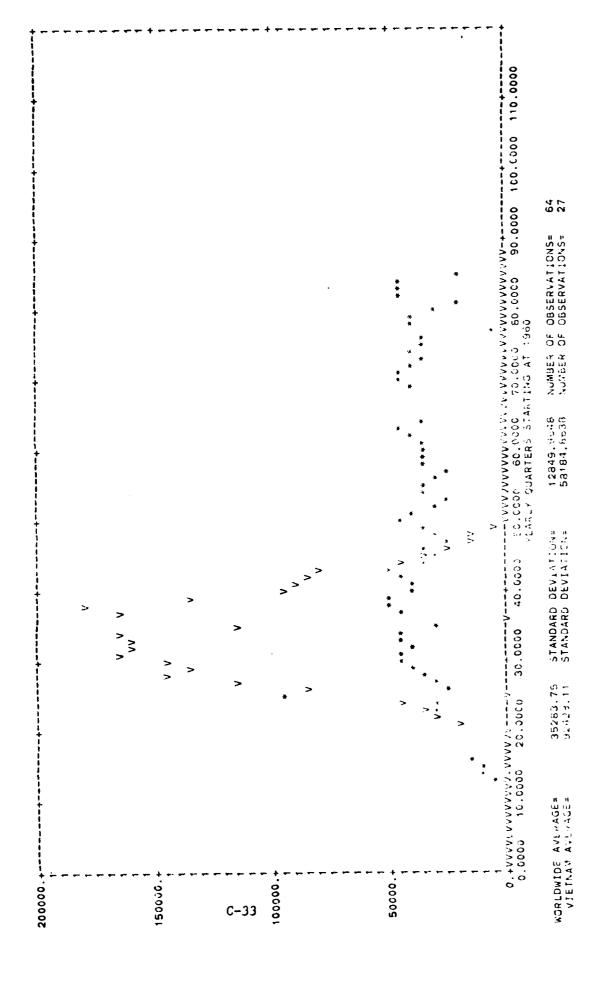


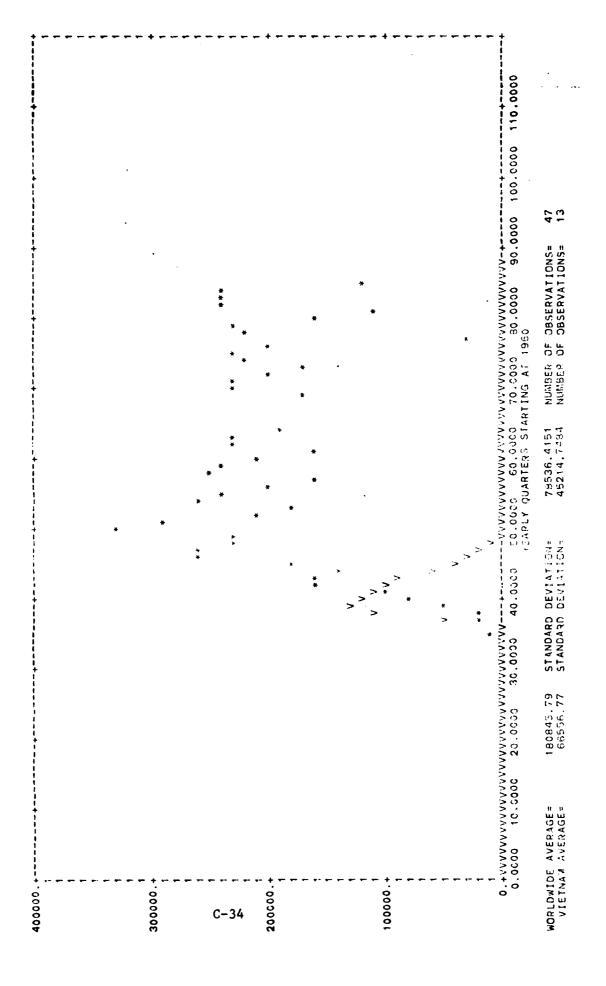
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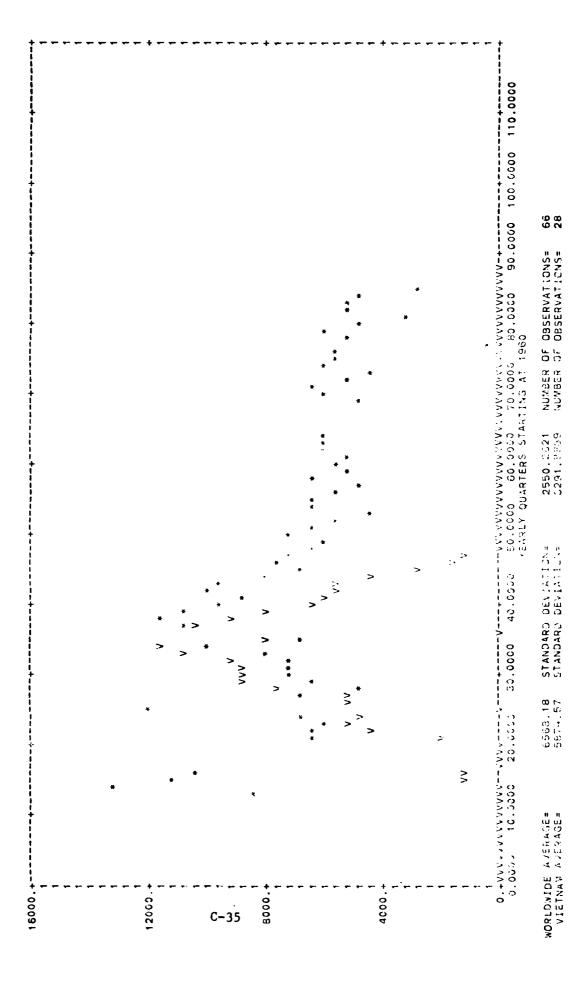
SYSTEM=UH-1 YVARIABLE=

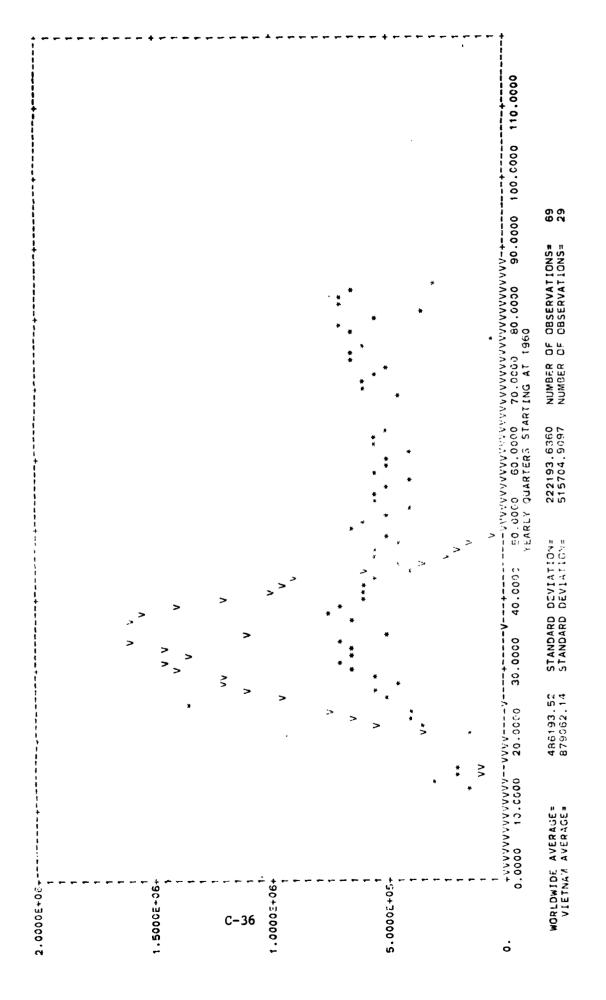


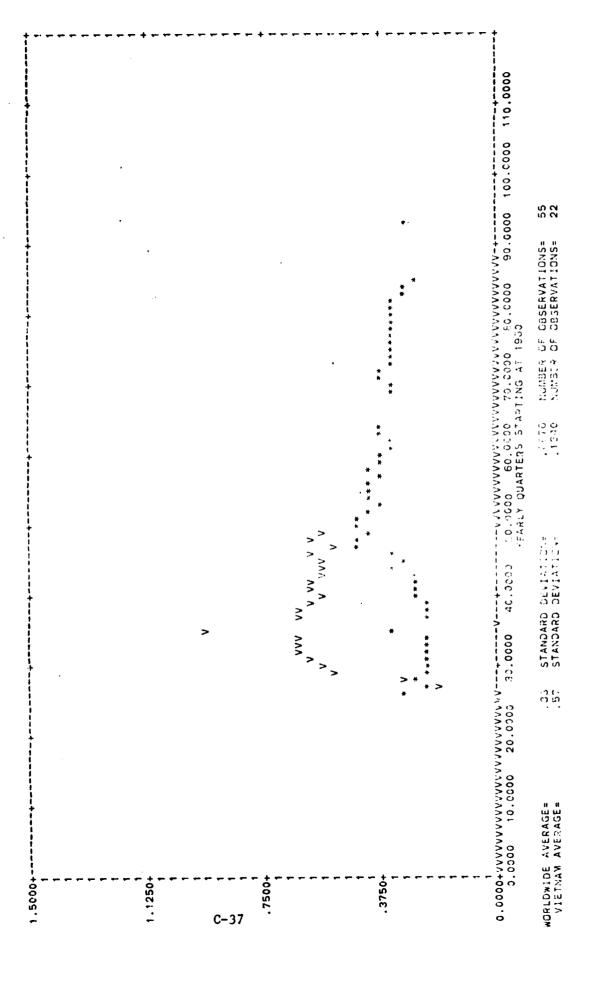
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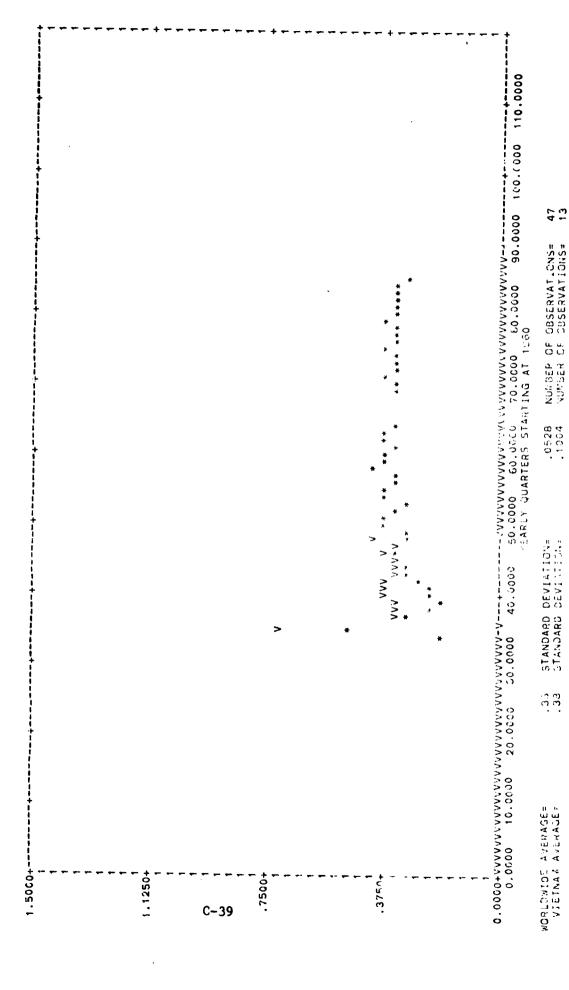




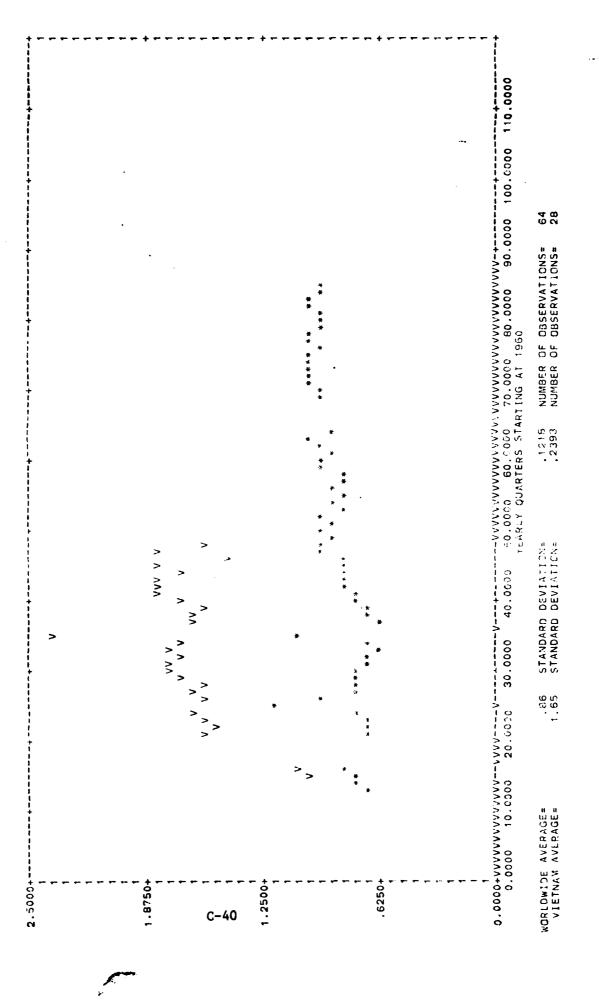


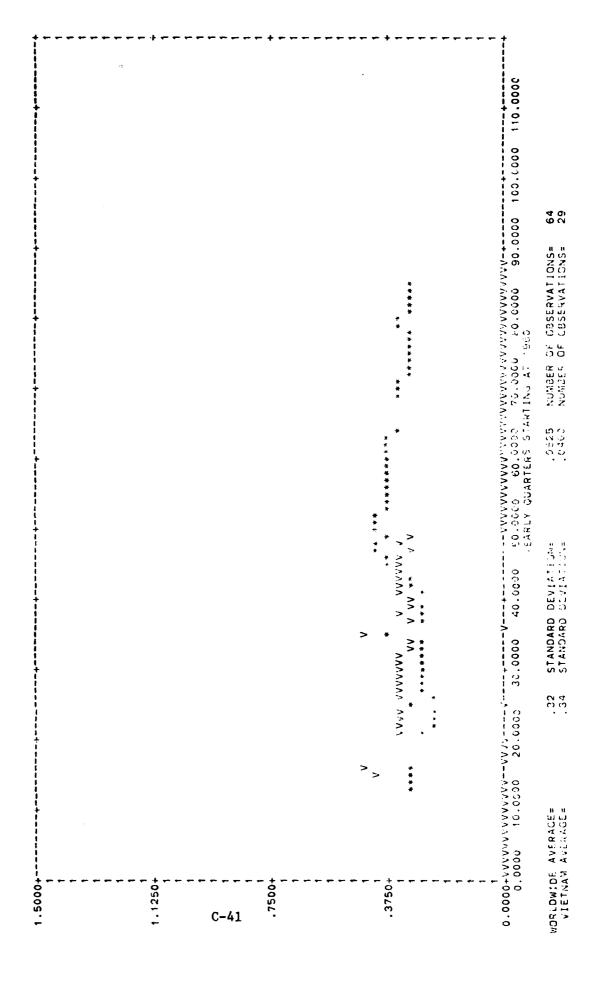


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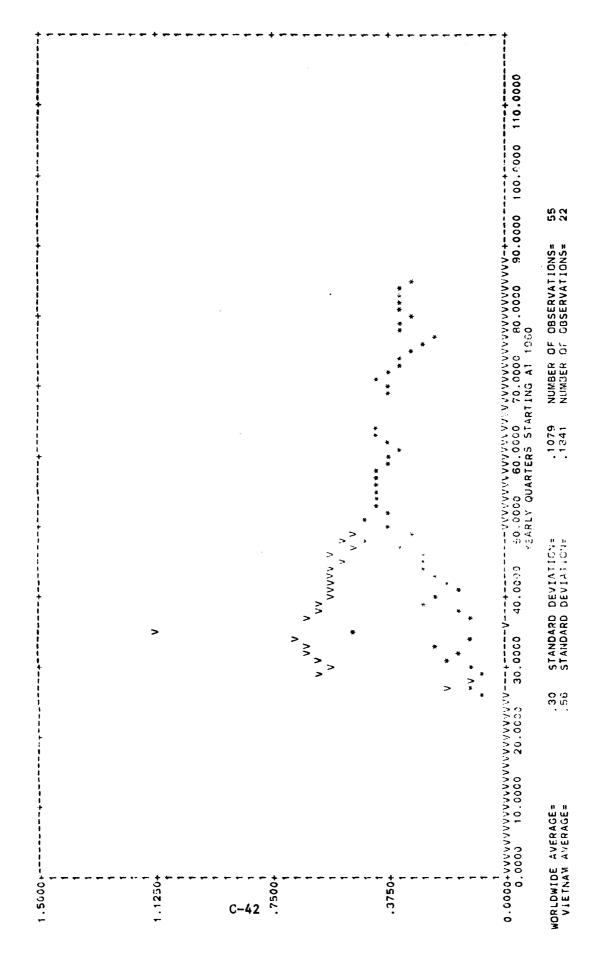


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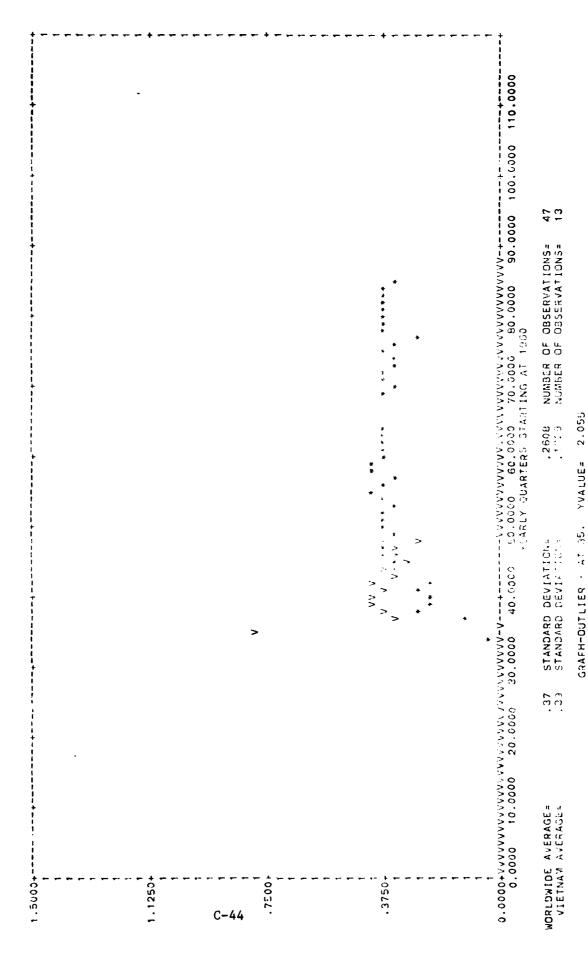


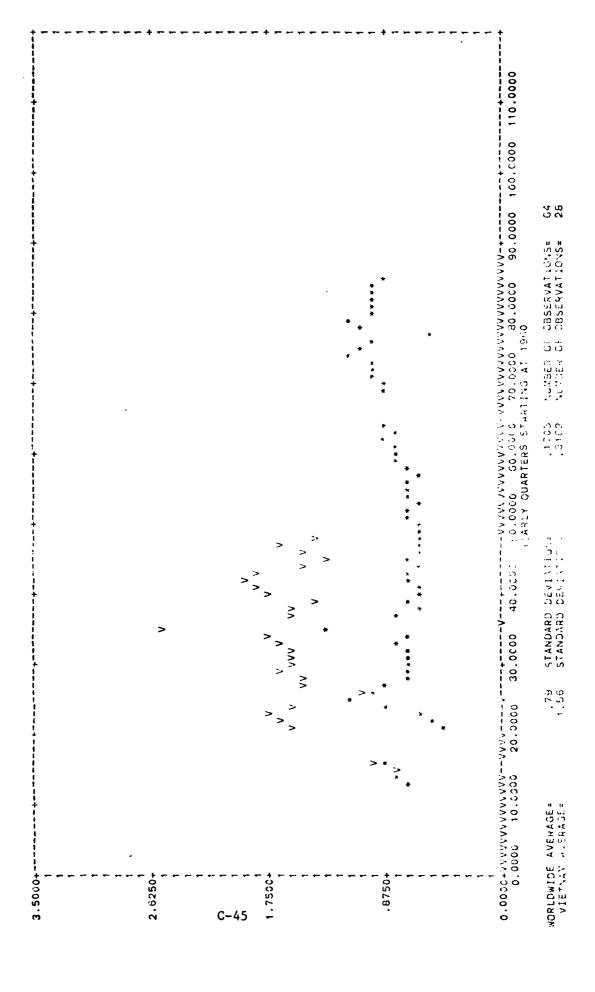


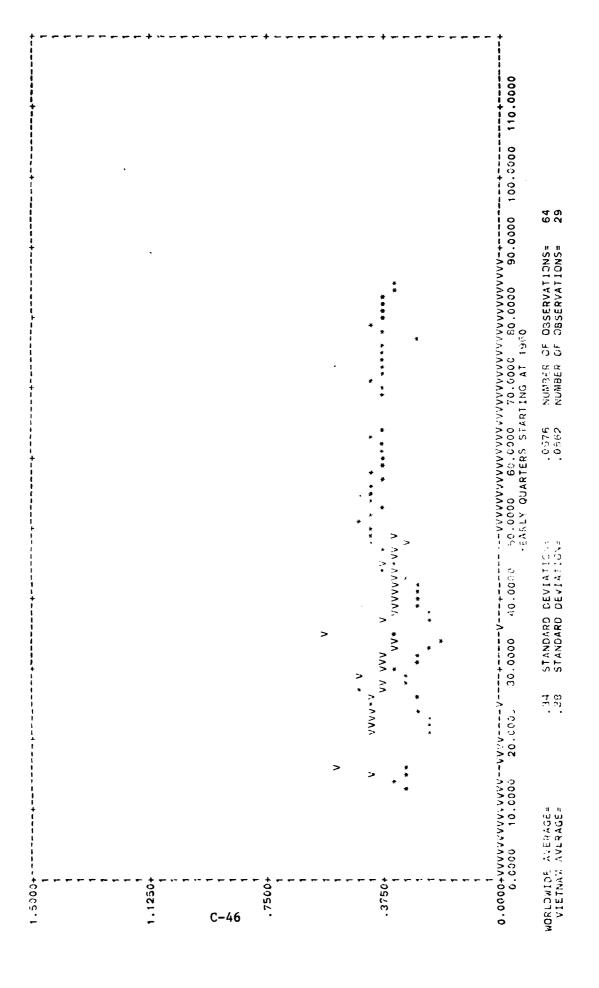
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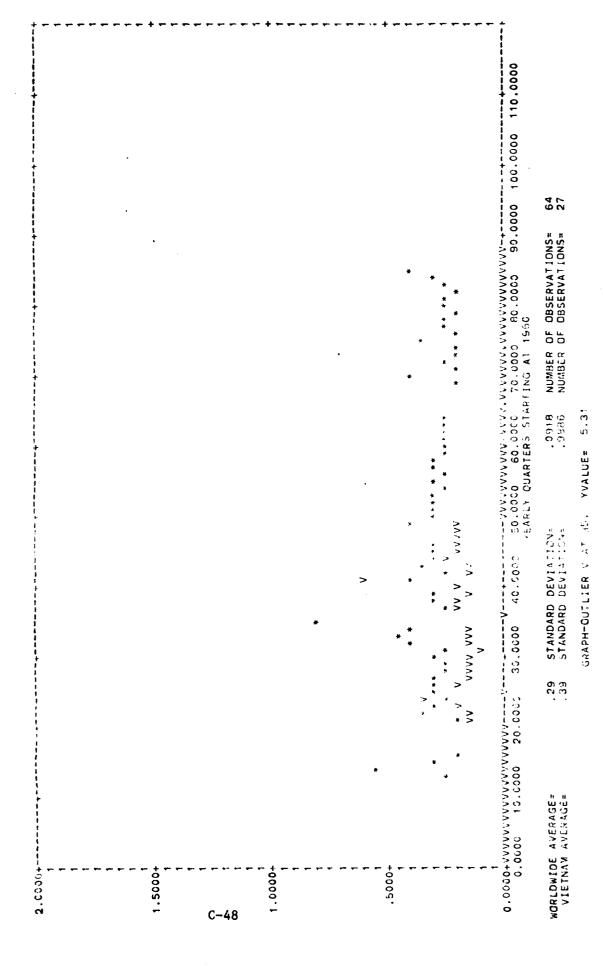
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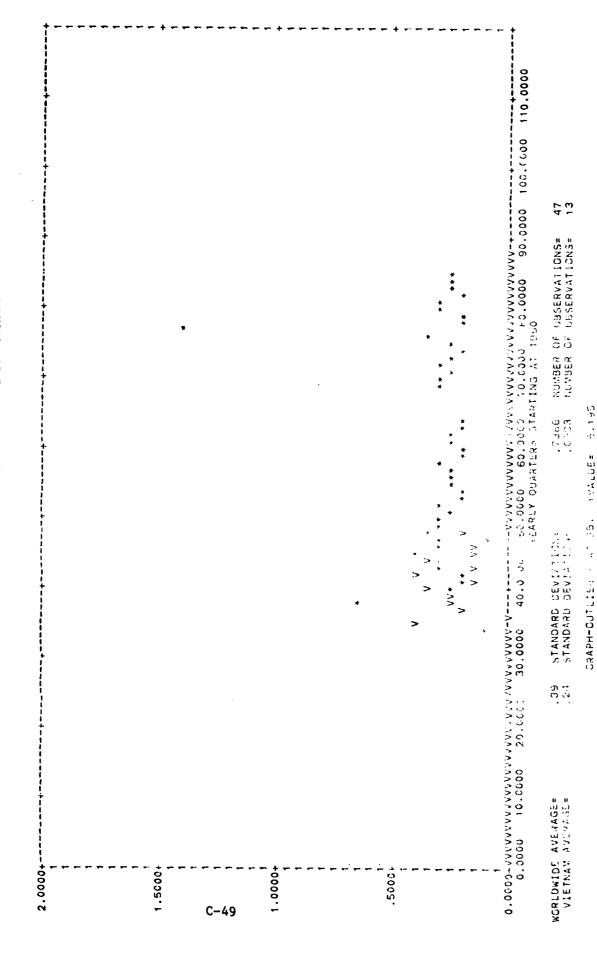


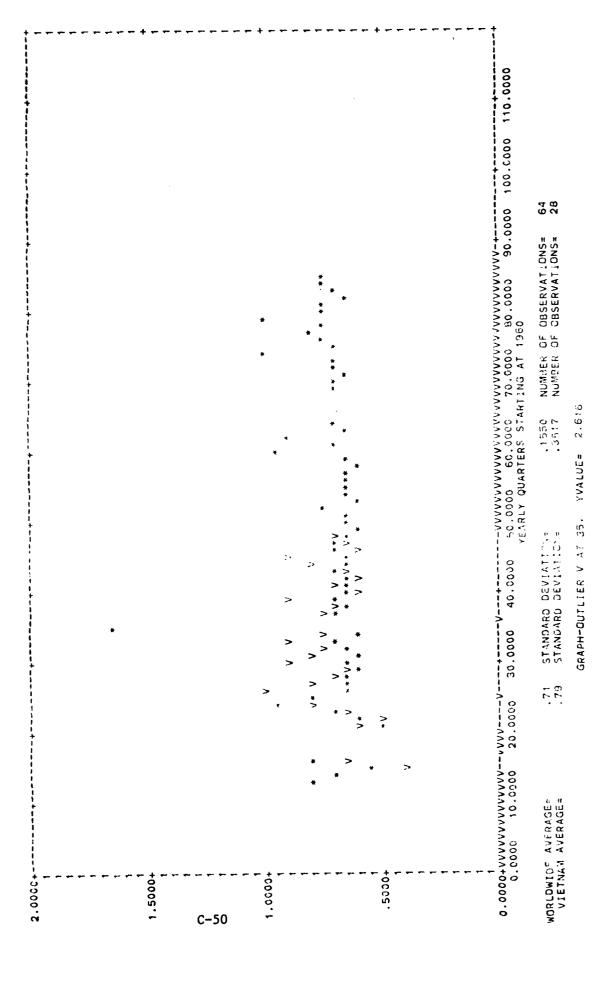


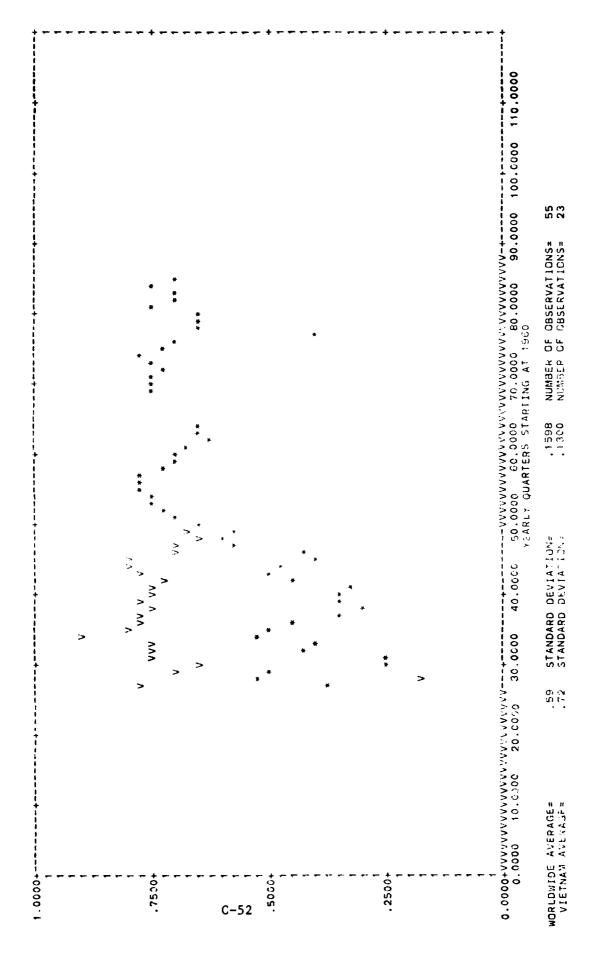


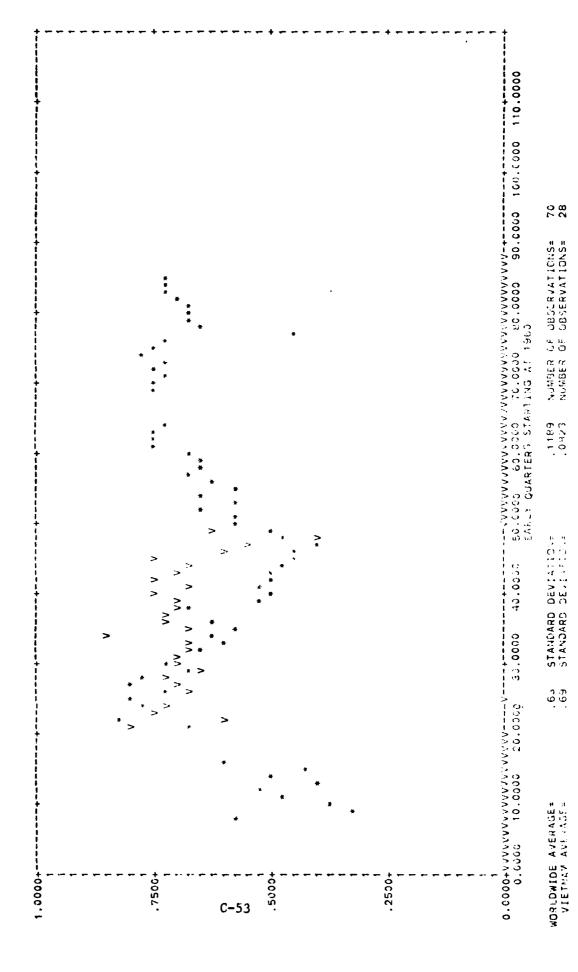
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WORLDWIDE A JERAGES VIETRAR A VINANCES	.37 STANDARD SEVIATIONS 14410 NUMBER OF UBSERVATIONS 55
	ORAPH-DUTLIER - AT 75. YVALUE= 0.448 Oraph-dutlier - At 15. Yvalue= 2.421

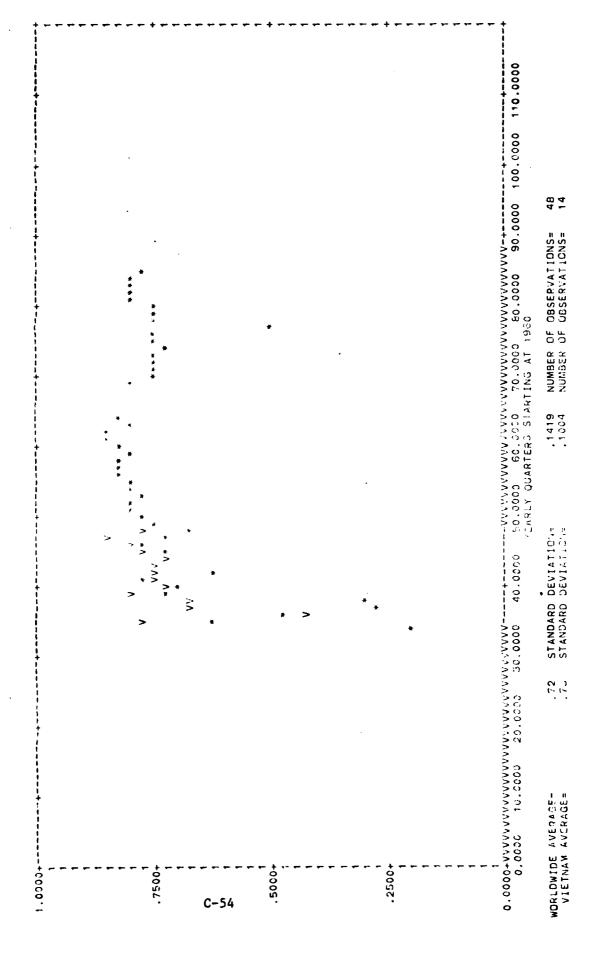


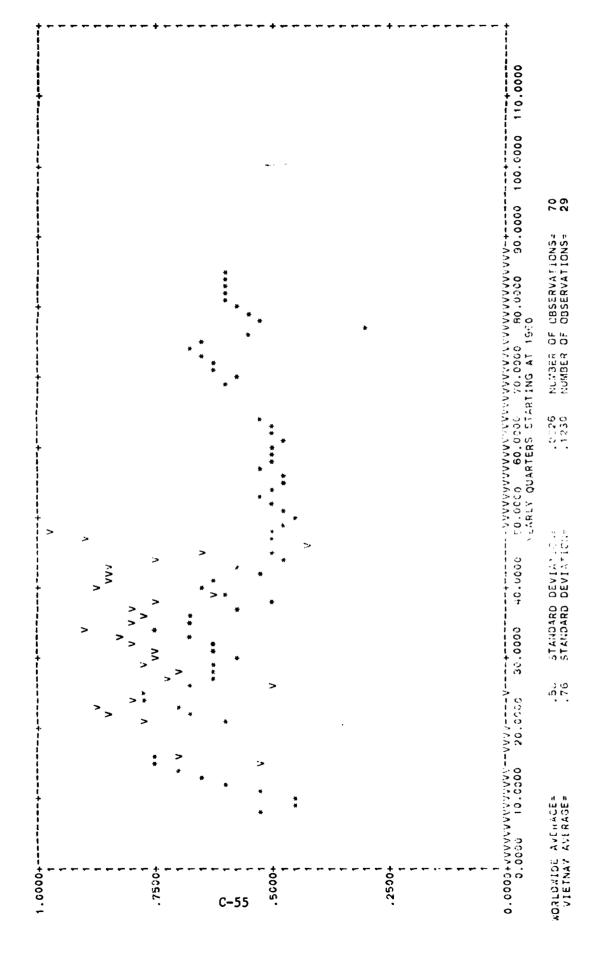


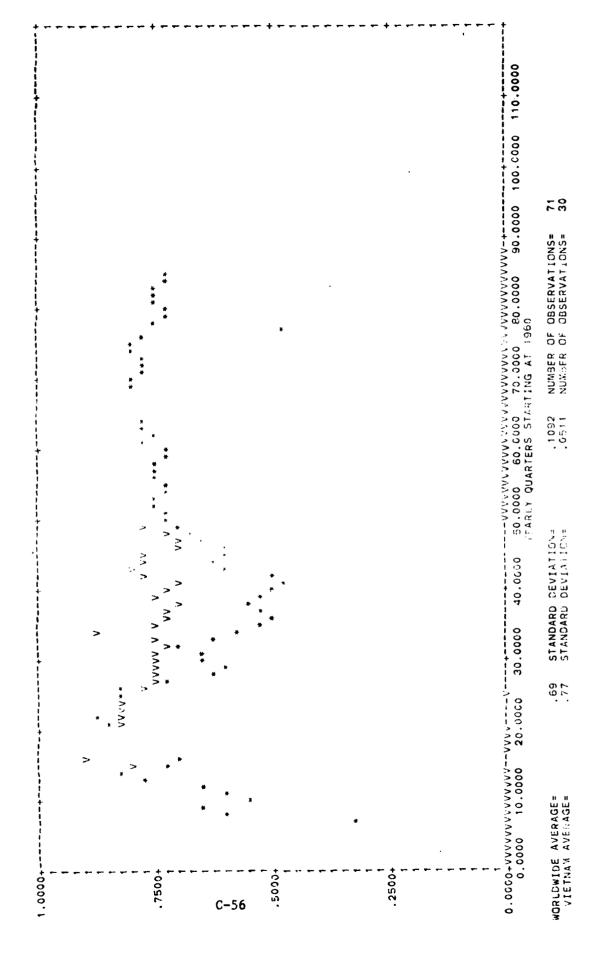


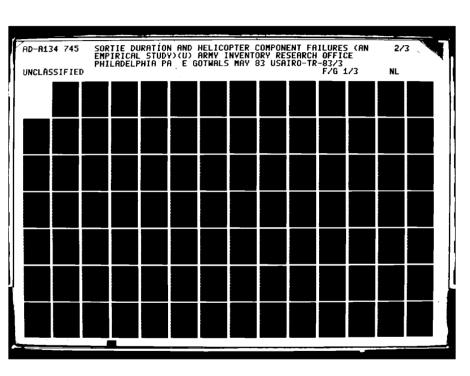


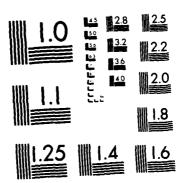






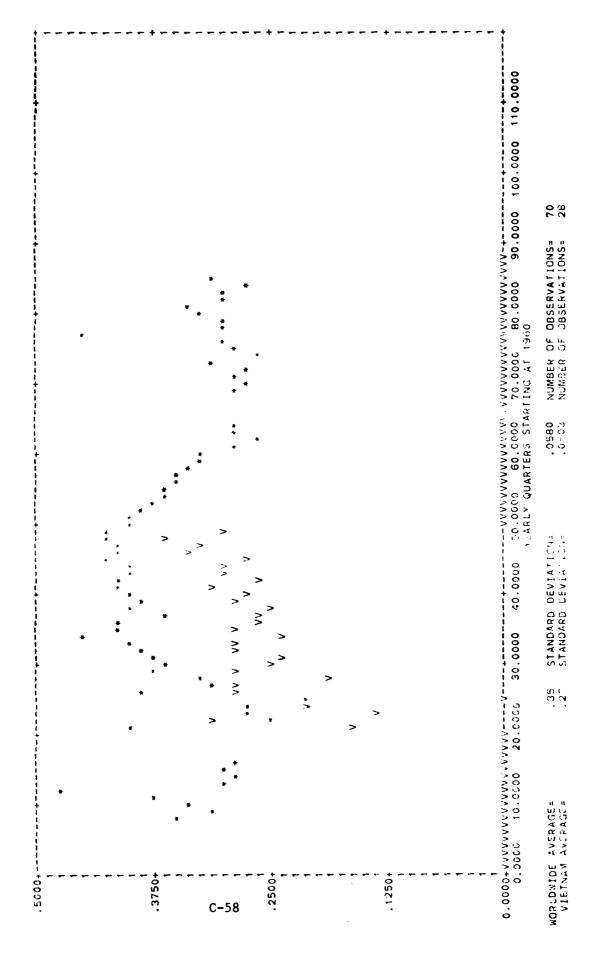


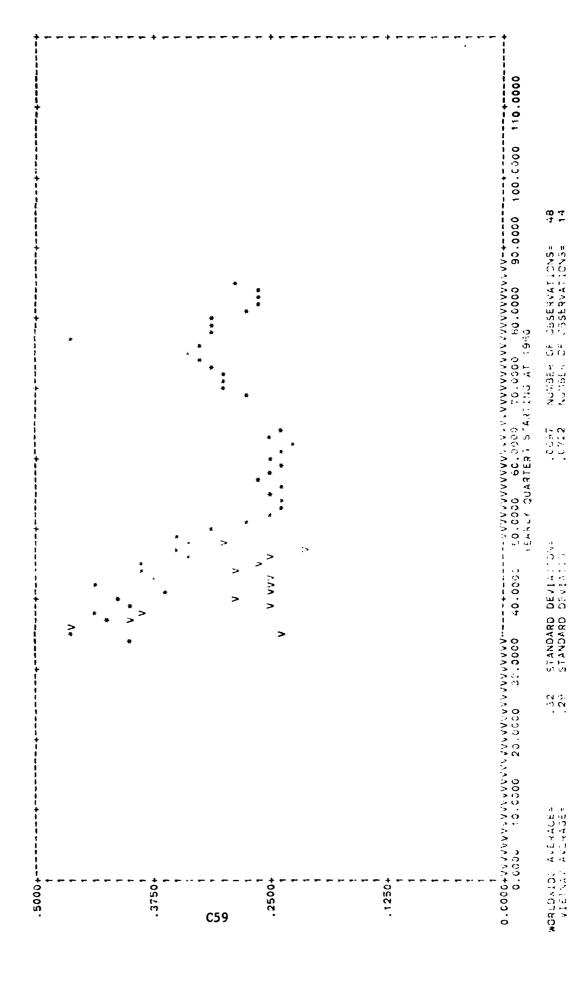


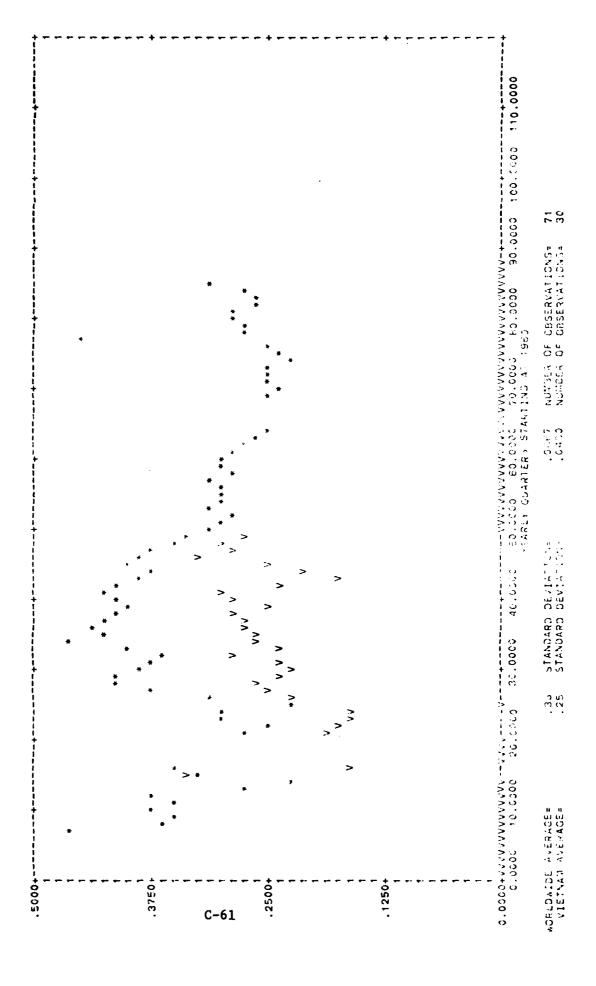


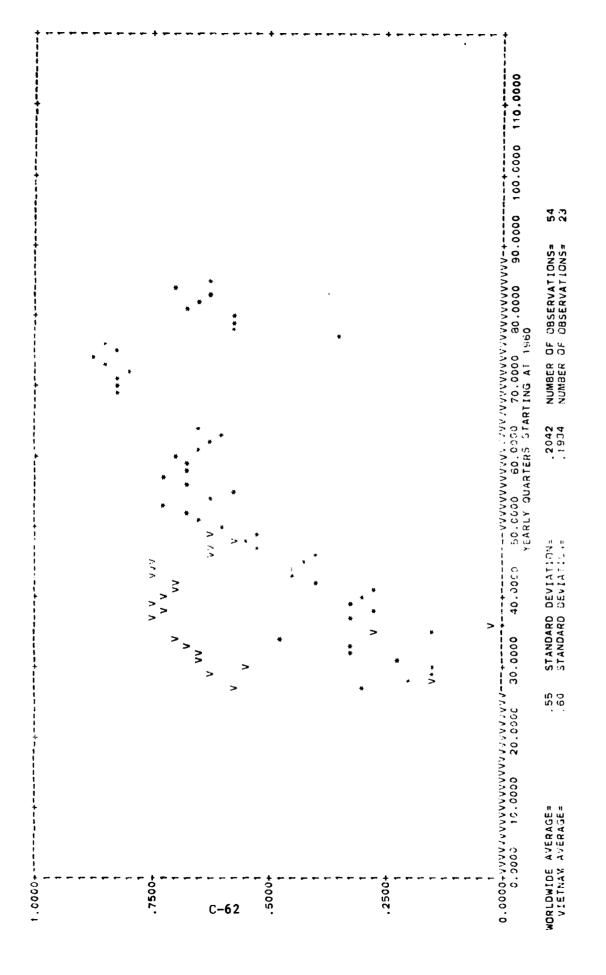
MICROCOPY RESOLUTION TEST CHART NATIONAL BUREAU OF STANDARDS-1963-A

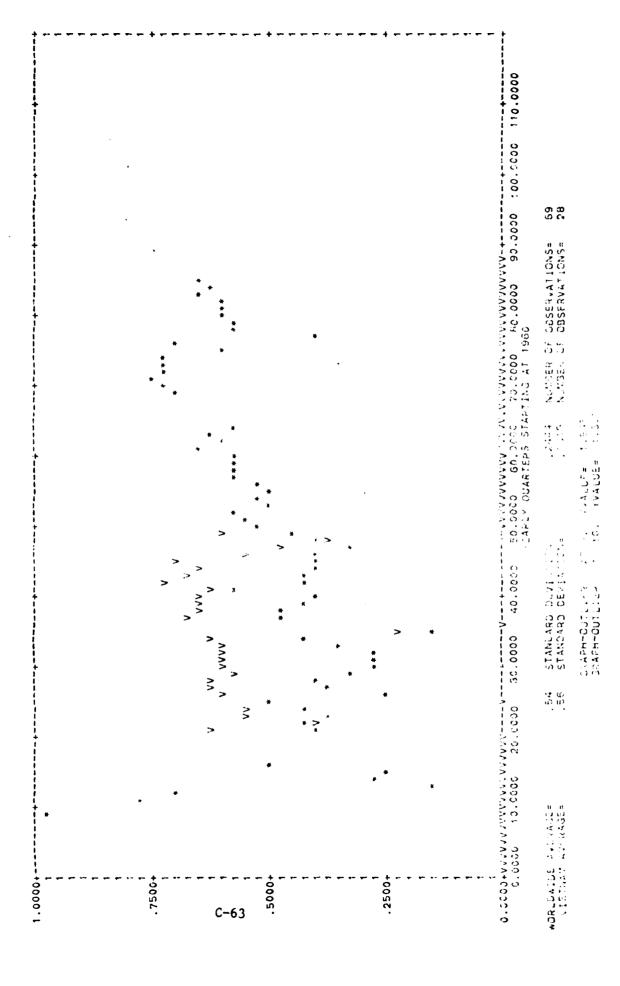
.5000+--

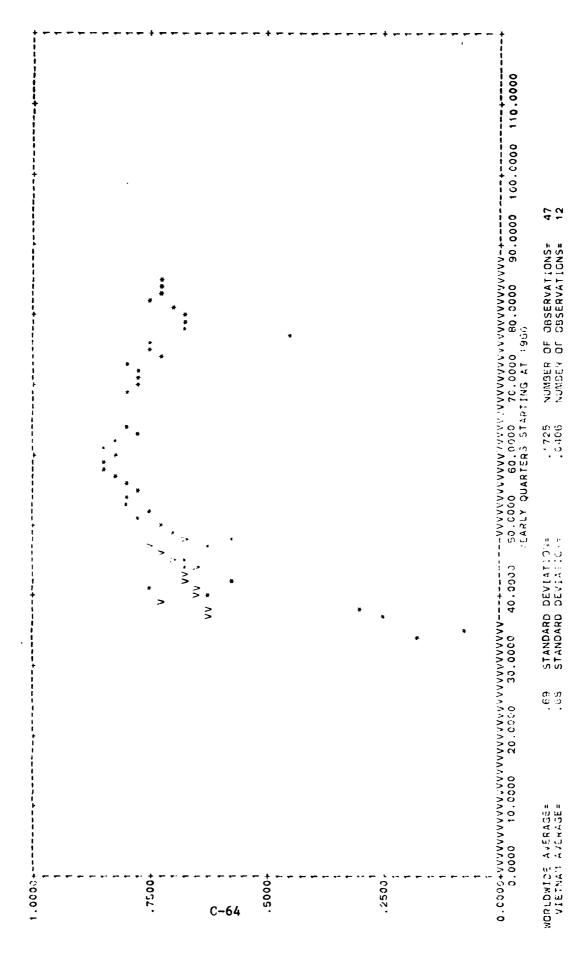




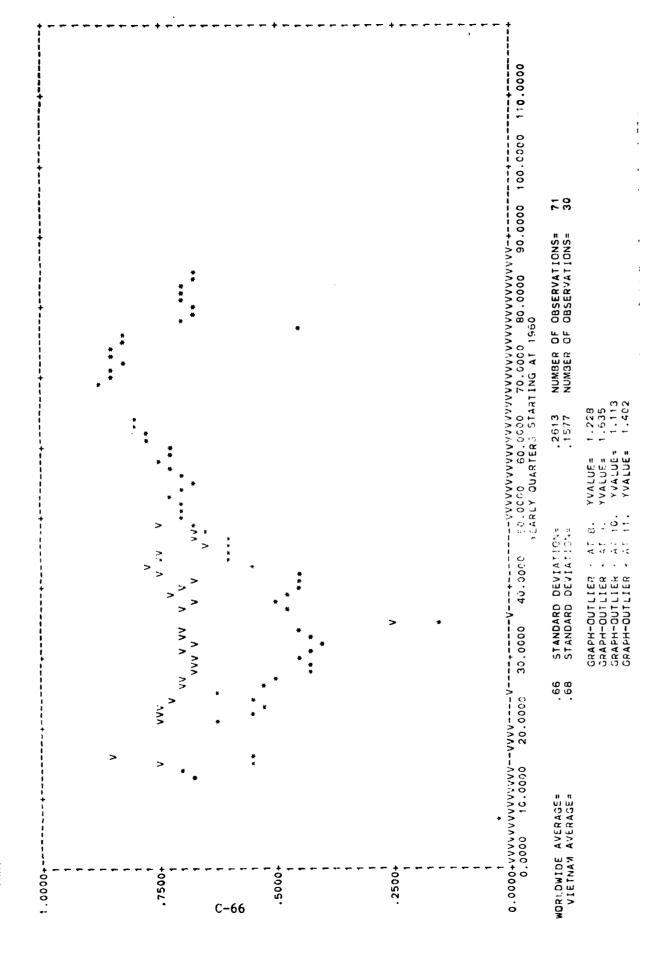




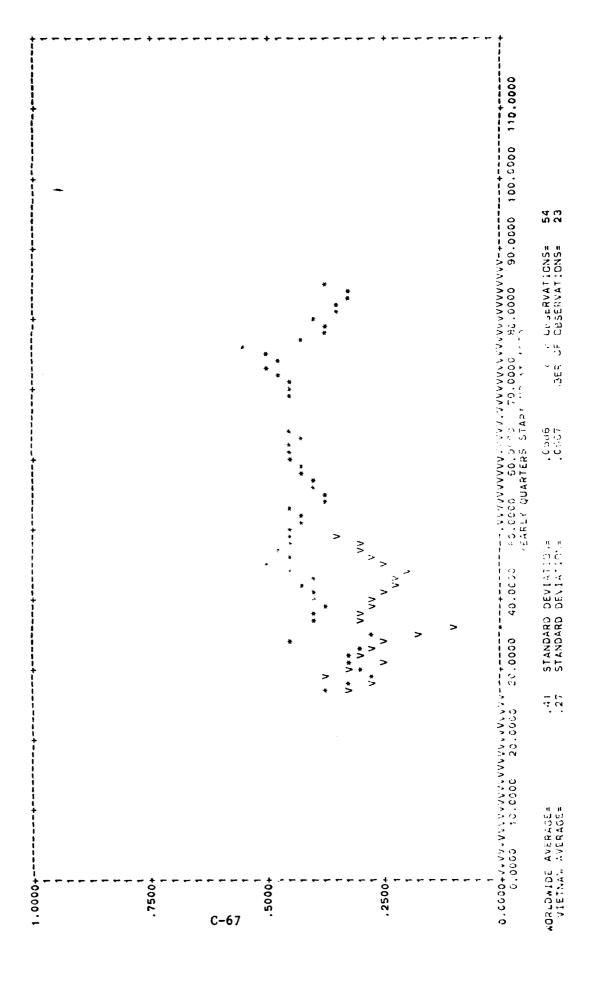


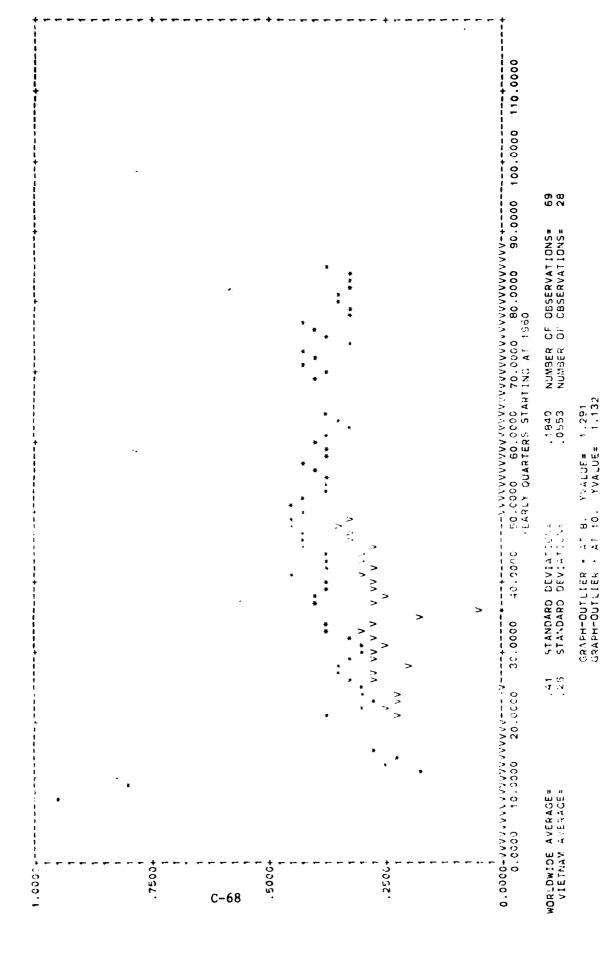


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WORLDWIDE AVERAGES SE STANDARD DEVIATIONS .1753 NUMBER OF DB	OF DBSERVATIONS= 70 OF DBSERVATIONS= 29



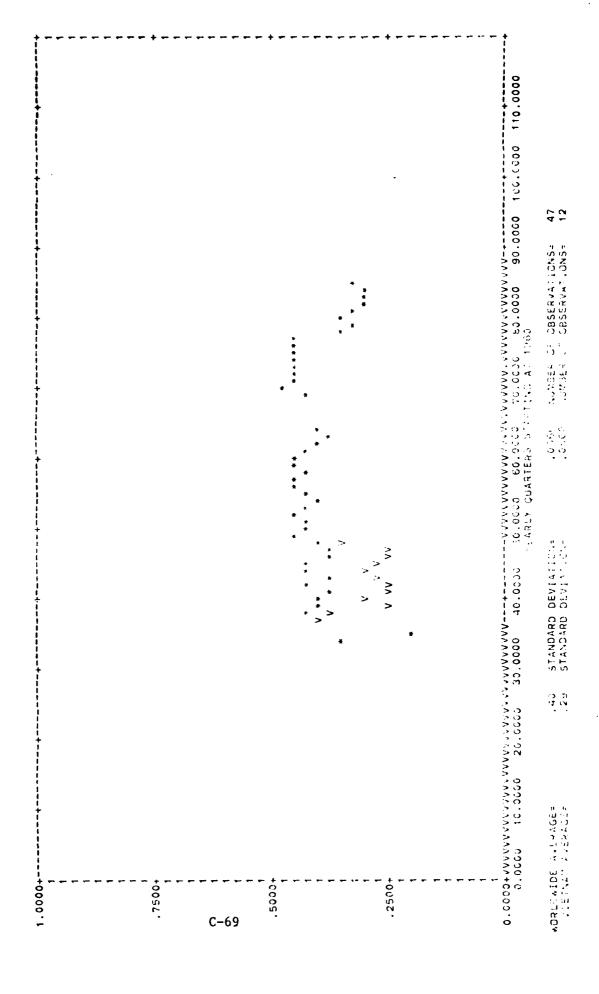
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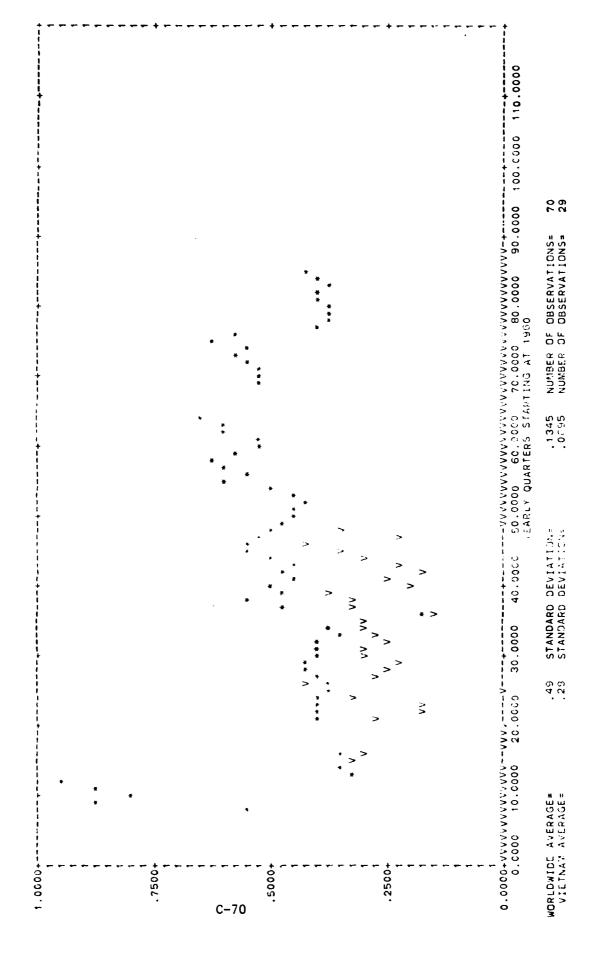


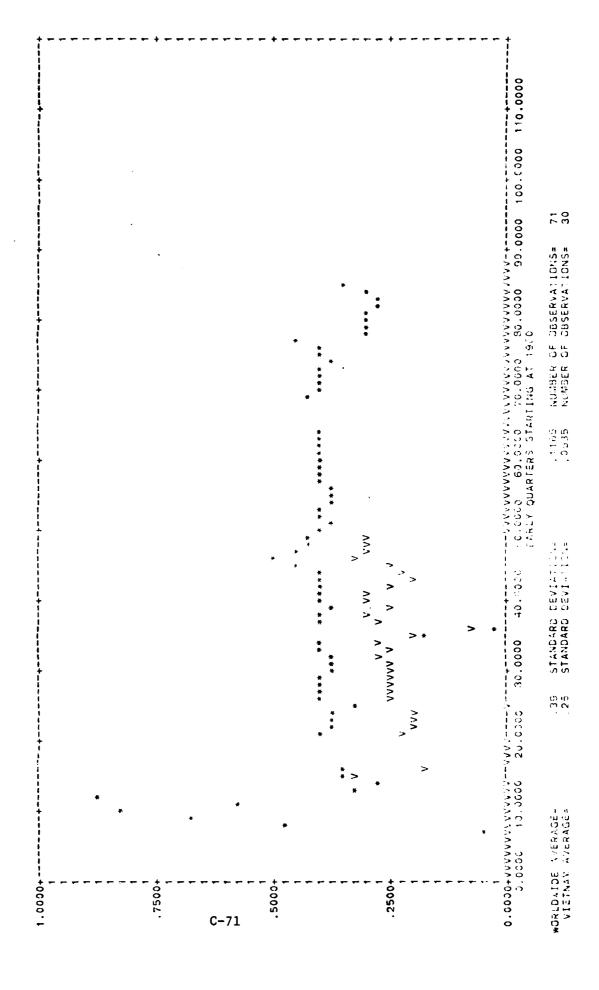


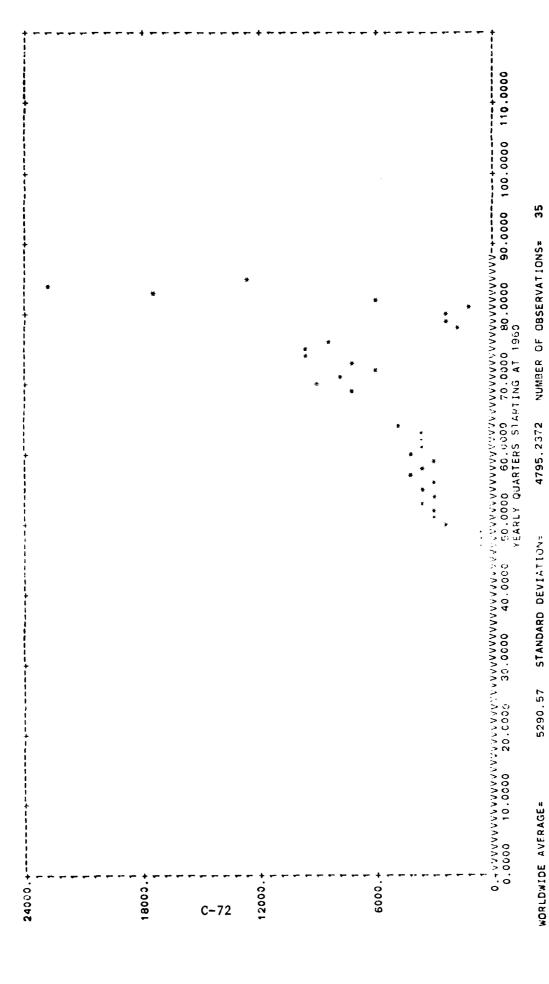
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NUMBER OF AUTOROTATIONS

SYSTEM=CH47 YVARIABLE= NUMBER OF UBSERVATIONS=

6435.3689

STANDARD DEVIATIONS

1454.37

WORLDWIDE AVERAGES

40000+ 100001 30005. 20002 C-74

35

NUMBER OF OBSERVATIONS=

8827.8916

STANDARD DEVIATIONS

17718.63

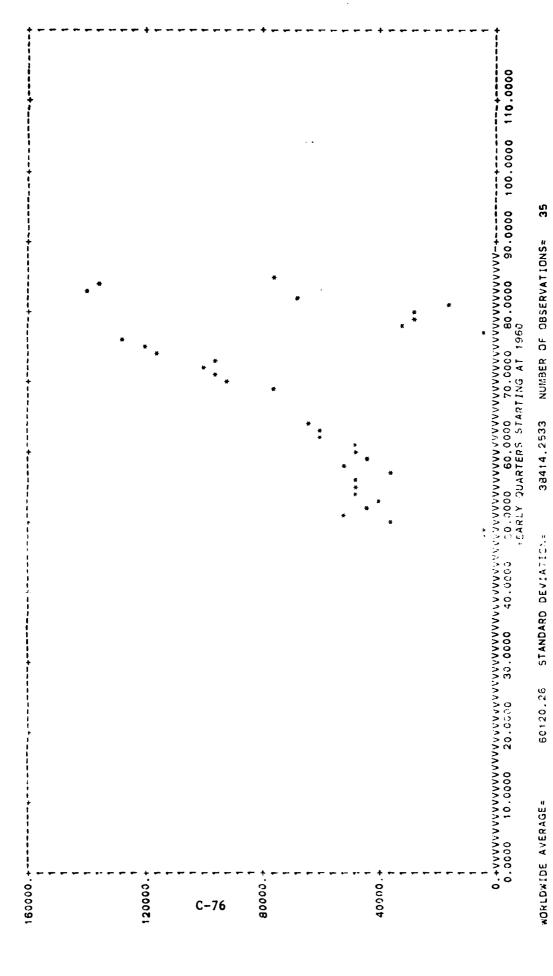
WORLDWIDE AVERAGE=

8.0000+--2.0000+ 6.0000+ 4.00004 C75

ADRIDAIDE AVERAGE=

STANDARD DEVIATIONS 3.33

NUMBER OF OBSERVATIONS* 3.2146



# APPENDIX D

# 1352 PROGRAM DATA

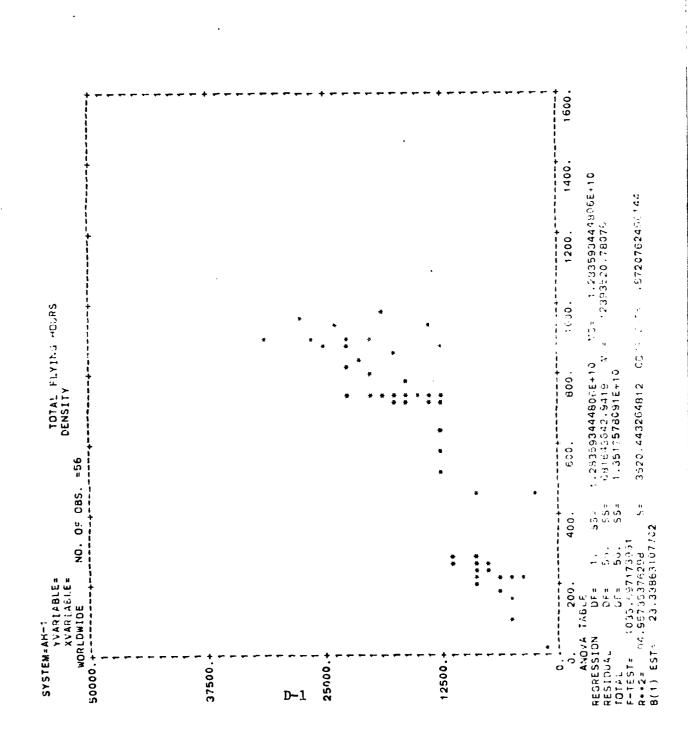
# REGRESSION ANALYSIS

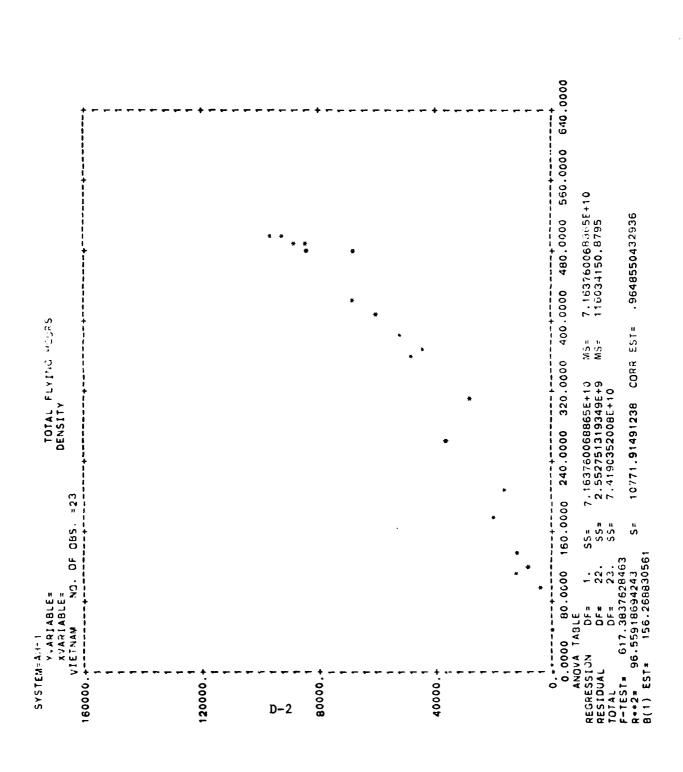
Y VARIABLE (DEPENDENT)

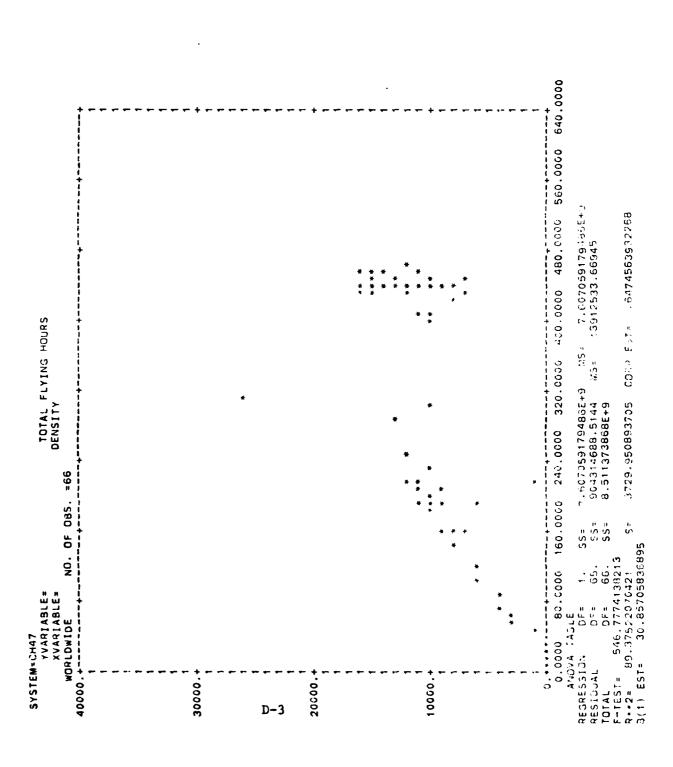
X VARIABLE (INDEPENDENT)

TOTAL FLYING HOURS
TOTAL NUMBER OF SORTIES

DENSITY
TOTAL FLYING HOURS







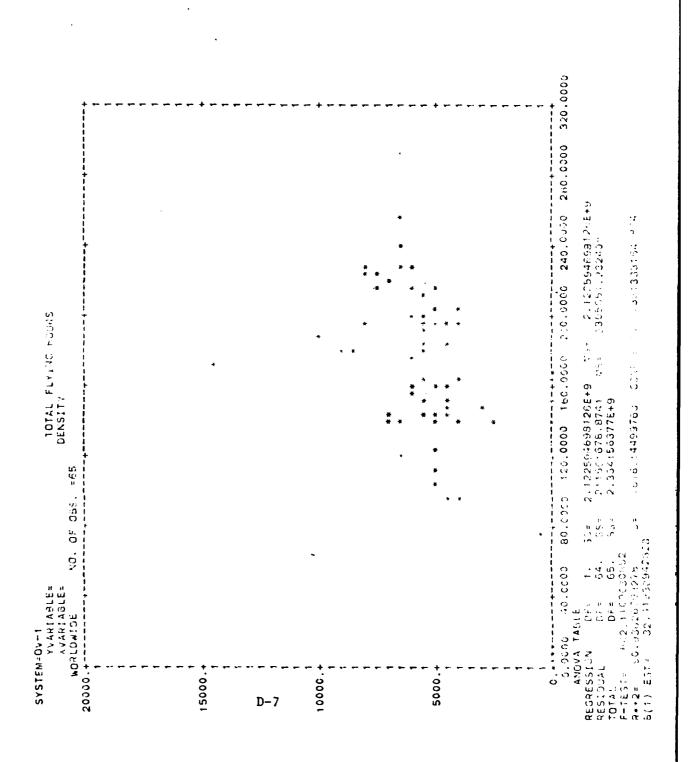
0.0000 80.0000 160.0000 240.0000 320.0000 400.0000 400.0000 560.0000 640.0000 ANDVA TABLE 3.654775875019E+10 44490521.73387 .9549990969963 TOTAL FLYING TOURS DENSITY 6670,121568148 CGRR EST= . S. <u>.</u> . 3.654775875019E+10 1.201244086814E+9 3.7749022837E+10 NO. OF 085. =28 \$8≈ \$8≈ š P21,4729188569 96.81781240156 7= 140,5163801332 1. 27. 28. 80000 . + - - - - - - - 0000B YVANIABLES XVANIABLES VIETHAN 0.0000 SYSTEM C. 47 REGRESSION RESIDUAL TOTAL F-TEST= R+*2= 96 B(i) EST= 60000+ 40C00.+ 20000.+ D-4

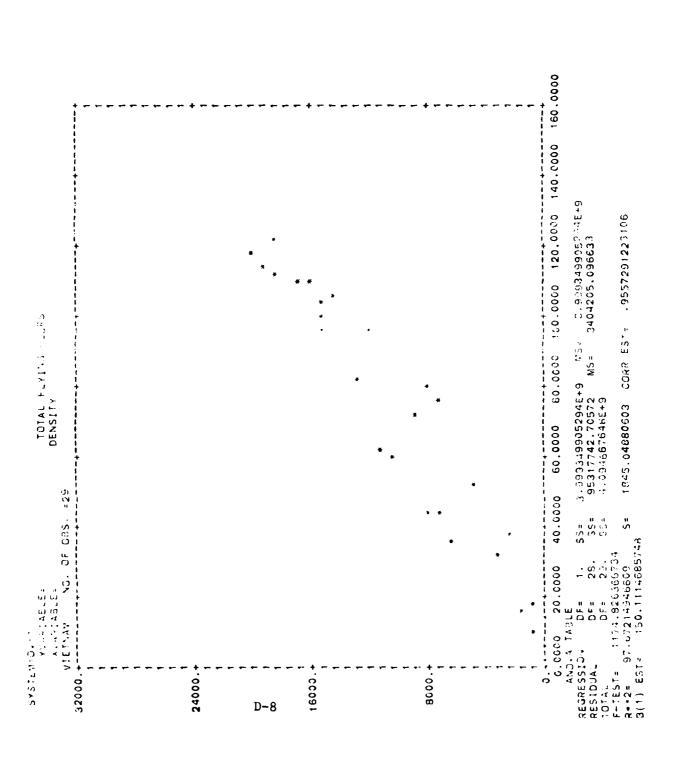
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3200. 2.085283250004E+11 300852855.3749 .8278409515436 2400. TOTAL FLYIMG HOURS DENSITY *7545,10978273 CO44 2774 2.085233250204E+11 1.414008316862E+10 2.22668408189E+11 1600. 1200. NO. OF 035. =48 58. 58. 88. ANOVA TALE
REGRESSION DF= 1. 05=
RESIDUAL DF= 47. 55
TOTAL DF= 46. SS
F-TEST= 693.1240;0C799
R**2= 93.6497;254454
B(1) EST* 00.6176657672 0. . - - - - - - 0 SYSTEM: 0H58 50000.+ 150000.4 1000001 D-5

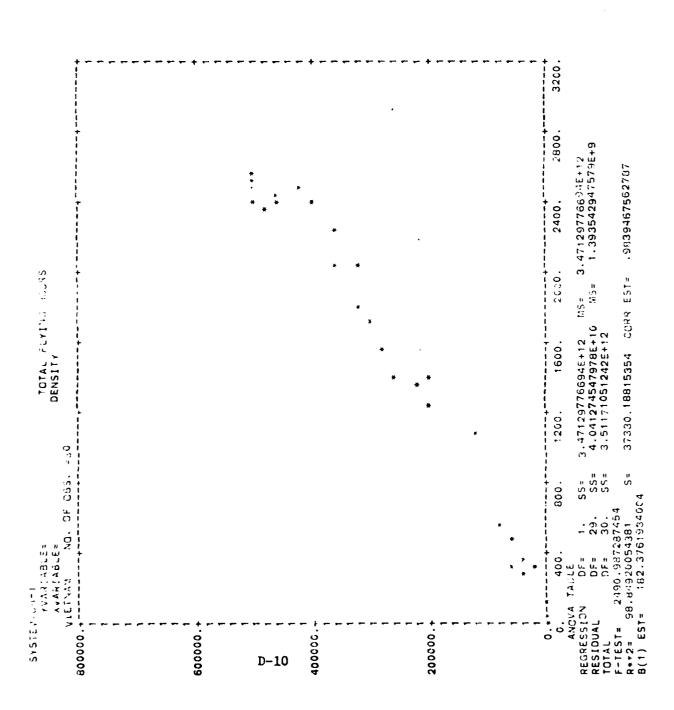
250.0000 300.000 350.0000 400.0000 5± 1.015602400786E+10 17892225.08777 4229,920222388 CORR EJT# .9733678581256 TOTAL FLYING HOURS DENSITY 150.0000 200.0000 1.015602400786E+10 232590926.141 MS= 1.039532934E+10 NG. OF 335. =14 100.0000 ;; \$5°= \$5°= \$5°= C.0000 50.0000 100.0

ANGVA TABLE
REGRESSIDN DF= 1. SST
RESIDUAL
TOTAL
F-TEST= 507.3221410936
R**2= 97.76102253765
B(1) EST= 143.2569228895 XVARIABLE = VIETNAM NO Y.ARIABLES 80000.+----SYSTEM - CRIS 4 20002. .00009 40000 D-6





6400. .0. 4800. seco. 1.85738687473 E-12 2.7896198715866+3 52816.64031403 | CDRR [STR | .6125035831107 *** **** .000; TOTAL FLYING HOURS DENSITY 1.357386874739E+12 1.783356825790E+11 2.035922527319E+12 3200. 2400. သ က က က က က က က Š ALGVA TVJE PESPESSION DF 1. SS. PESTURAL DF 54. 51. TOTAL DF 665,820P50154 F-TEST 665,820P50154 8.12 EST 91.207246379 SYSTEM=UH-1 0 600000 200002 400000 D-9



40000 3.000. 1,152808289195+11 59575592,05751 7 45.549878319 | COKR | DATE | 1978874041374 30000. TOTAL NUMBER OF SORTIES TOTAL FLYING HOURS .0.35. 4.15490828919E+11 3.233734970025E+9 1.18720563839E+11 20000. 15200. NO. OF C85, =55 10000: 1024,83793527 7,2711286354 2,030871549 5 SYSTEM= A:1- 1 40000 123006.4 80000 D-11

160000. 140000. 1,665550778453C+11 114508972,4636 .9767252218415 120000. TOTAL NUMBER OF SORTIES TOTAL FLYING HOURS 1000001 10700.886527 CORR EST= 1.665550778453E+11 2.404888421736E+9 1.63959766267E+11 80000 60000 NO. OF 095. =22 0. 20000. 40000. ANDVA TABLE
REGRESSION DF= 1. SS= 1
RESIDUAL DF= 21. SS= 1
TOTAL DF= 22. SS= F-TEST= 1454.515521901
R**2= 98.57676861488 S= 8(1) EST= 1.583709173628 YVARIABLE= XVACIABLE= VIETVAN NG. 320000.+----SYSTEM-AH-1 4.00008 240000. 160000. D-12

32300. 8.708460415611E+10 47530305.1093 UP14.22256919 CO4 1514 .8657190774948 24050. TOTAL NUMBER OF SURTIES TOTAL FLYING HOURS 8.7%54r6418611E+10 2.99/u09221896E+9 %.0012013408E+10 160000 12000. ₹9 = NO. 31 085. #L DF= 63. 53± DF= 64. 55± = 1832.101145902 = 0.1757975490° ST= ...21935588°802 \$\$ \$\$ \$\$ XVARIABLES WORLDWIDE 160000,+------TVARIABLE= 4000 ANGVA TASLE REGRESSION DF= RESIOUSL DF= TOTAL DF= F-TEST= 1832.15 R++2= SULTSTSTS SYSTEM=CH47 40000+ 120050. 80000 Ě D-13

TOTAL NUMBER OF SCHILES TOTAL FLYING HOUSE YVARIABLE= XVARIABLE= VIETNAM NO. SYSTEM: Crid7

70000 3.14442630492E+11 163061991.1143 12769.57286343 COPR EST# .9756337256978 60000 50000 3,14442630492E+11 4,239611768973E+9 3,18582242261E+11 40000. 30000. , \$\$± \$\$\$ \$§§ 20000. AEGRESION DF= 1, SS= RESIDUAL DF= 2A, SS TOTAL DF= 27, SS FHTEST= 1/2√50251013 R**2= S8.CvJ64292115 S B(1) EST= 2,994669915u07 O. 10000. ANOVA TABLE 320000.+----+.00008 D-14 09 240000.

TOTAL NUMBER OF SORTIES TOTAL FLYING HOURS NO. OF J35. =47 XVAHIABLE# WORLD#IDE YVARIABLE= SYSTEM=0:158 500000.+---375000.4 125000. 250000 D-15

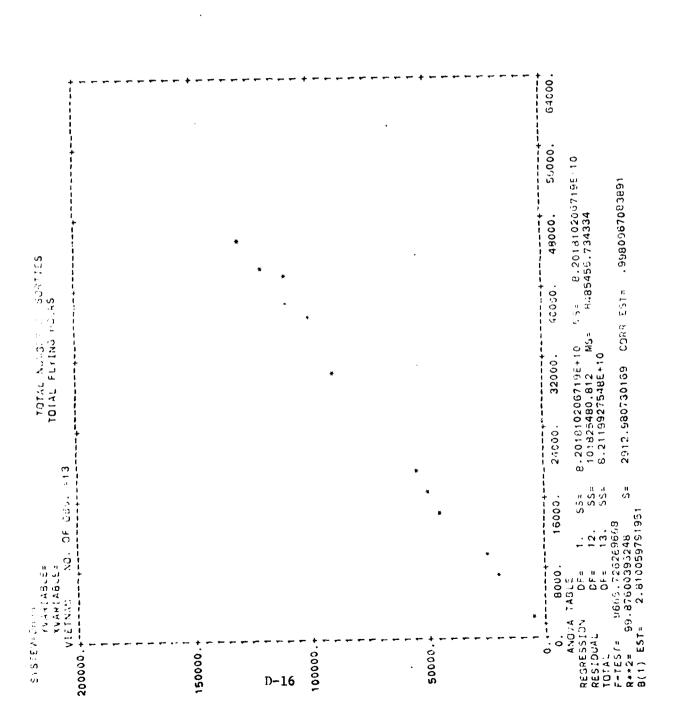
1.869011507445E+12 257812727.7563 0000. 120000. 1000001 # 27 71 14 80000. 60306. 40000

160000.

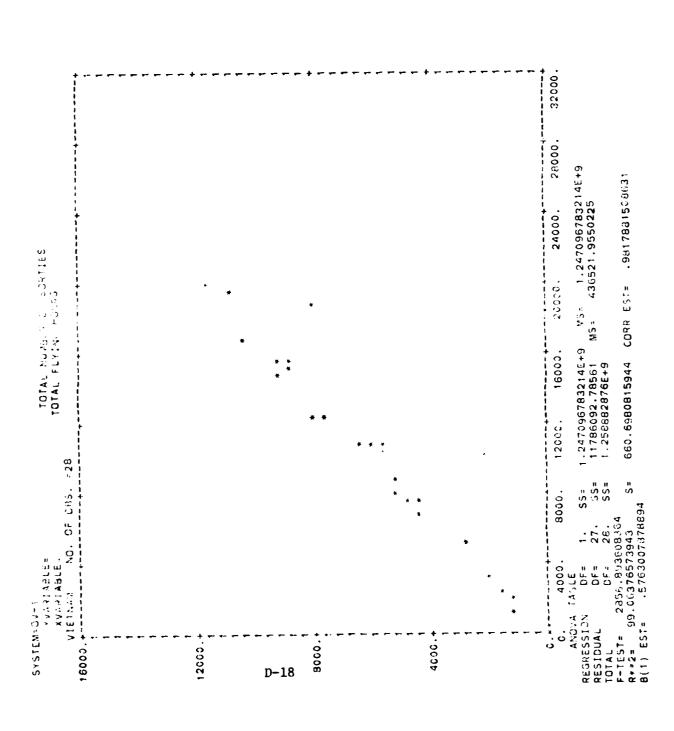
140000.

1.8090:1507445E+12 1.136036547725E+10 1.820070882922E+12 1 (0 .5 (0 .5 (0 .5) ## 2000. ## 2000. ## 40000. ## 2000. ## 2000. ## 2000. ## 2000. ## 2000. ## 2000. ## 2000. ## 2000. ## 2000. ## 2000. ## 2000. ## 2000. ## 2000. ## 2000. ## 2000. ## 2000. ## 2000. ## 2000. ## 2000. ## 2000. ## 2000. ## 2000. ## 2000. ## 2000. ## 2000. ## 2000. ## 2000. ## 2000. ## 2000. ## 2000. ## 2000. ## 2000. ## 2000. ## 2000. ## 2000. ## 2000. ## 2000. ## 2000. ## 2000. ## 2000. ## 2000. ## 2000. ## 2000. ## 2000. ## 2000. ## 2000. ## 2000. ## 2000. ## 2000. ## 2000. ## 2000. ## 2000. ## 2000. ## 2000. ## 2000. ## 2000. ## 2000. ## 2000. ## 2000. ## 2000. ## 2000. ## 2000. ## 2000. ## 2000. ## 2000. ## 2000. ## 2000. ## 2000. ## 2000. ## 2000. ## 2000. ## 2000. ## 2000. ## 2000. ## 2000. ## 2000. ## 2000. ## 2000. ## 2000. ## 2000. ## 2000. ## 2000. ## 2000. ## 2000. ## 2000. ## 2000. ## 2000. ## 2000. ## 2000. ## 2000. ## 2000. ## 2000. ## 2000. ## 2000. ## 2000. ## 2000. ## 2000. ## 2000. ## 2000. ## 2000. ## 2000. ## 2000. ## 2000. ## 2000. ## 2000. ## 2000. ## 2000. ## 2000. ## 2000. ## 2000. ## 2000. ## 2000. ## 2000. ## 2000. ## 2000. ## 2000. ## 2000. ## 2000. ## 2000. ## 2000. ## 2000. ## 2000. ## 2000. ## 2000. ## 2000. ## 2000. ## 2000. ## 2000. ## 2000. ## 2000. ## 2000. ## 2000. ## 2000. ## 2000. ## 2000. ## 2000. ## 2000. ## 2000. ## 2000. ## 2000. ## 2000. ## 2000. ## 2000. ## 2000. ## 2000. ## 2000. ## 2000. ## 2000. ## 2000. ## 2000. ## 2000. ## 2000. ## 2000. ## 2000. ## 2000. ## 2000. ## 2000. ## 2000. ## 2000. ## 2000. ## 2000. ## 2000. ## 2000. ## 2000. ## 2000. ## 2000. ## 2000. ## 2000. ## 2000. ## 2000. ## 2000. ## 2000. ## 2000. ## 2000. ## 2000. ## 2000. ## 2000. ## 2000. ## 2000. ## 2000. ## 2000. ## 2000. ## 2000. ## 2000. ## 2000. ## 2000. ## 2000. ## 2000. ## 2000. ## 2000. ## 2000. ## 2000. ## 2000. ## 2000. ## 2000. ## 2000. ## 2000. ## 2000. ## 2000. ## 2000. ## 2000. ## 2000. ## 2000. ## 2000. ## 2000. ## 2000. ## 2000. ## 2000. ## 2000. ## 2000. ## 2000. ## 2000. ## 2000. ## 2000. ## 2000. ## 2000. ## 2000. ## 2000. ## 2000. ## 2000. ## 2000. ## 2000. ## 2000. ## 2000. ##

15055,54781504 CORP 897: 1955175845451



20000 17500. 0,162601563754E+9 1372245,305424 15000. TOTAL NUMBER OF SORTIES TOTAL FLYING HOURS · :: 3.1830015837582+9 >6487604,24173 3.270556168E+3 100gn. 7500. NO. OF 085. =64 REGRESSION DF= 1. 55= 0 RESIDUAL DF= 1. 55= 0 RESIDUAL DF= 63. 05= 0 TOTAL DS: 00- 05= 04. 55= 0 F-TEST= 03: 00-0727228 R**2= 07:00-07372562 XVARIABLE XVARIABLE WORLDWIDE TO WORLDWIDE TO THE TOTAL T 0. * ----SYSTEM=DV-1 5000.+ 15003.4 100001 D-17



640000 .600044 1,01868927470355+13 7,599583497208:+9 0.151-81876736. -- 1.20.7909 180000. TOTAL NUMBER OF SORTIES TOTAL FLYING HOURS 1.918989274703E+13 4.787737004501E+14 1.000766130740E+13 240000, 320000. . 7175,59004221 NO. OF 085, =64 163330. YVARIABLE= XVARIABLE= WORLDWIDE A.0003E+06+---------#ESRESSION CERE RESIDAN CERE TOTAL CERE F-TESTE 2504.90208 1.0000E+06+ 5.0000E+C5+ 1.5000E+06 D-19 . 0

SYSTEM=UH-1

TOTAL NOTE 1 1/2 SONTES TOTAL FLYING 1/3 ALLEAST ASK SYSTEMBURE 3.2000E+06+ 2.4000E+06+ 1.600CE+08+ 8.0000E+05+ D-20

2.96374712⁹727E+13 7.818901393304E+9 480000. 320000.

240000.

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AMOVA TACLE
RESEDUAL
TOTAL

F-TEST= R**2= 59 B(1) EST=

640000.

.9851907323033 65424,55198249 CC4R ESTA 2.5637471299275+13 2.189292390125E+11 2.035640053828E+13 0730 1490440438 090.20472594464 090.20472594464 074 4.013780509918

# APPENDIX E

# 1352 FAILURE DATA

# REGRESSION ANALYSIS

Y VARIABLE	X VARIABLE
(DEPENDENT)	(INDEPENDENT)
	HC A CD

RELIABILITY	USAGE
AVAILABILITY	USAGE

RELIABILITY

(BY AIRCRAFT)

AVAILABILITY

LENGTH OF SORTIES

LENGTH OF SORTIES

(BY AIRCRAFT)

RELIABILITY TOTAL FLYING HOURS
AVAILABILITY TOTAL FLYING HOURS

AVAILABILITY RELIABILITY

.0640 .0560 . 0185053155475 . 0181:827912346 21.521.53.5 0360. 946685 - - 5399FB AVERAGE RELIABILITY AVERAGE USAGE 10.54 10.54 .4185653155475 .200087935404 1.00074109091 INTERCENT ESTA CORMELA LON ESTA NO. OF C85, =55 YVARIABLE= XVARIABLE= WG2LOWIDE SYSTEM=HH-1 1.6000+----0.600000000 お自立されるのはなる。 こうこうしょう .8000+ .4000+ 1.2000+ -1831-A**2= E-1

.1600 .1400 .1200 .1505375538**597** .01054941**2549**13 .1000 .4562400669305 .6360731730137 AVERAGE RELIABLUTTY AVERAGE USAGE 0080. MS= MS= .1505375538597 .2215376635316 .3720752173913 INTERCEPT EST= CORRELATION EST# 0090. NO. OF 085. -23 0.400 100AL DF= 21, 55 AL DF= 22, 58 EST= (4.26975702757 2= 40,45890369026 C .1027103332149 0.0000---------AVARIABLES XVARIABLES VIETAAN NO ACCRESION DES RESIDUAL TOTAL 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F = 0 0 F 1.8000+----SYSTEM=AH-1 SLOPE EST# F-TEST= R*+2= 4 +0008. +00007. 1.20004 E-2

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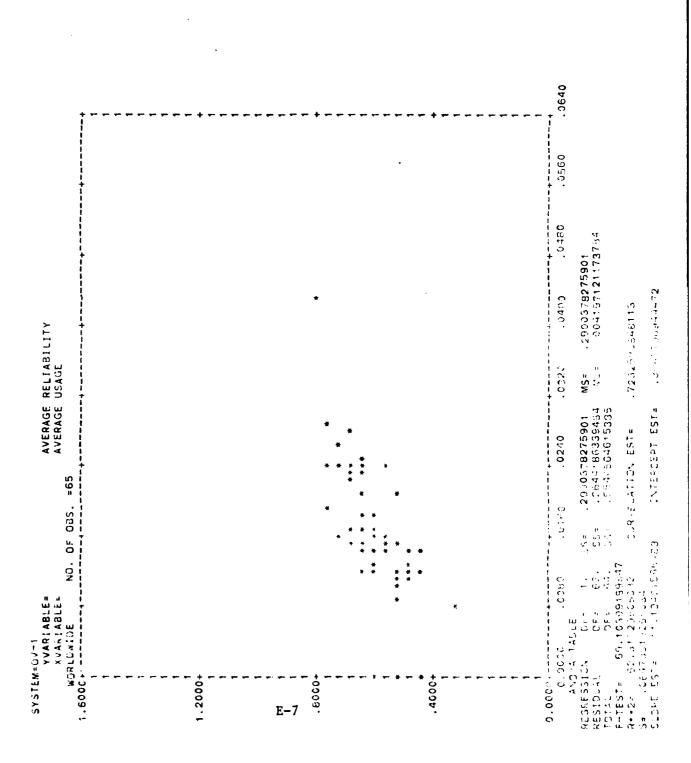
The state of the s

SYSIEM=CH47 YVARIABLE= XVARIABLE= WORLUWIDE NO. OF OUS. =65	AVERAGE RELIABILITY AVERAGE USAGE	RELIABIL	I T V	8 9 9 9 9 9 8 9	į
	*** * *		•		
. 0080 . 0160	.0240		800,	0350.	.0640
ANDVA TADLE SSION DES 1. SSE .0143 CAL SSE 33. SSE .70	.014255885296601 .9020805215494 .7050453346154	# # 7 :0 <b>≥</b> 15	.01025686296601 .01210140510555		
12,7451 52074 CDR-1	(II 4 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1	11207.4	0.031		
37.701	FOR FORES	·	88 20,890, 97 and		
57.59018400	11.01.11	1.0	0.000 d		

. 1600 . 1400 .1200 .03671936898968 .005615441852045 .1000 5.05.233209117 .4482843:41217 AVERACE RELIBBLITY AVERAGE USAGE : S : S : S .0800 .0.571936898963 .1460014881532 .1827208571429 INTERCEPT EST= CORRELATION EST= .0600 12. 3F 35 . 128 .0400 RECRESSION DF= 1. SS= RESIDUAL DF= 26. SS= TGTAL DF= 27. SS= F-TEST= 0.53090959566 R**2= 20.09508262875 COR S= .07493625192152 .0200 ## 10 mm 10 000 0 E E E 0.0000 ANDVA TABLE SVSTEN- 0:47 0.0000. 40004 1.6000.1 1.2000+ . B0004 E-4

.008u .01u) .008u .01u) .008u .01u) .008u .01u) .01u .55 .0008 .01	SYSTEM=OH58 YVARIABLE= XVARIABLE= WORLDWIDE	085. =48	AVER AVER	RELIABILITY USAGE	ILITY		
.008u .01u .024u .032u .046u .048u .056u .048u .056u .046u .048u .056u .056u .046u .048u .056u .046u .048u .056u .046u .048u .056u .056u .046u .048u .056u .048u .056u .048u .056u .048u .056u .048u .04au .				*	•		+ ~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~
DF= 1, \$5= ,0096053622620 MS= DF= 40, 55= ,0070508637737 MD. CF= 41, 55= ,04606725 014104420448 4418261757 CORRELATION EST= -,204912 9557331 -4.471294275542 INTERCEPT EST= ,84.4.4.4.4.4.4.4.4.4.4.4.4.4.4.4.4.4.4.	14. 	cato.	i !	032		!	.0640
4818261757 CORRELATION EST#27430 9057331 -4.4717942875342 INTERCEPT EST# .P.1	68310N DF= 1	1 (0 (1) 1 (0 (1) 2 (1)	186 <b>2262</b> 9 18637737 25	W. H.	.03769838622629 .0197:019209073		
	100 d		EST.	₩ ₩ ₩	54R359 .82776565		

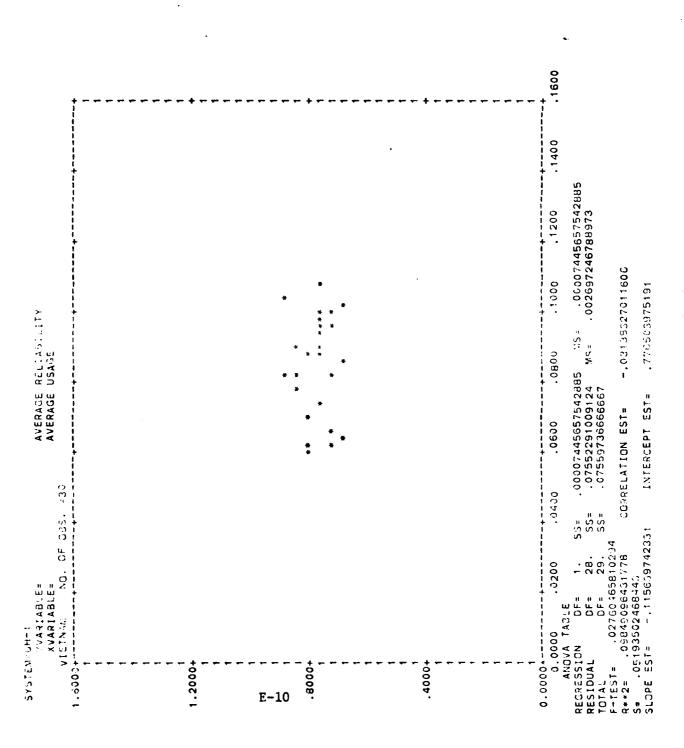
.1000 .0875 .0750 .01082009138223 .6736341684822 .0625 .09347123073933 AVERAGE RELIABILITY AVERAGE USAGE MS= .001144403413217 .1298410965868 .1309855 INTERCEPT EST= CORPELATION ESTA .0375 5.8 = 5.8 = 8.8 = 8.8 REGRESSION DF= 1. SS= RESIDUAL DF= 12. SS= TOTAL DF= 13. SS= F-TEST= .1057665201512 R**2= .8736870875926 CDF S* .1040196582471 SLOPE EST= .8527596223675 ANOVA TABLE REGRESSION DE-RESIDUAL DE-1.6000+----3YJJEW:00098 0.0000-----0.000.0 .4000+ 1.2000+ 8000. E-6



. 1600 .1400 .1200 .:416639089608 .01043945675624 .47845-3131081 000: .578340231997 AVERAGE RELIBETETTY AVERAGE USAGE MS= .1416639089608 .2413653324185 .4235292413793 INTERCEPT EST-CORRECATION ESTA .0600 NO. OF 033. =29 5.55 5.53 5.54 5.54 4,108425913752 REGRESSION DF= 1. S RESIDUAL DF= 27. TOTAL DF= 27. F-TEST= 13.57001605369 R**2= 33.44343646536 S* .1021736547950 0.0000*-----YVARIABLES XVARIABLES VIETNAS SYSTEM GV" 1.6000+--.4000+ 1.2000+ +0008. E-8

.0800 .0700 0000. .01116961907045 .01025335549705 INTERCEPT ESTA . 7400550853 .0500 -,1363747628321 AVERAGE RELIABILITY AVERAGE USAGE 至 25 25 25 25 25 .0400 .01116961907045 .6459613963142 .6571310153346 SOFFILL FION ESTA 0300 NO. OF 0.35. =65 .0203 55° 50° -1.3062695571 ANGVA TABLE 1. S RESIDUAL DE= 1. S RESIDUAL DE= 63. TOTAL DE: 1.089502713702 P=72 1.09502713702 S= 101214539793 S= 101214539793 0100 YVARIABLE= XVARIABLE= WORLD-JINE SYSTEM=UH-1 00000.0 0.0000. .4600+ 1.2000+ .8005 E-9

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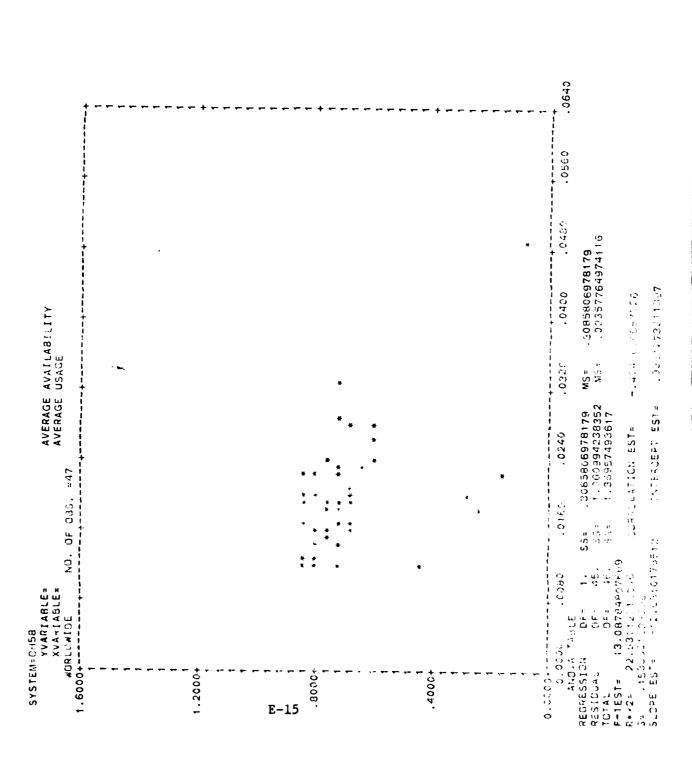


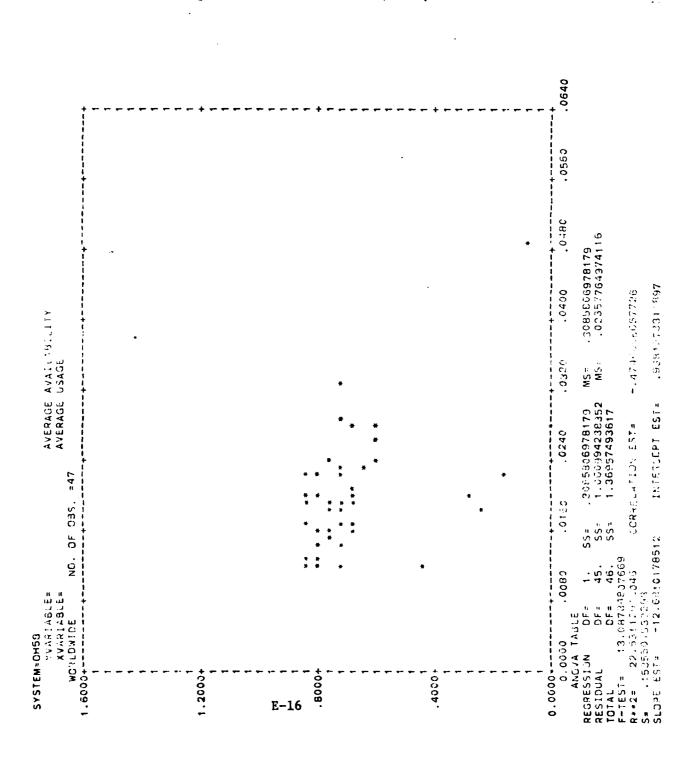
.0640 .7661193139064 .02775512152744 . 27 1 71 641 4778 .0400 ₹1550 1 7 108345 AVERAGE AVAILABILITY AVERAGE USAGE MS# THESCEPT EST# SSs 7758/190139064 SSs 1.744226319427 SS 2.010945333333 CREENTION ESTA . 0240 NJ. OF OBS. =54 F-TEST= 77.57299449044 4**2= 34.08+13070739 UC S= ,16,86+38+04887 \$LOPE EST= -10.02183287:54 YVARIABLE= XVARIABLE= WORLDWIDE 0.0000 ANOVA TABLE ANOVA TABLE RESIDUAL TOTAL F-TESTE 27.57297 SYSTEM=AH-1 .4000+ E-11. 1.2000+ 00000.0

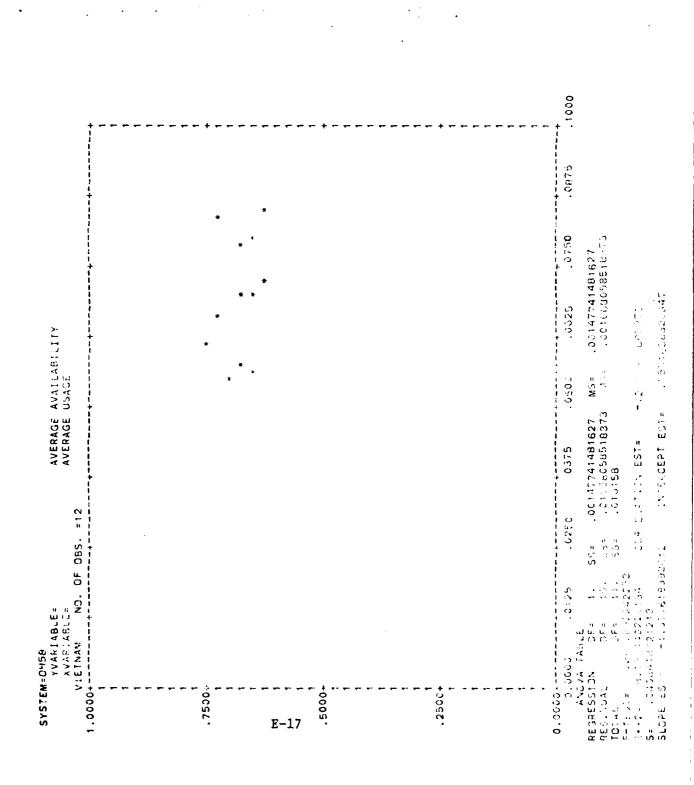
# * * * * * * * * * * * * * * * * * * *	#VERAGE USAGE  AVERAGE  AVERAGE			.1400 .1600			
# 1 0000 1 0000 0 0000 0 0000 0 0000 0 0 0000 0 0	AVERAGE  AVERAGE  AVERAGE  AVERAGE  * * * * * * * * * * * * * * * * * * *	 P-1		1000	.02184320639643 .03814682868092	2157272	
	*  *  *  *  *  *  *  *  *  *  *  *  *	AVERAGE NAACH	* * *				

.0640 .0560 0810 .238376414527 .01749163895924 . 0400 #1.04500 FUNE 40. -AVERAGE AVAILABILITY AVERAGE USAGE # 17 # 17 # 27 INTERCEPT ESTA 0530 ETSE VOITE STE NO. OF CAS. =64 -8.468.20.20.00-YVARIABLE= XVARIABLE= WORLJWIDE SYSTEM-CH47 51028 TST-0.0000.0 2.4000+ *000a. 1.63004 E-13

.1600 .1400 .00051**79970**791234 .023938**256980**47 1200 .02885099951974 .5475543366292 .1000 AVERAGE ATALL LITY AVERAGE USAGE MS= MS= .0800 .0005:79970791284 .6223946814923 .6229126785714 INTERCEPT EST= CSPRELATION EST= 0090. 13. Of JBS. UF= 27. S ...LEST= .02163338537257 R**2= .082157254:30:2 S= .1547199307797 SLOPE EST= .27033 .2783499245748 .0200 704 . 148.5 = XVAR.148.5 = XVAR.148.5 = XV.0 AMOVA TABLE
RESRESSION OF=
RESIDUAL DF=
TOTAL 5YSTE11-0147 0.0000. 0,000 .7500+ E-14". .2500+







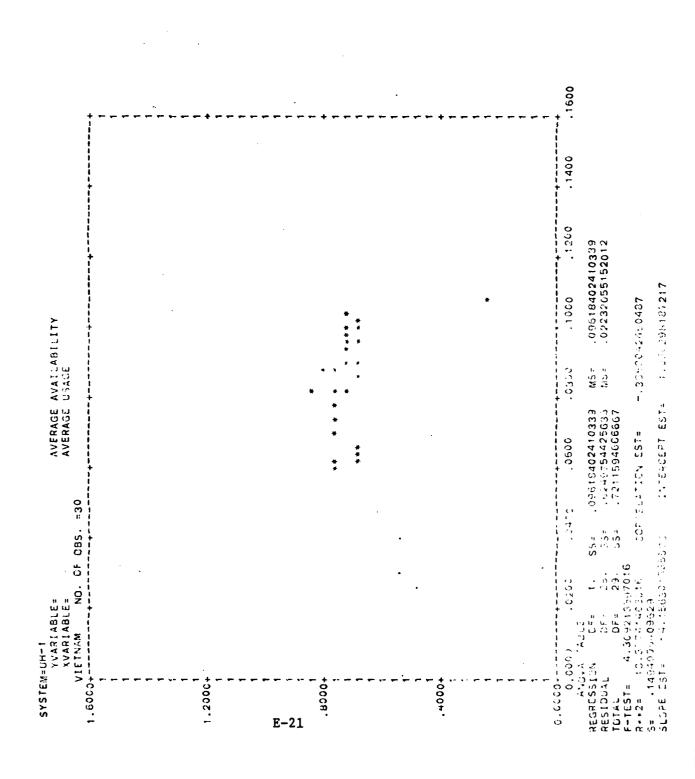
.0560 0480 .004873414639446 .5629037661354 -.07470088642462 0000 AVERAGE AUNICIBILITY AVERAGE GEACE INTERCEPT EST= .004872414639446 .8683279639759 .8737013846154 CORRELATION ESTA .0240 .0163 -1.451052204778 1) E 63. DF= 63. DF= 64. .35.081990547. .1174009525107. .0080 ANDVA TABLE SYSTEMOVE 0.0000 REGRESSION RESIDUAL TOTAL FITESTE R**2= .55 S# .11740 SLOPE EST= ---------.4000+ 1.2000+ .80004 1.6003 E-18

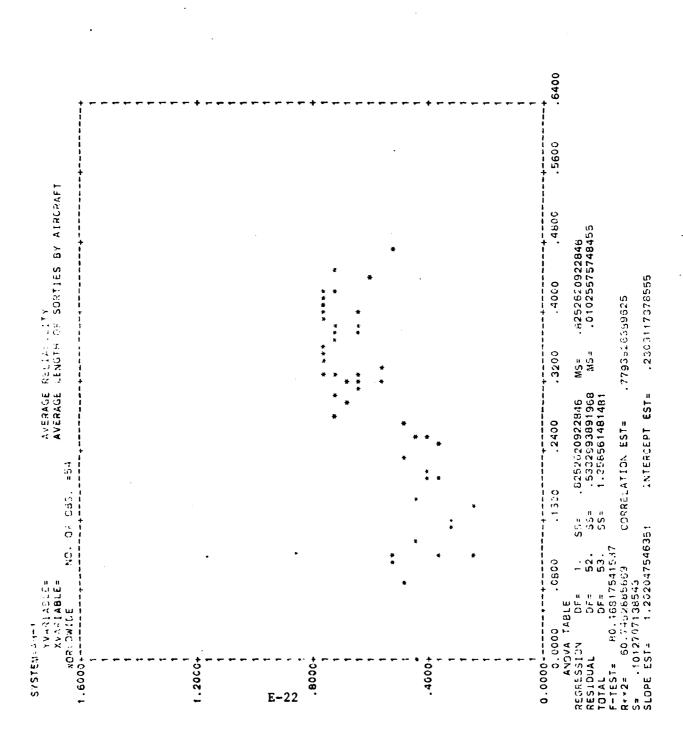
.1600 .1400 .1200 . 1842503623546 . 02423912578763 .33.1102463581 1000 1461710174314 AVERAGE AVAILABILITY AVERAGE USAGE . 0800 INTERCEPT EST# .1802563623546 .8539160062601 .8300847536207 CORRELATION ESTE 0090. NO. OF OBS. =29 0400 20 0 0 10 0 0 10 0 0 4,634400473311 28. 7.421550550467 44.27 21.86953494965 5. . .15584707697258 5.005 E574 4.501777 YVARIABLE= XVARIABLE= VIETNAM NO 1.6000+-----0.0000-----SYSTEM=07-1 0.000.0 .4000+ E-19 1.2000+

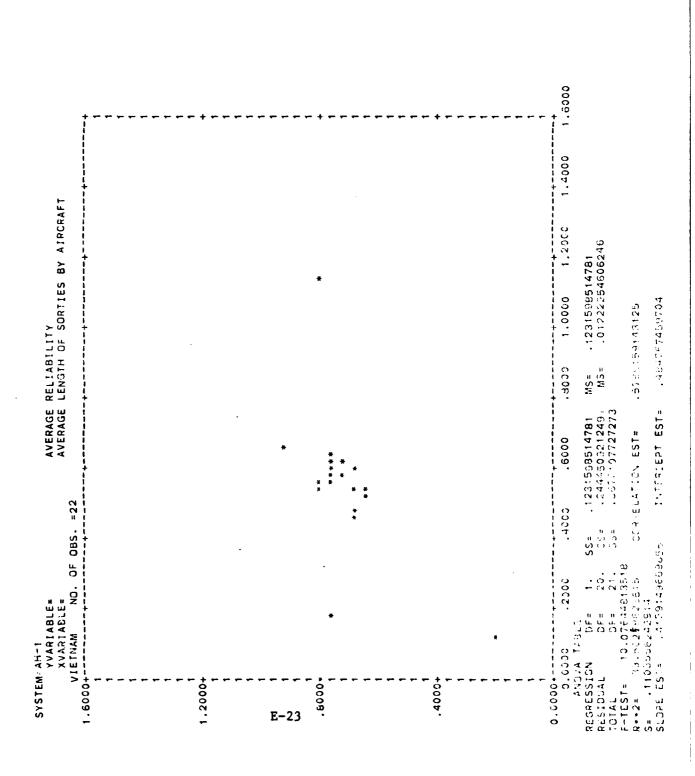
FUNTY YVAM(AG.Es XVAM(AG.Es AVARIABLES RCDWILE NO	AVERAGE AVA! AVERAGE USAG	avai_ao;_;;y usade	•
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. 0100	.0200 .0300 .0400	0090. 0030.	0080. 000
5400A PICE 1. S 5100AL DE= 63.	55- 554647438468 MS= 55= 1,214145846147 MS= 55= 1,748792984615	.554647438468 5= .01927215152615	
1269428 6356636 227	CORRELATION EST#	+.5529225577872	
.205954	18 INTERCEPT EST:	. 843 395784966	
-0.205251796418	:∂ INTERCEPT EST=	84V 395784966	

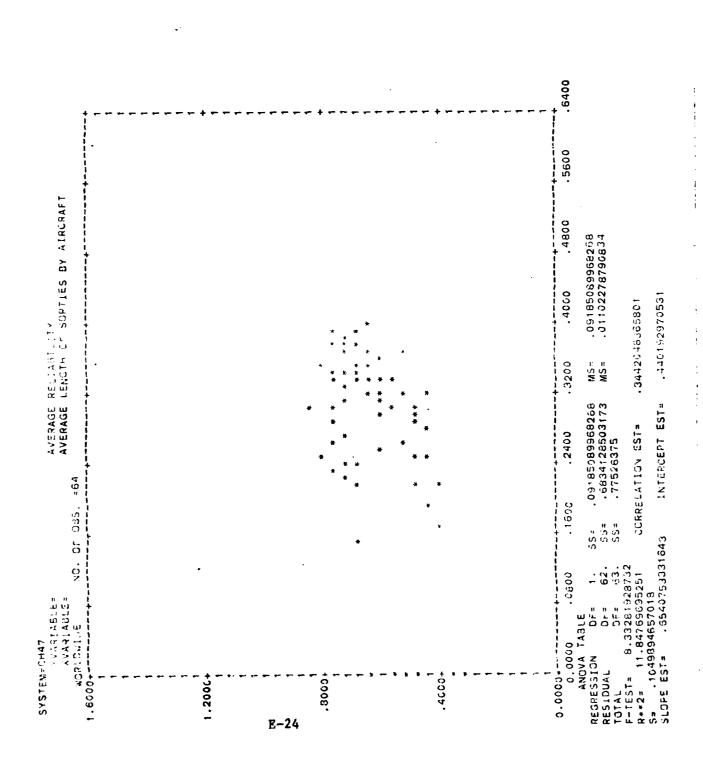
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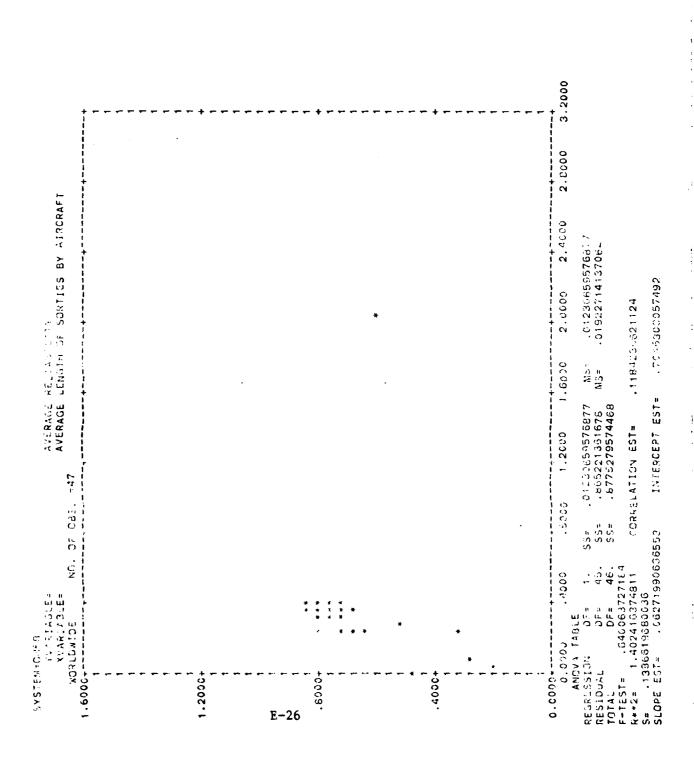








1.0000 .8750 AVERAGE RELIABILITY
AVERAGE LENGTH OF SORTIES BY ATRORAFT .008792555499368 .00081012446612 .8133148740900 .6250 . 22--- 21 -- 178.772 .002792555496868 .0728281111608 .108620006667 INTERCEPT ESTR CORPELATION ESTE .3750 NG. 3F 085. =27 SSI .1753156274207 1,2718642009A8 4,841184643152 8311,270%4401 0.000.0 0.000.0 ANGVA TABLE YVARIABLE= XVARIABLE= 0.0000.0 SYSTEM=CH47 SECRESSION
40537048,
1078,
11078,
111651=
11.8 SLOPE ESTA 1.2000+ .40004 .83004 E-25



1.0000 .8750 AVERAGE RELIABILITY
AVERAGE LENGTH OF SORTIES BY ATRCRAFT .0001945264162464 .002872105990131 7500 .6250 .0782272 049475 POTODOC DIAC. EST TREDEBLY % 1- ∞ 1- ∞ .5000 .00013457.44162464 .03184316589145 .03178769230769 CORPELATION ESTA .3750 NO. OF 085. =13 # 12 (2) (3) (3) (3) (5) (5) (7) . 0327-0000-4720 XVARIABLE= VIE'NAM NO YVARIABLE= 0.0000------SYSTEM= Jin58 0000.0 1.6000+--1.2000+ 40004 .8000 E-27

State of the state

1.6000 1.4000 AVERAGE PELIASTLIFY
AVERAGE LENSTH OF SORTIES BY ATRORAFT 1,2000 .09746137253908 .007255726854208 .3978477120566 1.0000 .4219030021352 = 5 = 5 = 5 = 5 .09746;37253508 .4498550649609 .5473164375 INTERCEPT ESTA .6000 CORRELATION EST= NO. OF Oas. 3366. 62. 63. .2000 REGRESSION DESTORAL SYSTEM-CU-1 0.0000----1.2000+ .4CCO+ . 9C03 E-28

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:

3.2000 2.8000 AVERAGE RELIASILITY
AVERAGE LENGIN OF SORTIES BY AIRCRAFT 2,4000 .1401677146707 .01089012498513 2.0000 .40735.4464465 .57840.0021854 #S= .1401677146707 .283143249615 .4233109642857 INTERCEPT EST= CLANGLATICN ESTE 1,2000 NO. OF 335. : 28 . 2276909049718 YVARIABLE= XVARIABLE= VIETNAM NO A10.4 A20.8 AES104 OFF AES10.41 OFF TOTAL 1.6000+-----SYSTEM=0V-1 SLIPE EST .4000÷ 1.2000+ . B0009. 0.0000 4177511 E-29

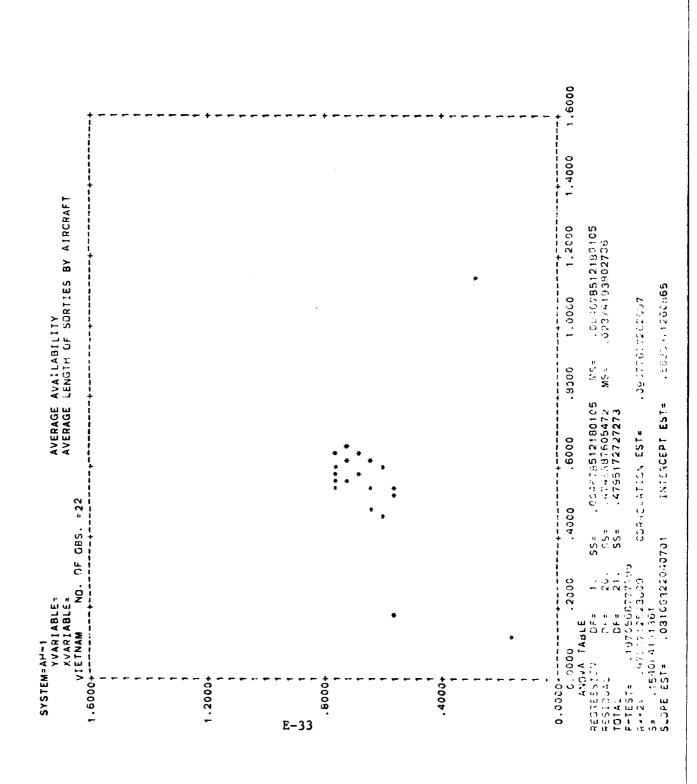
.6400 .5600 AVERAGE RELIABILITY
AVERAGE LENGTH OF SORTIES BY AIPCRAFT .4800 .1223237030**826** .007910987571248 .4000 .4475145019874 .3230 MS∸ MS= .1228267080826 .4904812294174 .613307937**5** CURRELATION ESTR .2400 16. 0- 033. ±64 ć RECRESSION DF= 1. SS= RESTDUAL DF= 52. SS= TOTAL DF= 53. SS= FHTEST= 15.52609040712 R**2= 20.0203239439 CLAF S= .08994375261365 S= .08994375261365 .0200 REGRESSICY DF= RESIDUAL DF= TOTAL DF= F=TEST= 15.52608 30. vojech 30. vojech 1.6000+1 SYSTEM=U-1-1 0.0000. 40004 1.2000+ .80004 E-30

.4613097324421

INTERCEPT EST=

.8000 .7000 AVERAGE RELIABILITY
AVERAGE LENGTH OF SORTIES BY AIRCRAFT 0009. .01593816663458 .002173802257468 ,e0% 30%1204983 .5000 .462:21.209387 25 .4500 .01593916663458 .04833266095162 .0144,002758621 TATERCEPT ESTR CORRECATION ESTR 3000 NO. OF CES. =29 .4217154006610 YVARIABLE= XVARIABLE= VIETNAM NO 0,0000 ANGVA TABLE REGRESSION DF= 5= .04c...40... \$LOPE EST- .4 1.6000+-----SYSTEM=UH-1 0.0000----RESIDUAL TOTAL .4000+ 1.2000+ F-TEST= .8000+ R . . K . E-31

BY AIRCRAFT	:	. 560	308 33763
AVERAGE AUTOMOTICS SORTIES E		.3200 .4000	003 MS= 1.058825577008 6326 MS= .02215614723703 3333
0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000		009:	\$5= 1.05(825677003 \$5= 1.152119656326 \$5= 2.213945333333 09
SYSTEW=AH-1 YARTABLEE XVATABLEE 1.6006+	E-32	.030	REGRESSION OFF 1. SS RESIDUAL OFF 52. S FOTAL OFF 53. S

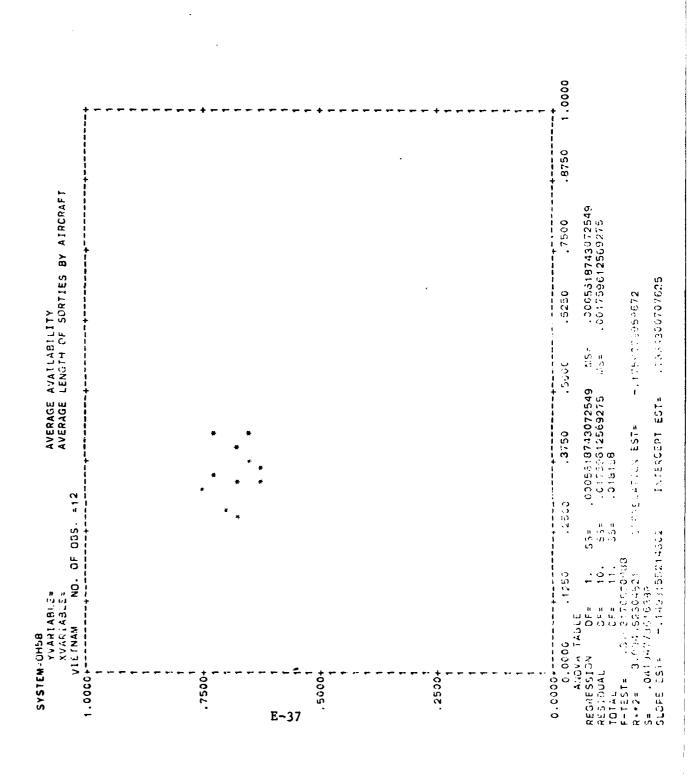


.6400 . 5600 AVERAGE AVAILAMILITY
AVERAGE LENGTH OF SORTIES BY ATRORAFT 4800 .5229691161834 .0133692360293 .01213087267024 .4000 .6223377362358 :A.5.≅ :#S.= .5239691161834 .8288926338166 1.35286175 INTERCEPT ESTA CORRELATION EST* .2400 NJ. 17F 045. =64 .1600 1.562203183765 DF= 63. 63. F**2* 39.19215092284 F**2* 38.73042579431 S* .1156254125584 SLOPE EST= 1 F .0800 YVARIABLEE XVARIABLEE WGRLDWIDE 3.20004----+-ANGVA TABLE SYSTEM-CH47 0.0000.0 REGRESSION RESIDUAL .8000* 2.4000+ 1.6003+ E-34

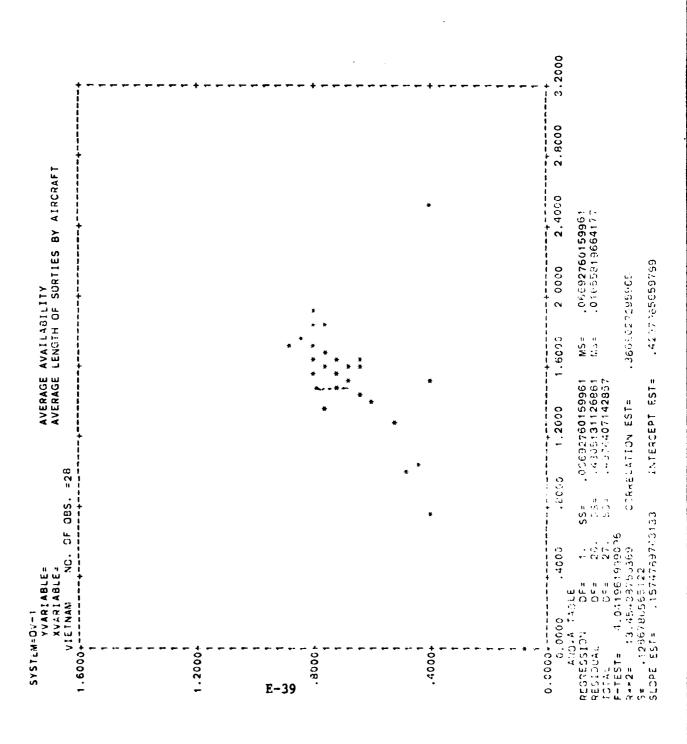
1.0000 . 6750 AVERAGE AVAILABILITY
AVERAGE LENGTH OF SORTIES BY AIRCRAFT .7500 .05014361**72**6346 .009814080494347 .7021010273714 .0250 .5000 MS= ₩5= .05014261726346 .2453520123662 .2954356296296 INTERCEPT ESTA CORPUTATION ESTA NO. OF 085. =27 .2500 RESIDUAL DE= 25. S.=
TOTAL DF= 26. S.=
F-IEST= 0.10935459422
R*2= 16.07902613311 CD9
S= 00.00 0.04107583
SLOPE EST* -.4258319527651 YVARIABLE= XVARIABLE= VIETNAM NO ANDVA TABLE SYSTEM=CH47 1.0000+ 0.0000----RECRESSION .2500+ .75004 .5000+ E-35

3.2000 2.8000 AVERAGE AVAILABILITY
AVERAGE LENGTH OF SORTIES BY AIRCRAFT 2,4000 .1426520882684 .7678:6948239 2.0000 -.3227350747607 1.6000 INTERCEPT EST-. 1426520582684 1.226922847902 1.36957493617 CORRELATION EST# 1.2000 <u>ে</u> NO. OF 085. . 3000 \$5° \$5° \$5° F-TEST= 5.232068:63909 04-2= 10.41579284937 CO 5= .1651210228:54 SLOPE EST= -.2135381620033 46. 4000 YVARIABLE= NVARIABLE= NORTURIUE 0.0000 ANGVA TABLE REGRESSION POR RESIDUAL SYSTEW=0H5H TOTAL F-TEST= .8000 .4000+ 1.20001 E-36

. .



1.6000 1.4000 AVERAGE AVALUTBULITA AVERAGE LENGTH OF SORTIES BY AIRCRAFT 1,2000 .0000442906164113 .3946306161281 1.0000 .3195,00054618 M5= M5:: .8300 .06650925532463 .5854601821754 .6526854375 INTERCEPT ESTE CORRELATION ESTA .6000 NO. OF Cas. #64 .4360 R**2= 16.21505555311 5= .097:7461627149 SLOPE EST= .1907374173605 7.053893596628 .2000 XVARIABLE MORLUMICE SYSTEMBLE +0001. 1.2000+ .80004 E-38



.6400 .5600 AVERAGE AVAILMOLLITY
AVERAGE LENSTY OF SORTIES BY APPRART .4800 . c004624692565 .01224456904425 000t. 135165547496 .6645585400552 %S= MS= .3200 .8004624692565 .7591632807435 1.35962575 INTERCEPT EST= CORKILATION EST= .2400 NO. OF DES. 104 DF= 62.
DF= 65.
SF= 65.
SF= 65.
SF= 65.
SF= 44.16380531602
SF= 110.551808288
SLOPE EST= 1.4430 1,443915228334 080. YVARIABLE= XVARIABLE= %D%LDWIDE ANGVA TABLE RESPESSION DER SYSTEM 6.17 0.000.0 +0008. 2.4000+ 1.60004 E-40

.3000 AVERAGE AVALLABILITY
AVERAGE LENGTH OF SORTIES BY AIRCRAFT ------.6000 . 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% | 60% .5464 55120035 . 5000 T. 17 1743 01 01 001 .4000 55= 0.700905203021 55= 1207362952456 50= 12067454432759 #184 - L1141314E .3000 NO. OF 035. ±29 ** 61210 or 144.5 YVARIABLE= XVARIABLE= VIETNAM NO. 1.3000+-----SYSTEM=UH-1 5.39E E33-+0008. 1.2000+ .4000+ E-41

1110 CEPT EST#

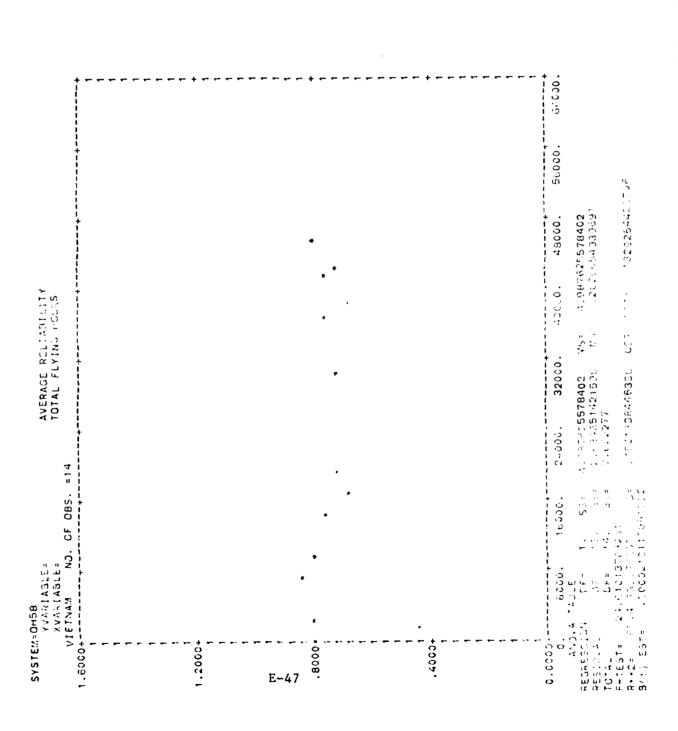
40000 35000. .1787224832966 COMR EDIA .7596658391052 18.72633379407 .03194172603579 30000. 20.00. AVERAGE RELIABILITY TOTAL FLYING POWS :88: MS: 20000. 18 72833079467 1.724853205933 20.451184 15366. NO. CF JAS. =55 REGRESSION DF= 1. SJ= 1. SG= 1. SGS= 0.0000. 5000. 10000. 24.831.401.61 X4191.481.63 ANDVA TABLE WORLDALDE 515:2:3:4: 1.2000-40005. .8000 E-42

160000 140000. . 0013963700906 | CD84 | 651 | 15180101697133 1,1498(2760964 ,1484c34109562 120000. :000000: AVERAGE RELIBBILITY TOTAL FLYING HOURS 80000. 9.149862760964 3.200195239036 12.500058 NO. OF 035. #23 REGRESSION FRANCE RESIDUAL DFR 1. SSR 9. FFIEST DFA 22. SSR 101AL DFA 23. SSR 1 SSR 17.05 TOTAL DFA 23. SSR 1 SSR YVARIABLE= XVARIABLE= XVARIABLE= VIETNAM NOTE .4000t 1.2000+ .80004 E-43

32000. 25000. .2793648451072 CORR EST= .3665464355012 24,25610472667 ,05258071158962 24000. .00000 AVERAGE PELIALE 11Y TOTAL FEYTWA HOLS MS = MS = 16000. 24.25010472067 3.417746279325 27.667851 12000. SION DF = 1. SS = 2 AL DF = 65. SS = 66. SS = 64. SS = 64. SS = 65. SS = 64. SS = 6 ACACIDADES XVAQIABLE CQCUXCOS ANDVA TABLE
REGRESSION DF=
RESIDUAL DF=
TOTAL DF=
F-TEST= 461.197724
R**2= 87.64722983 1.6600+-------5 × 5 TEM= 5.137 0.0000.0 . 4000b. 1.2000+ .8000+ E-44

60008 70000. .3137, 6327, 1131, 60009 10.35107266403 .11202.8272582 56600 AVERAGE RELIABILITY TOTAL FLYING HOURS ·, . 8 :-30000. 40000. . . 347585758374 . 03 .0.35107266403 3.024E0533597 10.305878 NO. OF 085, =28 YVARIABLE= XYARIABLE= VIETNAM NO. 0 0 0 H H H H ... SYSTEM=CH47 0.0000.----AND REGRESSION RESTORAL TOTAL .4000+ 1.2000+ .800C+ E-45

160000. 140000. .2600318682776 CORA EST= .6914319250881 22.33335009156 .06761657251993 120005. 1,0000. AVERAGE PILIN INTERPRETATIONS MS= MS= 80000. 22.3335909156 3.177978908437 25.011338 .00009 1.00 T NO. CF OSS. 237 653797868 87.78233204137 75 00001012811218534 40000. ANOVA TÄBLE
REGRESSION DF =
RESIDUAL DF =
TOTA! 05 +
F-TEST 337 65379
R**2 = 87.78233204
B(1) EST = .000010 20000 8 12 14 21 8 18 0.0000*---4000+ 1.2000+ .8000 E-46



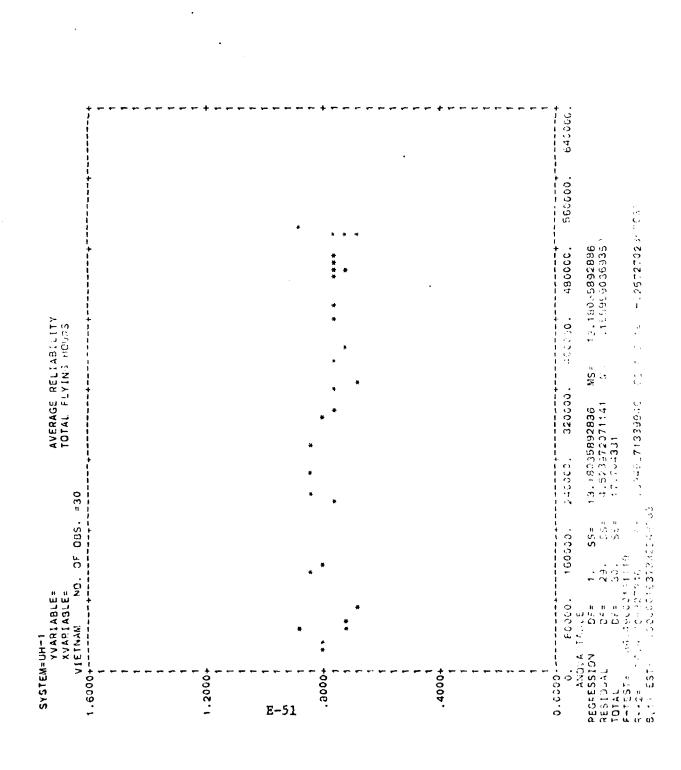
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• •
5230.
REGRESSION DES 1. S RESIDUAL DES 65. TOTAL DES 65.
3.1708.78083 3112118083 .0000943101945

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32000. THE REPRESENTATION 17.540(1161324 1139(50138125 24000. AVERAGE RELIBETATY
TOTAL FLYING HOURS 10.34091161324 4.001300386762 11.002112 .0000 NO. OF GBS. -29 YVASIABLES XVASIABLES VIETNAM NO. SYSTEM=0V-1 PESSOCESSIC: TOTAL TOTAL FITESTE 8:12 ESTE ESTE -.0000.0 +0008. .4000 1.2000+ E-49

640000. **560000.** .0019333907727054 24,47843048339 .06540146:19703 480000. )600. 160600. 243830. 320060. 405.00. 4800 .2557370933012 CG98 F37* AYERAGE STLLT (T) 29.47343043339 4.18569351661 32.634124 #EGYESSION TABLE
REGYESSION DF= 1. SS= 28
RESIDUAL DF= 64. SS= 4
TOTAL DF= 65. SS= 38
F-TEST 435.44027E0033
R**2= 87.1E065507341 S= 38
8(1) EST= .000003740049340812 0.0000*----1.6000+----+0004. 1.2000+ .80004 E-50



40000 35000. .1648016896514 C043 EST# .7554270053635 18.81737736367 .02715959691183 AVERADE MIALLHOLLOS TOTAL FLVING HO PS 23000. 35. 35.= 20000. 16,81737736367 1,439458636328 18,256836 15000. ANDVA TABLE

REGRESSION DF= 1. SS= 11

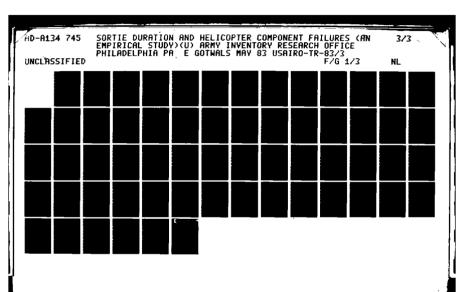
RESIDUAL DF= 53. SS=
TOTAL DF= 51. SS=
F-TEST= 619.205765381

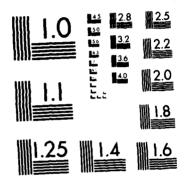
R**2= 92.11550275339

8**2= 92.11550275339 #31841344K SY3777 8--1 0.0000.0 +3005. 1.2000+ .80004 E-52

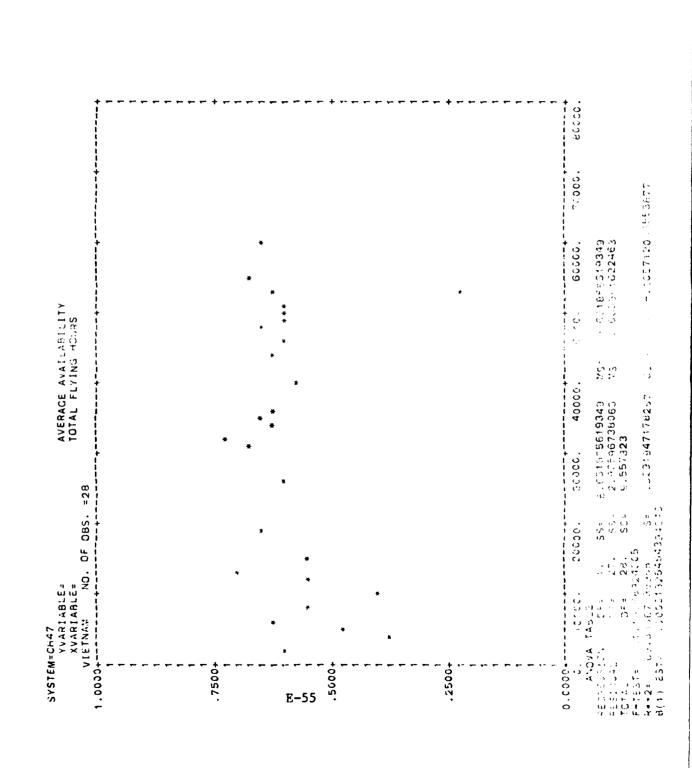
140000. 160000. .0900305543485 CGVA FSTS .07611851817179 5 841969544619 .1521028336803 80000. 100000. 120000. AVERAGE AVAILABILITY TOTAL FLYING HOURS MS÷ _≥5: 5,841959544619 3,347502455381 9,185352 .0000 NO. OF 085. =23 08.20128586.57 63.50128448980 Tr. .00000874724240168 40000 A10/A TABLE
REGRESSION OFF YVARIABLE= XVARIABLE= VIETNAM SYSTEM=AH-1 0.0000*---TOTAL F-TEST= 63 R+*2= 63 .4000+ 1.2000+ .8000 E-53

32000. 28000. .1859824011912 CORR EST# .4412046615905 15.18615290735 .03455226707261 24000. AVERAGE AVAILAGE (219) TOTAL FLYTAS OF SS 20000 MS= 16000. 15.18615290735 2.211345092647 17.397498 12000. \3. QF OBS. ≈65 REGRESSION DF= 1, SS= 15 RESIDUAL DF= 64, SS= 2 TOLAL D7= 65, S3= 3 F+TEST= 439.5124891843 R**2= 87.28329244511 S= 3 B(1) EST= .00004251304165923 7VA2)ASLES XVA2]ASLES WOALDALLE 3.2000+-------ANGVA TABLE REGRESSION DF= RESIDUAL DF= SYSTEM=CHAT 0.0000 -----. 8000-E-54 9: 2.4000+





MICROCOPY RESOLUTION TEST CHART NATIONAL BUREAU OF STANDARDS-1963-A



160000. 140000. TOTAL DF= 47. SS= 23.6CC234
F-TEST= 403 4444727715
R**2= 39.7652566558 S= .2291488683827 CORP LST= .6915324177336
B(1) EST= .6000097547883357 123000. 21.18481062147 .05250320388108 AVESAGE AVALLAGILITY TOTAL FLYING HO 45 163500. MS± 80000 21.18461062147 2.41542337853 23.600234 ecooc. NO. OF CAS. 147 \$5°= \$5°= \$5°= 40000 20000. ANOVA TABLE REGRESSION DE 1 * VALLABLE = XVARTABLE = SYSTET C 12 0.0000.----.4000+ 1.2000+ . 80004 E-56

64000. 5.000.3 . J. 02.060978422 | CO-72 | T. 14475403431547 330. 3000. 23000. 23000. 40000. 40000. J.05856533291 .1371206060991 AVERAGE AVAILABILITY TOTAL FLYING HOURS 4.05256533291 4.007-39266709 0.50958 NO. OF OBS. = 12 70 - 404 - 4947 (384 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 474 - 4 YVARIABLE= XVAPIABLE= --+-----+0000°1 VIETNAM SYSTEM-0H58 6.00030-----2119 051-.7500+ E-57. .2500+

20000 17500. .1839558708863 CCT | Date | 115579983 | 2863 15000. 17,42079909556 .0360832328819 AVERAGE ALA TOLITA MS= RS= 10000. 17.42079909556 2.309326904442 19.730126 0000 i.) JEN DF= 1. SS= 17 AL DF= 64. SS= 2 DF= 65. SS= 1 - 482.7342524617 - 73.29542743763 S= 3 ST= .0000000111157547 5000. 0. 2500. Angva table SVSTEW CV-1 0.0000.----REGRESJON RESJOUAL TOTAL F-1EST: 48 R++2= 70.4 40004. 1.2000--coca-E-58

32000. 28000. 9.0 0 0 128243 C 9 851718623655 .1335792634407 24000. AVERAGE AVAILABLLITY TOTAL FLYING HOURS MS = ... 16000. 9.6317:8623655 5.455219376345 6.1307958 1375285516 12000. NO. 0F 09S. =29 180.603403B3+0300 TVARIABLE= XVARIABLE = VIETNAM NO SYSTEM=OV-1 0.0000.0 R( 1 | EST = 40004 1.2000+ E-59 

A DESCRIPTION OF THE PROPERTY OF THE PROPERTY

640000. 560000. .2372093716197 COPR EST: .2166192269487 23.1348475156 .05326852319381 483000. 400.00c AVERAGE AUNTED TOTAL TOTAL FUYING POLICES 23.134347515G MS= 3.601185484404 MS= 26.736033 320000. 240000. AND/A TABLE
RECRESSION DE= 1. SS= 23
RESIDUAL DE= 64. SS= 3
TOTAL DE= 41. 1507450184
R**2= 86.53059185381
B(1) EST= 150003370952995465 3.2000+-----54574474 0.0000.0 .80c0+ E-60 -2.4000+

645300. 1,41459924 (7.1) . 456495423904 .1708078124519 40,0000. 480000. AVERAGE AVAILABILITY TOTAL FLYING HOURS NO. DF 085. =30 0.0000. YVARIABLE= XVARIABLE= VIETNAM SYSTEM=UH-1 1.60001--. 400¢+ 1.2000+ .80004 E-61

1.0000 .8750 .7500 17.76221859054 .009332403952044 .6250 AVERACE AUNILAGE TY AVERAGE RELIABLITY .5000 .3750 15. Ca Ca. . . 54 AMOUNT TABLE 1. SS= RESPENDING DF= 1. SS= TOTAL DF= 54. SS TOTAL FFEST= 1903.28437151 R*2= 97.29073212551 S*3(1) EST= .9365495484457 T. 14 TABLE 3Y3TEV- .... 1.6000+ .4000+ 1.2000+ E-62

N'S = |∏S = 17.76221859054 .4946174094583 13.256836

.09860436818304 | CORR EST# .8970740922895 i)

1.5000 1,4000 1,2000 1,540041178744 ,0295/3219147⁴⁹ 0010 0110 0146624323 0 1.0000 AVERAGE AVAILABILITY
AVERAGE RELIABILITY . 30% ž: 9.540041178744 ......06212555 9.135052 . 171-132206309 . 6000 NO. OF 085. ≈23 YVARIABLE= XVARIABLE= VIETNAM NO. OF 2005. PECAESOLA PESIOUS TOTAL TOTAL F-16STS 8(1) ESTS 0.0000-1.2000+ .8000+ .4003+ E-63

SYSTEM= AH-1

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0000.0		.1250	. 2550	! ! ! !	.3750	. 56 33	1 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0	3006	.8750	1.0000
REGRESSION RESIDUAL TOTAL		63. 63.	10 CO CO TO	19.196. 4.732. 25.92	19.19624974027 4.732732259727 25.926982	202 203 203 203	19,19521974027 ,06959900381952	4027 381952		
R**2= 80 8**2= 80 6(1) EST=	60.22175678126 80.22175678126 8260325627961	6126 2562796	s,	.2633	.2633162311525	C0.3	ESI: .0053	.05328920111891		

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1.000 .875 1757-350-357:00 ... 0,987827338469 ,021*1057820403 00004 AVERAGE AVAILASILITY AVERAGE RELIABILITY . 1432 BT87858 CC14 -215 -∠⊠ 9, 437837388469 ,5744, 55115368 9,557823 .3750 XVARIABLE= VIETEAN NG. OF OSS. ±28 YVARIABLE= ANGVA (AUE PESAESTINA RESIDUAL TOTAL FHIESTE ASSIGNED SYSTEM=CH47 0.0000-0-1.0000+--1,1 E2 (1) .2560+ .7503+ .5000+ E-65

COLORS DECEMBER OF STATES OF STATES

1.0000 .8750 .688+1120776519 CLR .ST= .8754767133457 .7500 23.26151008481 .00730356337812 AVERAGE ABLIANTERS .503 23.26151008461 .33872\$5150905 23.600234 .3750 .2500 5.5 5.5 5.5 5.5 5.5 5.5 0.6000------0.0000 ANDVA TABLE RESIDAL TOTAL F-1551. 31 F-1551. 31 R**2* 50.15 .830¢+ +0004. 1.2000+ E66

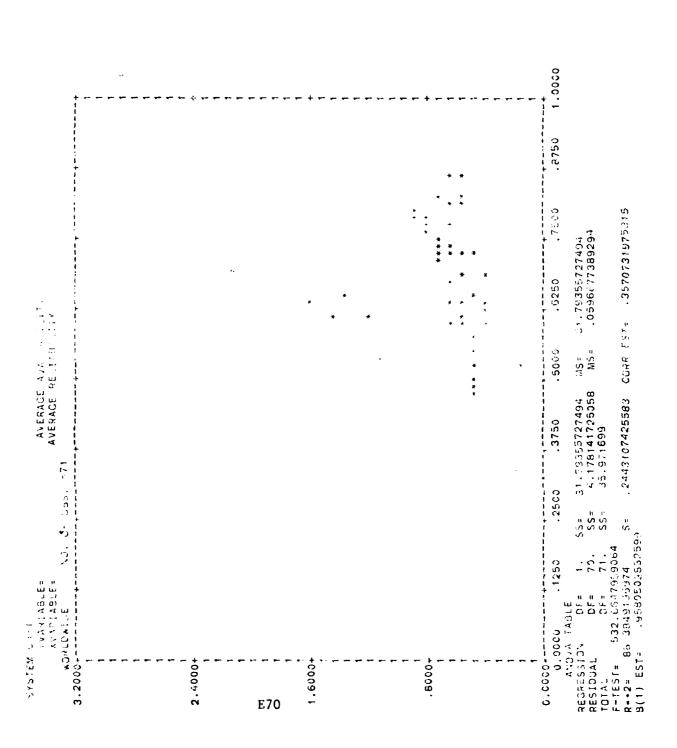
1.0000 0000 029608152713 AVERAGE AVAILABILITY
AVERAGE RELIBELITY ι . . . 2.1.13765000667 1.1.117.039630 3.507.038 NG. OF 385. = 12 YVARIABLES AVARIABLES VIETNAM NO SYSTEM=DH58 1.3003+ .7503+ .2500+ .50004 E67

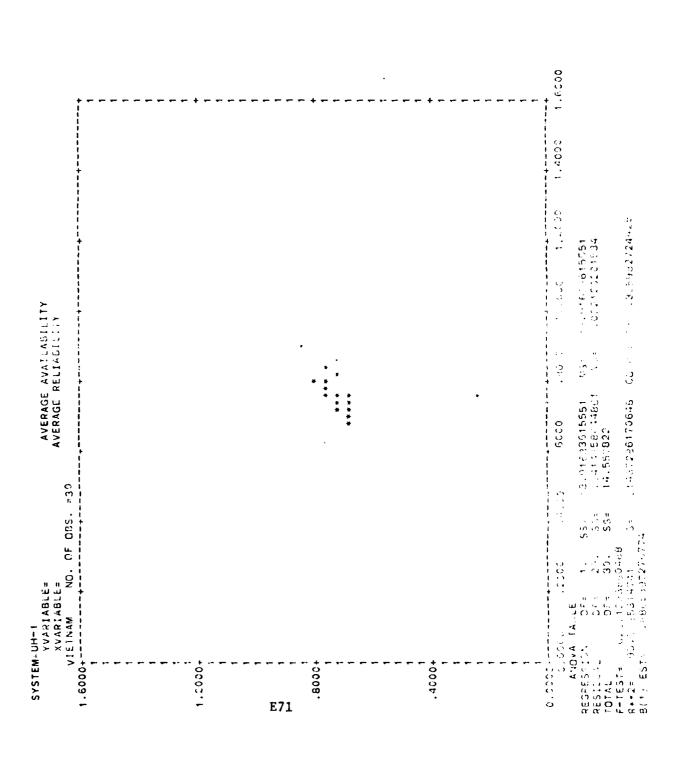
.0170,244607010

1.0000 .2006463289379 COMM TSTA -.03905993268572 MS= 12.51434049718 MS= .04095894931627 AVERAGE ASSESSED 119 22.51498649718 2.777867502823 25.132848 0.0000. 1.2000-.400C+ .8000 E68

1.6000 1.4000 ...365146000164 1.2000 12,779T2102647 ,01536489191172 1.0000 AVERAGE AVAILABILITY AVERAGE RELIASILITY . : 270155246141 | 66. | 23 90 E .0.77372107647 .5282169735282 0009. NC. OF 085. =29 0.60% 777,4740074494 .03 61275.17 .807097894093 YVARIABLE≖ XVAPIABLE= RESPESSION OF VIETNAM SYSTEM=0V-1 0.0000-0 .8000+ .4000+ 1.2000+ E69

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## APPENDIX F

## CAUSAL VARIABLES

## REGRESSION ANALYSIS

Y VARIABLE

X VARIABLE

(DEPENDENT

(INDEPENDENT)

USAGE

TOTAL FLYING HOURS

LENGTH OF SORTIES

TOTAL FLYING HOURS

(BY AIRCRAFT)

USAGE

LENGTH OF SORTIES (BY AIRCRAFT)

35000. 15000. 20000. 7.000. 30000. 350 -,420225916 175 AVERACE USADE TOTAL FLYING HOURS ANDVA TACLE

REGRESSION
OF 1. 55 ... C67801776676513 This
RESIDUAL
TOTAL
F-TEST
A. 25 ... LONGOUT COME
F-TEST
A. 25 ... LONGOUT COME
B. 17 LONGOUT CO 15000. NC. OF OBS. =55 ______00000.0 YVARIABLE= XVARIABLE= WOPLCHIDE SYSTEM= A'1-1 +0090. .02C0+ .0400+ F-1

160000. 100000. 120000. 140000. 160000 .02350406583389 CDRR EST# .9494134050422 .099372295**56**408 .0005524411107237 AVERAGE US: 1 MS= ¥Ç≌ 80000. .09937229556408 .01215370443592 .111526 60006. AND:A ASEE
REGRESSION DF= 1. SS= .65
RESIDAL DF= 22. SS= .70
F-TEST= .79.8755311860
R**2= 89.10235780363 S= .66
B(1) EST= .000001157334622037 - コークマン・サート ******** .1630+----315722.23-5 0.0000. .1200+ .0400+ +2080. F-2

32000. 22000. 7.0550301077.098 .02031:22761494 .0001:294456:5:50 2.:000. 0 AVERAGE USACE TOTAL FLYING HOUMS O 16000. .07.72277233000 .07.72877233000 .04754 .0 7 0007348087 12002. NJ. OF DAS. =65 6030. YVARIABLE= XVARIABLE= #ORLD#[DE SYSTEM=CH47 0.0000. .0630. .0206+ +0040 (i) (i) (i) (i) (i) (i) F-3

F31: 2002: 12:0003:

700007 60000 50000 Si AVERACE UST E TOTAL FLYING H. 40000. 70. Cf 0.5. 1.8 20000. 47578 ABLES 47491 ABLES 100001 VIETNAM SYSTEMPORMS 0.0000.0 .0400+ .1200+ 0080. F-4

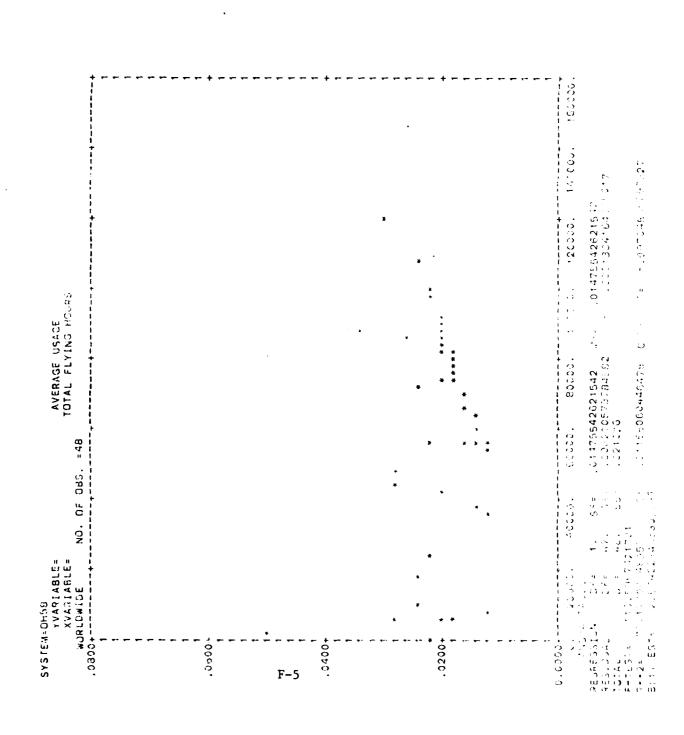
.1039646146268 .0004254216804896

ANOVA TABLE

8000B

MS= MS= .1039546146268 .01148638537322 .115451 .5184814032592 .02062575284661 CORR EST=

REGRESSION RESIDUAL TOTAL F-TEST= 24 R**2= 90.0

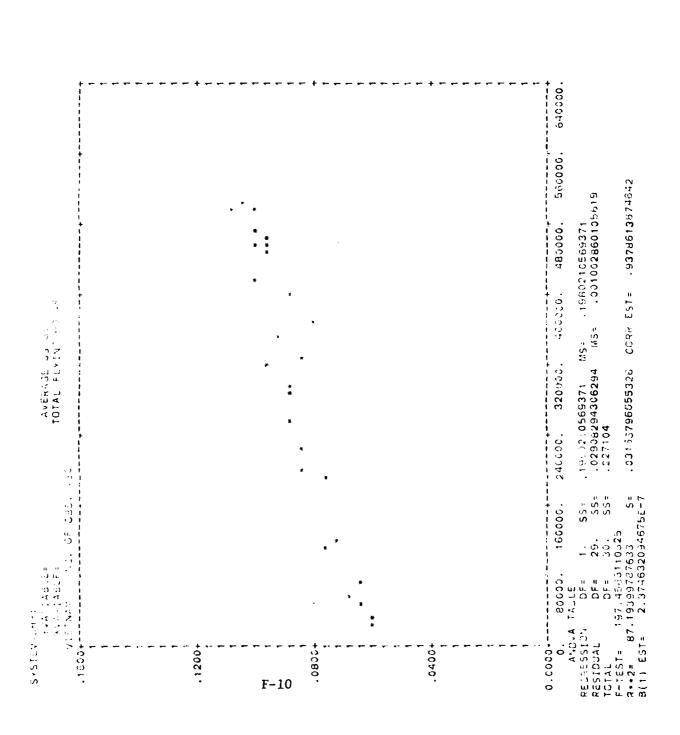


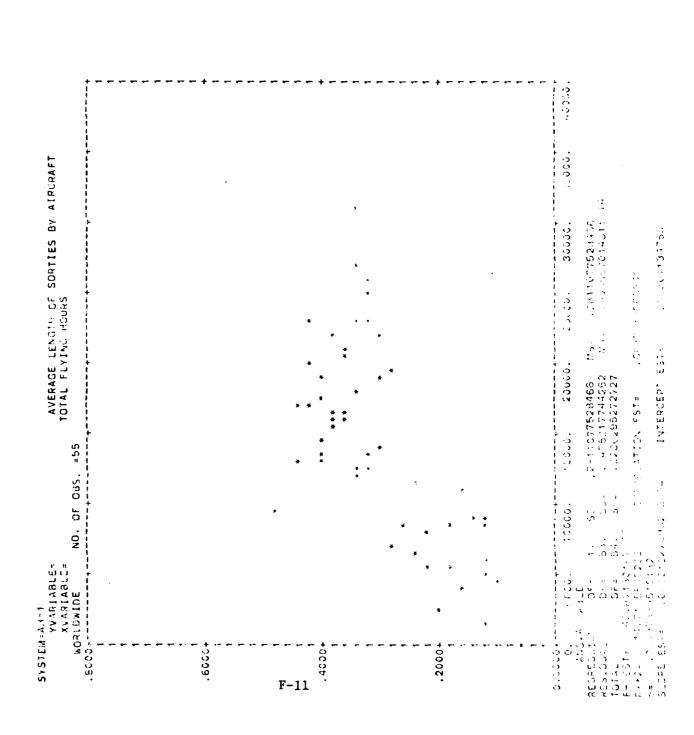
64000. 56000, .03554293556361 | CORR EST# .7354819549004 .04924336997991 .00129189461693 48000. .0000 ¥8. ₹8.5± AVERAGE SOLD 32006. .0492433699793; .01679463002009 .066038 2-1000. 13. Of C35. F.4 38.11717251128 74.55823341393 7= .000002177182783363 888 888 888 888 16000. 145 146_E: *VAK1A8_E: V1E1433 \13. 0, 8000. 4NOVA TABLE LESIDUAL DF= TOTAL DF= F-TEST= 38.1171725 3**2= 74.508233410 515 EB: 0 48 a ----+0000. 1606+ .1200+ .0630. .040c+ F-6

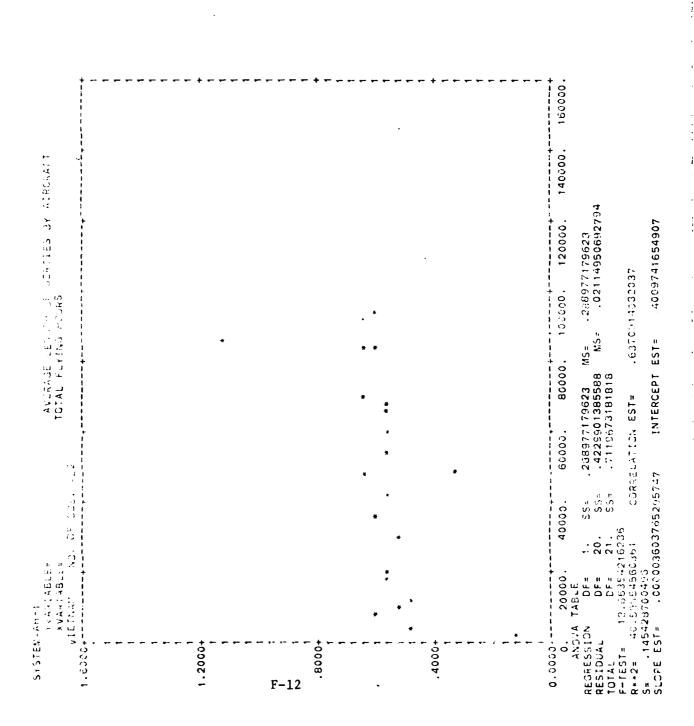
20000 17500. 20000 .0189641308531 .00002534170542035 .003 94654570557 JAH 1814 64175917 FT4 15000. AVERAGE USAGE TOTAL FLYING HOURS 10000 7300. NO. OF 085. =65 #EGRESS104 DF=
#RESTOCAL
TOTAL
F-115'= 711, 98710"
8(1) ES70 SYSTEM: OV-1 0.0000----.0660 .0200+ +0C:C. F-7

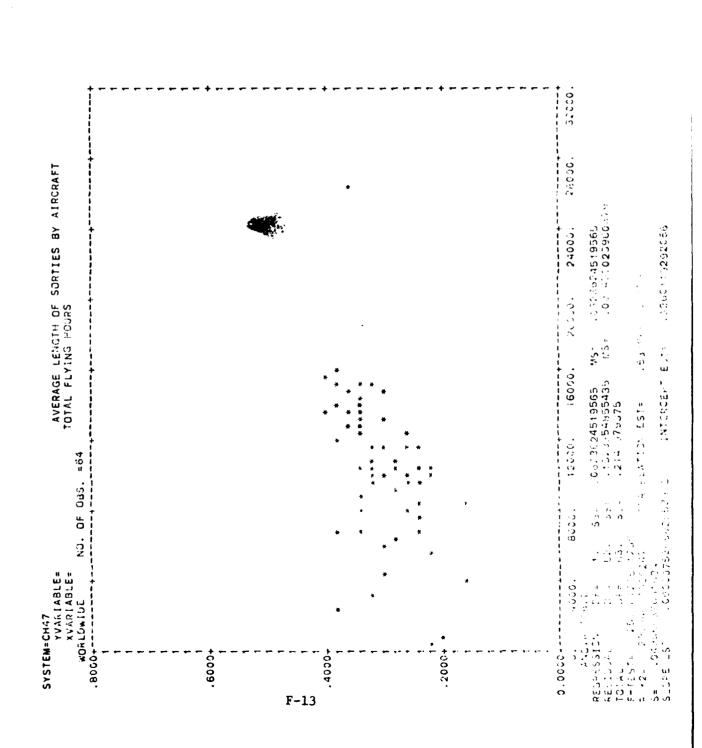
32000. 28000. .03031022151249 CORP EST# .569727885262 .1202331**332122** .000518709**52**81361 24000. 20000 AVERACE USE TOTAL SEFTING HOUSE MS= 16000. .02572386678781 .02572386678781 .148057 12000. YOASTAGLE-XV-+17ABCE-VIETVOT 0.0000*-----SYSTER CULT .1600+---.0400+ .1200+ .0800+ F-8

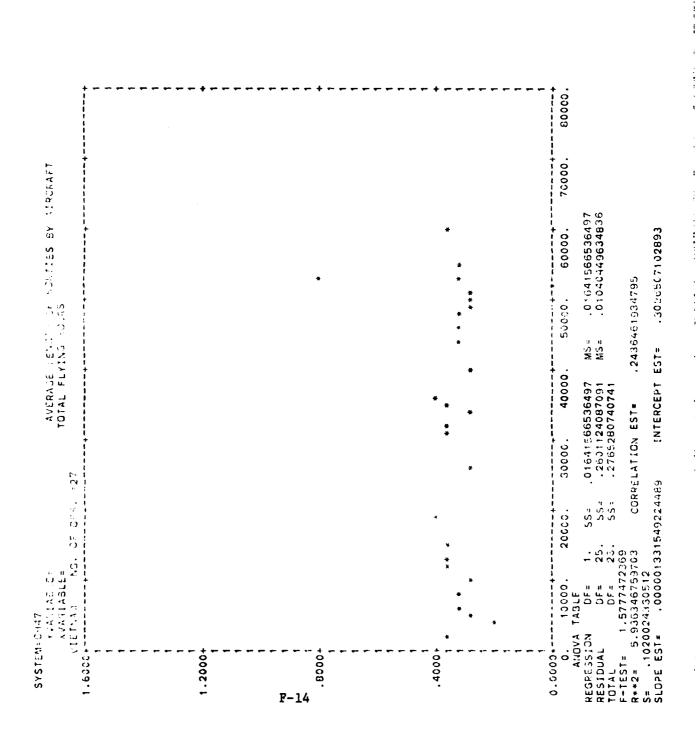
040000. Ac1000. Service Contractions 100019188639 000100781382 sacoco. AVERAGE USAGE TOTAL FLYING HOURS 0.0000 320000 320000 320000 4 ģ. 15 2192905932... .0001013180637 .0111-20411300 .00174 NJ. OF OBS. =65 SYSTEM=UH-1 .0750+ .0530+ .0250+ F-9











160000. 140000 AVERAGE LENGTH OF SORTIES BY AIRCRAFT TOTAL FLYING HOURS .01258473341963 .06424111656231 120000. 119103,394531 -,653-0470321215 160000. TYTERCEPT ESTA 80000 .01755473341963 0.115650245304 3.125434978723 C. A. B. ATLU. ESTA 60000 NO. GF C3S. =47 -5.6100712010448-7 1 0 5 10 0 5 10 0 0 10 40000. YVARIABLE XVARIABLE XVARIA 30008 RESTORE STORE TOTAL TOTAL TESTS OF THE STORE TOTAL TESTS OF THE STORE .8000+ 2.4000+ 10000 0.0000 **F-1**5

64300. 56000. AVERAGE LENGIN OF BORTIES BY AIRCRAFT TOTAL FLYING HOURS .002243247356163 48000. .4091833538379 -.1112482807423 40000 MS# INTERCEPT EST= 32000. .002243247356163 .1790119834131 .1812552307692 CORPELATION ESTA 24000. F-TEST= .1378439613222 R**2= 1.237617296912 CDRPE S* .1275636703168 SLOPE EST= -8.383206352965E~7 \$5°= \$5°= YVAR! ABLE = XVARIABLE= VIETNAT SYSTENADASES REGRESSION RESIDUAL 1.2000+ .8000+ .4000+ 0.0000 F-16

20000 .3800. AVERAGE LENGTH OF SORTIES BY AIRCRAFT TOTAL FLYING HOURS .05567.676975958 .01846193136412 15000. 17,55071750203 9890611. · ハラン j. INTERCEPT ESTA 16000. .00587676978958 1.784915464575 1.808.98234375 TRANSTATION ESTE 7506. NO. OF 045. =64 174123271 . 23 YVARIABLE= XVARIABLE= WORLEWIDE SYSTEM=UV-1 .5000+ 2.0000+ 1.5000+ 0.000.0 1.0000+ F-17

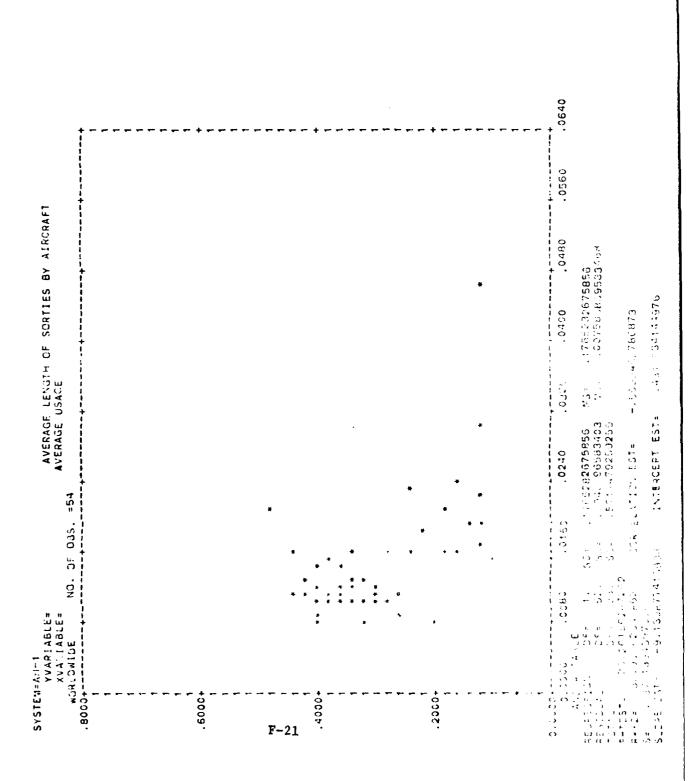
AVERAGE LENGT OF SOMTIES BY ALRORAFT TOTAL FLYING MODAS 

7. 8000. 12000. 16000. 2000. 24000. 28000. 1. 55= ,8176599611137 MS= .8176599611137 26. 55= 1.881142753172 MS= .07235164435277	8000. 12000. 16000. 20000. 24000. 28000. 28000. SS= .8176599611137 MS= .8176599611137 SS= 1.881142753172 MS= .07235164435277 SS= 2.698802714286 SS= .5504283531972	SS= .8176599611137 MS= .817659611137 SS= 1.881142753172 MS= .07235164435277 SS= 2.69802714286 SC= 2.69802714286 SS= 2.69802714286 SS= 2.69802714286 SS= 2.69802714286 SS= 3.69802714286
1. SS= .8176599611137 MS= .817659611137 26. SS= 1.881142753172 MS= .07235164435277	SS= .8176599611137 MS= .817659611137 SS= 1.881142753172 MS= .07235164435277 SS= 2.698802714286 31 CGRPELATION EST= .5504283531972	SS= .8176599611137 MS= .8176549611137 SS= 1.881142753172 MS= .07235164435277 SS= 2.698802714286 CGRRELATION EST= .5504283531972
1. SS= .8176599611137 MS= 26. SS= 1.881142753172 MS=	SS= .8176599611137 MS= . SS= 1.881142753172 MS= . SS= 2.698202714286 81 CCRPELATION EST= .550428	SS= .8176599611137 MS= .SS= 1.881142753172 MS= .SS= 2.69802714286 31 CCRPELATION EST= .550428
	SS= 2.698802714286 31 CGRRELATION EST= .550428	SS= 2.698802714286 31 CCRRELATION EST= .550428 20442752 INTERCESS 6572
	9759	129641752 INTERCET FAT
CORRELATION EST#		100850757 INTERCELL FORE

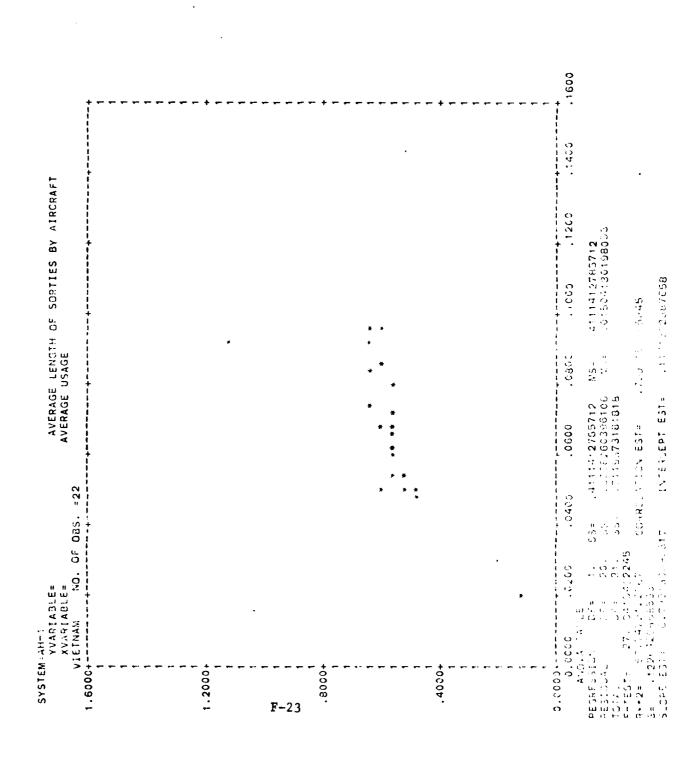
640000. 550000. AVERAGE LENGTH OF SCRIIES BY AIRCRAFT TOTAL FLYING HOURS .073183946**512** .003404371**5784**36 480000, INTERCEPT EST- 10472342014258 £0408 40408. 400000 M3± 335= 320000. .073183946512 .214822037863 .288005984375 CORME, 4,10% EST= 240000. .05.4631.977278 EST= 5.7057*24211516-7 AVARIABLE= #ORLD#:DE SYSTEM=UH-1 .80000 0.0000----.2005. .60004 .40004. F-19

: ⊷	10 G	6 IV C C C C C C C C C C C C C C C C C C	-29	AVERAGE LEGGET TOTAL FLYING HT	Sa ( ) 22	S E S	BY AINCRAFT	
F-20 .	• • •	•	•	* *	•	*	* * * * ·	
47.02 <b>4</b>	30000	130000.	240000.	320000.	40000		56	640000.
REGRESSION RESIDUAL TOTAL	00 00 00 00 00 00 00 00 00 00 00 00 00	110N DF= 1. SS= 11 DF= 27. SS= 12 DF= 23. SS= 13 SS=		5.961631981299E-8 .08835718176299 .08635724137931	MS=	5.961631981299E-8 .003272488213444	11299E-8 113444	
R**2= .00	.00008747191 .00008747191	1161963	CORRELATION EST=	ION EST=	.00062	.0008214128780317	7	
ď	2.7383	2.738373610879E-10		INTERCEPT ESTA		.3804373389423	_	

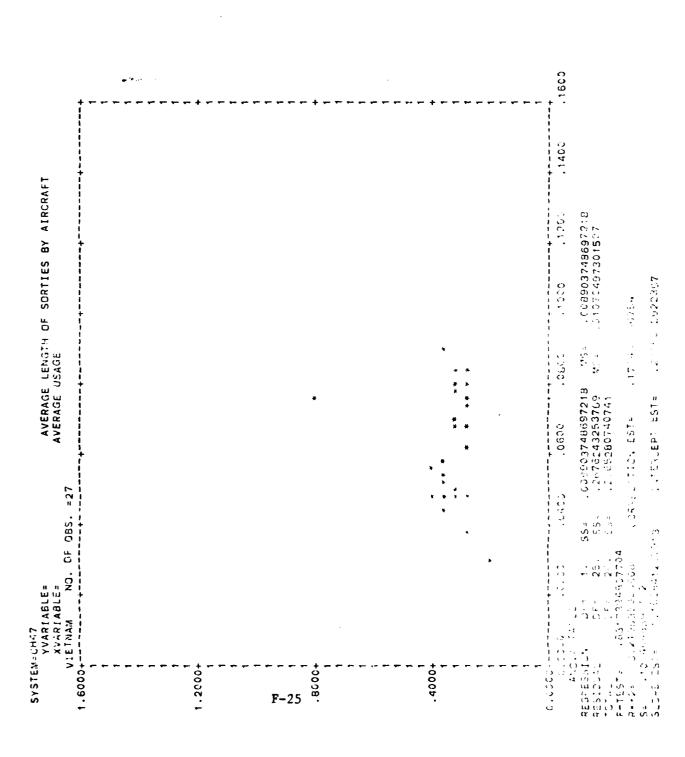
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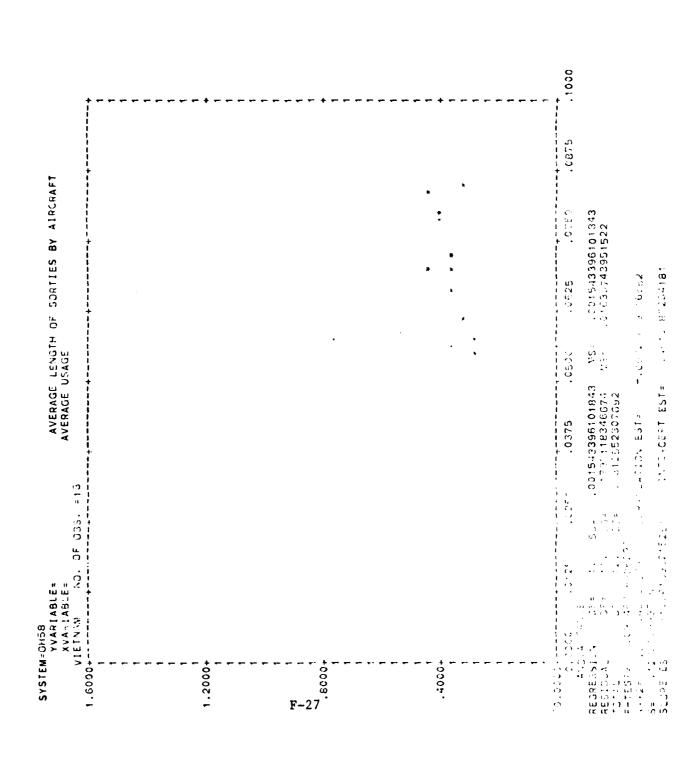


.0640 .0560 AVERAGE LENGTH 7 SCHIIES BY AIRCHAFT AVERAGE USAGE 0480 .003191426198645 .348 8921907807 0050. -.27997us678437 #S# MS= .0320 INTERCEPT ESTA .01682551318404 .197868424316 .2146979375 CORRELATION EST= .0240 13. 38 535, 864 .0160 -2.111990385754 F-TEST= 5.273581829719 R**2= 7.838693459287 S= .05349270925212 SLOPE EST- -2.111001011 .0380 0.0000*-------YUNCURDURA AKACIABUBU O.0000 ANDVA TABLE REGRESSION DEFT RESIDUAL FOTAL MORE ALVE .8000+---System 17 4000+ .6000+ .2005. F-24

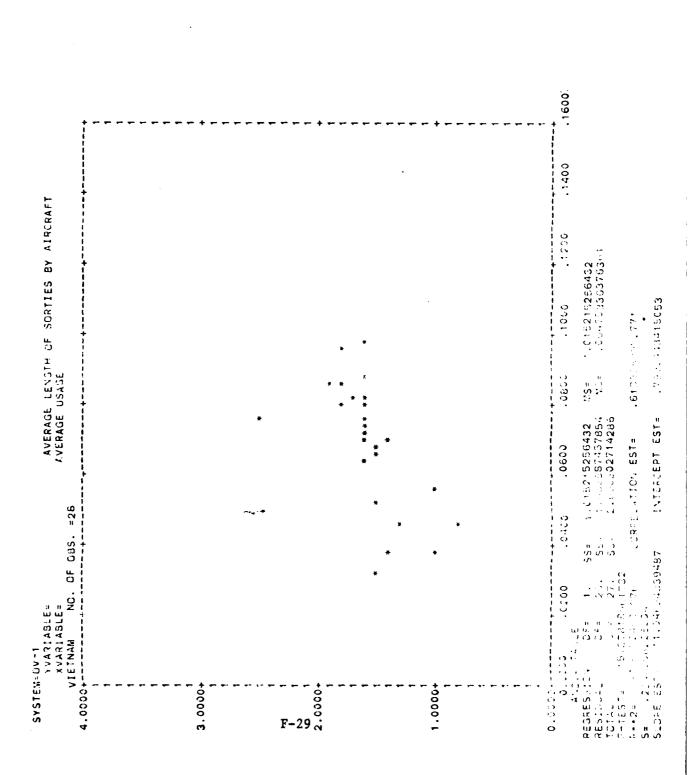


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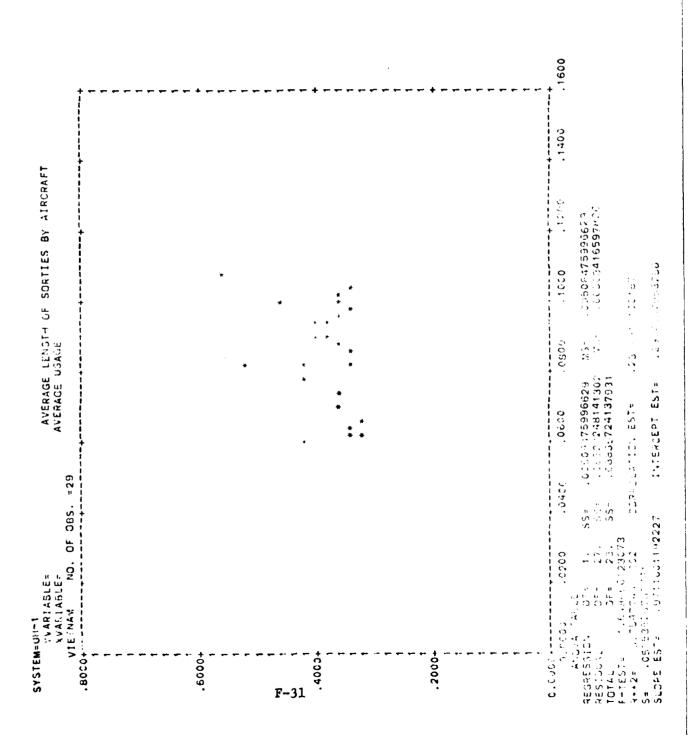
.0640 .0560 AVERAGE LETUT - OF LOPTIES BY ATSCRAFT AVERAGE CLASIC .0480 1,241058831645 .0419415921573 -,12715-1254019 .0400 .629843451295 MS= MS= 1.241058831645 1.887376147078 3.128434978723 INTERCEPT EST# CORRELATION ESTE .0240 .0160 F_TEST= 29.5900cm20226 R**2= 30.6707731391 S= .204730702442 SLOPE EST= 25.4311453598 45. 46. . მიგა ANGVA TABLE 0.0000 SYSTETIONS REGRESSION RESIDUAL TOTAL 0.0000.0 .8000+ 2.4000+ 1.60004 F-26



	.0640			
	.0560			
	0480	<b>c53</b> 169		
U)	.0460	.23377262141 .025759990 <b>53</b> 169	.3573268228433	.6245490222234
	.0320	S= MS=	.357326	.624
TO T	.0240	.25377282141 M 1.597119412965 1.830892234375	CORPELATION EST=	INTERCEPT EST=
	.0160	30 CO CO		•1
But and a second a	0800.	ANCE   DF= 1.   DF= 62.   DF= 63.	12.76824583233 04991916855	T= 10.11154387964
7 F-28 -	0000.0	REGRESSION RESIDUAL TOTAL	8**2= 5**2=	_



.0800 .0700 TES BY AIRCRAFT 0090. .05096871192117 .003823181813375 .425-030106089 .0500 -.4200792327851 :5 AVERAGE SENCE + CE AVERAGE USAGE .0400 INTERCEPT EST# .0330372724478 .2370372724478 .238005984375 CCRYELATION ESTA .0300 NO. OF 033. #64 .0200 REGRESSION DF= 1. SS= RESIDUAL DF= 63. SS= TOTAL DF= 63. SS= F-TEST= 10.33149047342 R**2= 17.69710168967 CCR-S* .06183188347134 SLOPE EST= -2.852857312952 \$S= \$S= \$S= .0100 ALPRIABLES ALPRIABLES WORLDAIDE ANDVA 148LE ANDVA 148LE RESIDUAL DF= TOTAL DF= F-TEST= 10.33143 SYSTEM-111-1 .8000+ +0009. .2000+ 40004. F-30



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### APPENDIX G.

## SAMPLE DATA COLLECTION - BRIGHT STAR EXERCISE

### UH-60 - BLACKHAWK

## REGRESSION ANALYSIS

Y VARIABLE X VARIABLE

(DEPENDENT) (INDEPENDENT)

TOTAL SORTIES TOTAL FLYING HOURS

MAINTENANCE EVENTS SORTIE LENGTH

PER FLYING HOUR

MAINTENANCE EVENTS FLYING HOURS

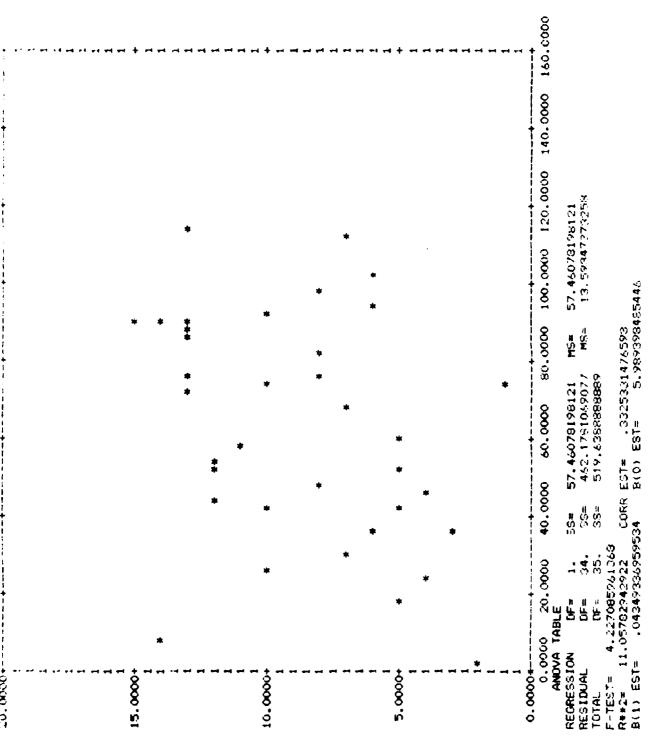
MAINTENANCE SORTIES SORTIES

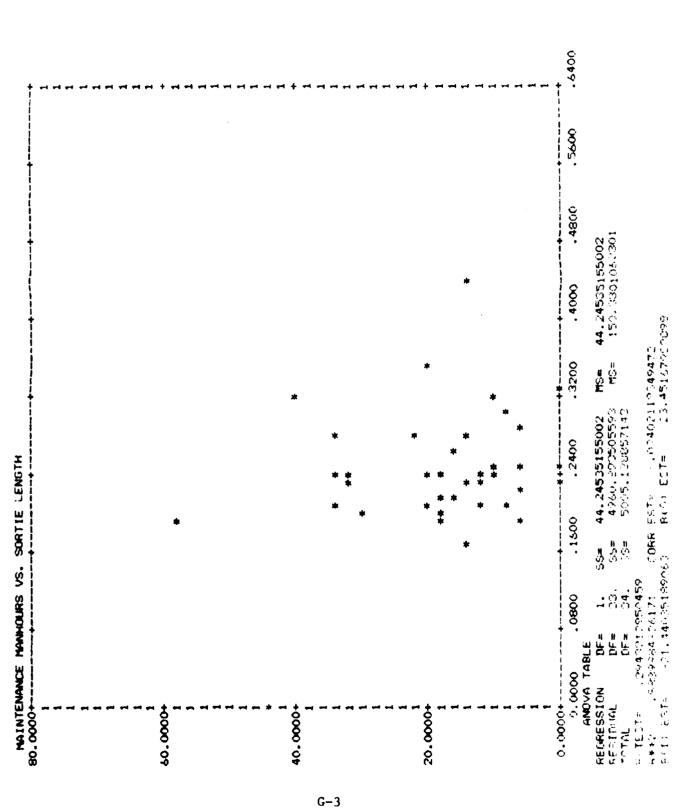
MAINTENANCE MANHOURS SORTIE LENGTH

MAINTENANCE MANHOURS FLYING HOURS

G-1

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		•	•			ਸਰਾਜ <b>ਰ †</b> ਜ਼ਰਾਜਦਾਜ਼ਜ਼ਜ਼ <del>,</del> ਜ਼ਰਾਜਦਾਜ਼ † ਜ਼ਰਾਜਦਾਜ਼
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T= .2708209197608	(01/01/001/					

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1	DRCSM
1	DRCSM-E
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<del></del>	DALO-SMZ-B
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1	Ofc, Asst Sec'y of Defense, ATTN: MRA&L-SR, Pentagon, Wash., DC 2031
<del></del>	Commandant, US Army Logistics Management Center, Ft. Lee, VA 23801
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	DRXSY-PM ATTN: Sandy Johnson
<u>i</u>	Commander, US Army Logistics Center, Ft. Lee, VA 23801
<u>-</u>	Commander, US Army Logistics Evaluation Agency, New Cumberland Army
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1	Depot, New Cumberland, PA 17070
1	Commander, US Army Depot Systems Command, Chambersburg, PA 17201
<u> </u>	Commander, US Air Force Logistics Cmd, WPAFB, ATTN: AFLC/XRS, Dayton, Ohio 45433
1	US Navy Fleet Materiel Support Office, Naval Support Depot, Operations
	Analysis Department, Code 93, Mechanicsburg, PA 17055
1	Mr. James Prichard, Navy SEA Systems Cmd, ATTN: PMS 3061, Dept of
	US Navy, Wash., DC 20362
1	George Washington University, Inst. of Management Science & Engr.,
	707 22nd St., N.W., Wash., DC 20006
1	Naval Postgraduate School, ATTN: Dept of Opns Anal, Monterey, CA 93940
1	Air Force Institute of Technology, ATTN: SLGQ Head Quantitative Studies
	Dept., Dayton, OH 43433
1	US Army Military Academy, West Point, NY 10996
!	Commander, US Army Logistics Center, ATTN: Concepts & Doctrine
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1	Scientific Advisor, ATCL-SCA, Army Logistics Center, Ft. Lee, VA 23801
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Ţ	Commander, Materiel Readiness Support Activity, Lexington, KY 40507
1	Director, Army Management Engineering Training Agency, Rock Island
	Arsenal, Rock Island, IL 61299
1	Defense Logistics Agcy, ATTN: DLA-LO, Cameron Sta, Alexandria, VA 22314
- <u>-                                  </u>	Dep Chf of Staff (I&L), HQ USMC-LMP-2, ATTN: LTC Sonneborn, Jr.,
	Wash., DC 20380
1	Commander, US Army Depot Systems Command, Letterkenny Army Depot,
ı	ATTN: DRSDS-LL, Chambersburg, PA 17201
<del>1</del>	Logistics Control Activity, Presidio of San Francisco, CA 94120
L L	Operations Research Center, 3115 Etcheverry Hall, University of
1	California, Berkeley, CA 94720
1	Dr. Jack Muckstadt, Dept of Industrial Engineering & Operations Research,
	Upson Hall, Cornell University, Ithaca, NY 14890

1_	Dept of Industrial Engineering, University of Michigan, Ann Arbor, MI 48104
1_	Prof Robert M. Stark, Dept of Stat & Computer Sciences, University of Delaware, Newark, DE 19711
_1_	Prof E. Gerald Hurst, Jr., Dept of Decision Science, The Wharton School, University of Penna., Phila., PA 19104
1 1 1	Director, AMSAA, ATTN: Allen Hill, DRXSY-FLSO, Ft. Lee, VA 23801
	Director, AMSAA, ATTN: Paul Arvis, DRXSY-PRO, Ft. Lee, VA 23801
	Dept of Industrial Engr. & Engr. Management, Stanford University, Stanford, CA 94305
_1_	CC-LOG-LEO, Ft. Huachuca, AZ 85613
	Commander, US Army Test & Evaluation Cmd, ATTN: DRSTE-SY, Aberdeen Proving Ground, MD 21005
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1	AFLMC/LGY
1	AFLMC/XRP, Bldg. 205
1	Engineer Studies Center, 6500 Brooks Lane, Wash., DC 20315
	Commander, US Army Missile Cmd, ATTN: Ray Dotson, DRSMI-DS, Redstone
	Arsenal, AL 35898
1_	
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•	Rochester, NY 14627
	Prof Barney Bissinger, The Pennsylvania State University, Middletown, PA 17057
	Prof Garry D. Scudder, Mgmt Sciences Dept, College of Business Adm, University of Minnesota, 271 19th Ave., South, Minneapolis, MN 55455
	Center for Naval Analyses, ATTN: Stan Horowitz, 2000 N. Beauregard St., Alexandria, VA 22311
1	Dr. Warren H. Hausman, Industrial Engr. & Engr. Management, Stanford University, Stanford, CA 94305
1_	Dr. Paul Zipkin, Graduate School of Business, Columbia University, 416 Uris Hall, New York, NY 10027
1	Mr. Craig Sherbrooke, Logistics Management Institute, 4701 Sangamore Road, Wash., DC 20016
1_	Commander, Aviation Center, ATTN: LT Thompson, OTB, ATZQ-T-AT-NT,
	Ft. Rutger, AL 36362
	HQ, WPAFB, ATTN: Virginia Williamson, AFLC/XRS, Ohio 45433
1_	Mr. Maurice Shurman, 6525 46 NE, Seattle, WA 98115

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